

DESIGNING TURBINE ENDWALLS FOR DEPOSITION RESISTANCE WITH 1400C COMBUSTOR EXIT TEMPERATURES AND SYNGAS WATER VAPOR LEVELS

**Brian Casaday, Josh Webb, Carlos Bonilla
Dr. Ali Ameri, Dr. Jeffrey Bons
“THE” OHIO STATE UNIVERSITY**

**Robert Laycock, Dr. Thomas Fletcher
“THE” BRIGHAM YOUNG UNIVERSITY**

(3-year grant awarded Oct 2008 – 6 Month Extension)

MOTIVATION/NEED

- Operational Issues
 - Fuel flexibility (range of feedstock heat release)
 - Diluent use (e.g. steam)
 - Filtration requirements
- Technical Challenges
 - Higher firing temperature
 - Increased heat transfer (steam diluent)
 - Potential for increased levels of airborne contaminants
 - Deposition rate increases with temperature
 - Advanced cooling, greater reliance

OBJECTIVES

The objectives of this work are to study turbine deposition at elevated gas temperatures and water vapor levels

AND

explore modifications to turbine endwall geometries that reduce the potential for degradation due to deposition.

The effort includes both experimental and computational components, with work divided into three phases.

- 1) Modeling and Experimental Validation
- 2) CFD and Experimental Endwall Design Study
- 3) CFD Design Study with Cooling & Experiments with Steam

RESEARCH TEAM

TEAM LEAD

Focus: Experimental Heat
Transfer and Deposition
Measurements in OSU
Turbine Reacting Flow Rig



Dr. Jeffrey Bons

Professor
Department of Mechanical and
Aerospace Engineering
Ohio State University
Columbus, OH

Co-PI

Focus: Deposition Model
Development and Heat
Transfer CFD



Dr. Ali Ameri

Research Scientist
Department of Mechanical and
Aerospace Engineering
Ohio State University
Columbus, OH

Co-PI

Focus: Experimental Work in
BYU's Turbine Accelerated
Deposition Facility with
Elevated Temperatures
(1400C) and Steam Injection



Dr. Thomas Fletcher

Professor
Department of Chemical Engineering
Brigham Young University
Provo, UT

RESEARCH FOCUS



Gas Turbine Deposition Accelerated by
Steam Diluent Injection used for NO_x Control

Deposition models developed at **OSU** will be validated with experimental data from **OSU** and **BYU** deposition facilities and incorporated into commercially available CFD.

Research at **BYU** will explore the sensitivity of deposition to:

- i) Gas temperatures over the range 900-1400C
- ii) Water vapor concentrations up to 25%

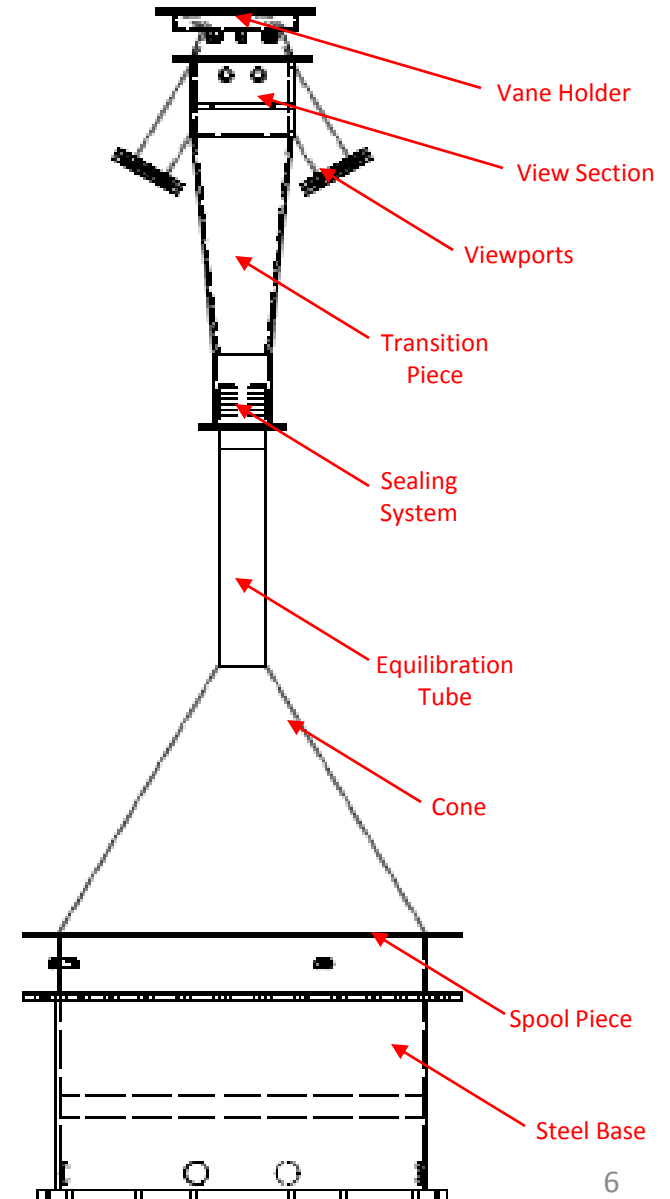


Failed turbine rotor platform
with elevated deposition

OSU's **T**urbine **R**eacting **F**low **R**ig (TuRFR)

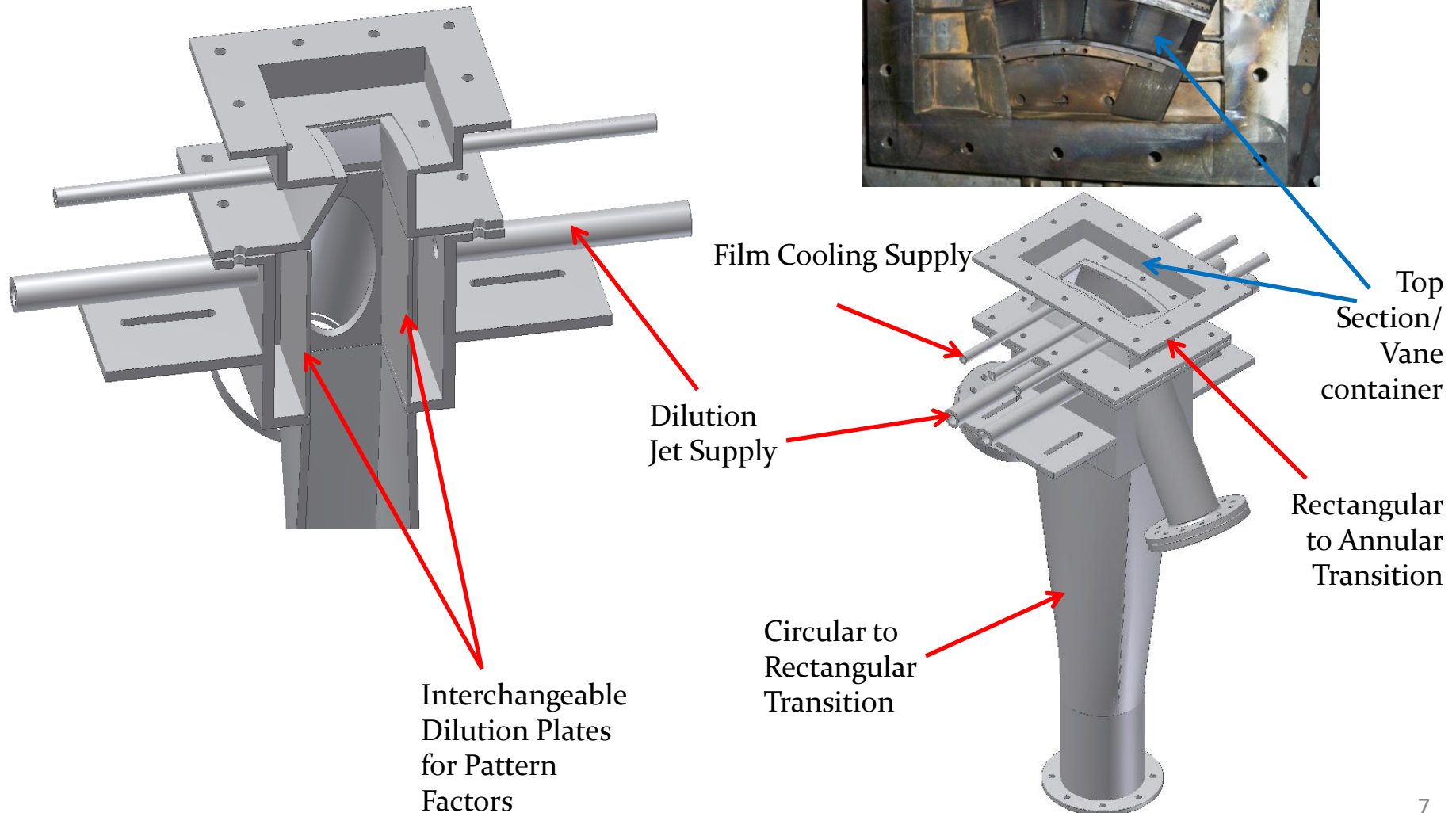


- Natural gas burning combustor rig
- Combustor exit flow accelerated in cone nozzle
- Transition from circular to annular sector
- Real vane hardware (CFM56) installed in annular cascade sector
- Tt4 up to 1120°C (2050°F)
- Inlet Mach number ~ 0.1
- $300,000 < Re_{cex} < 1,000,000$
- Adjustable inlet temperature profiles
- Adjustable inlet turbulence profiles (through dilution jets)
- Film cooling from vane casing and hub (density ratio 1.6-2.0)
- Ash particulate feed in combustion chamber (10 μm MMD)

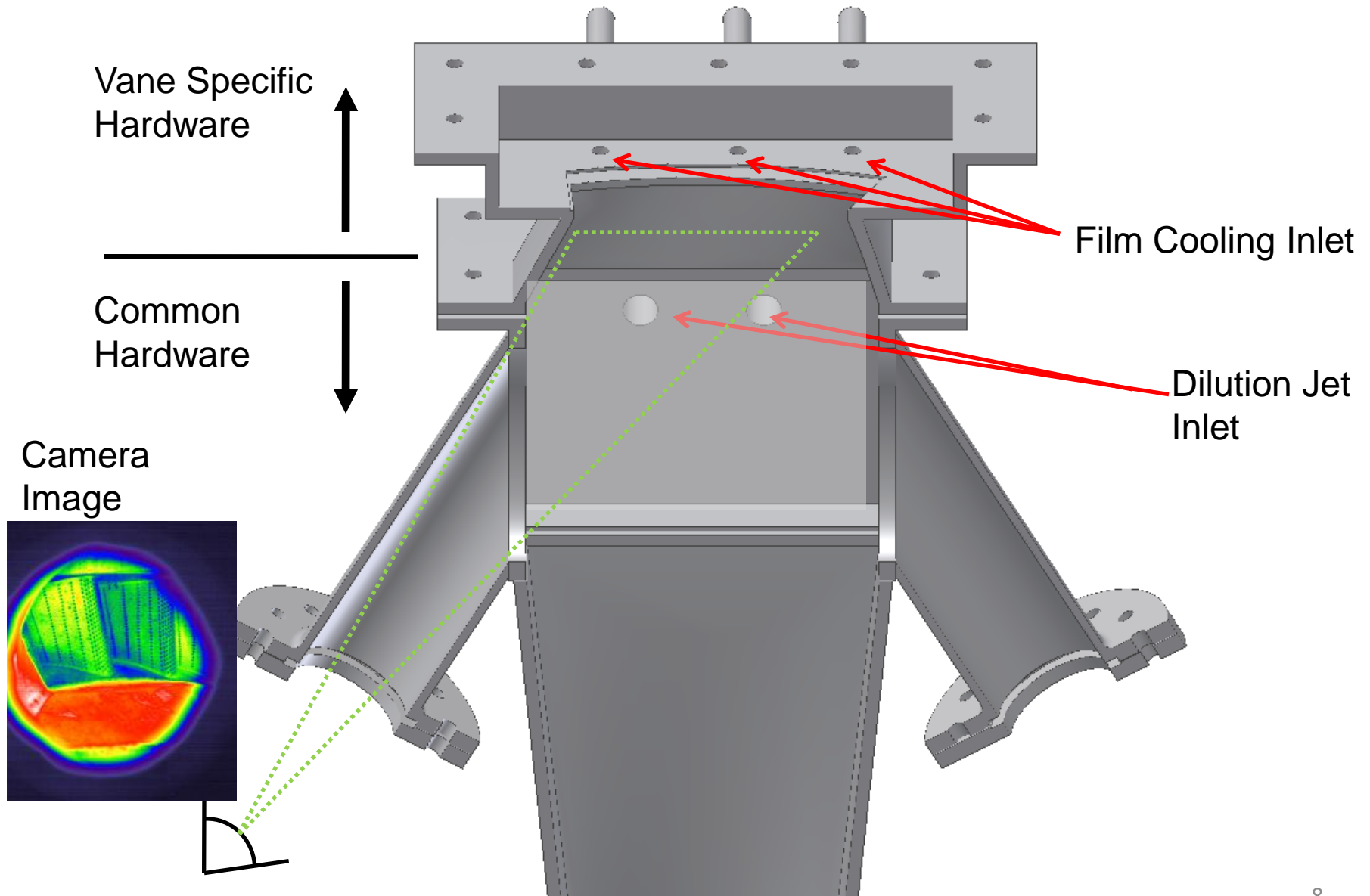


OSU's Turbine Reacting Flow Facility (TuRFR)

Vane Holder and Upstream Conditioning

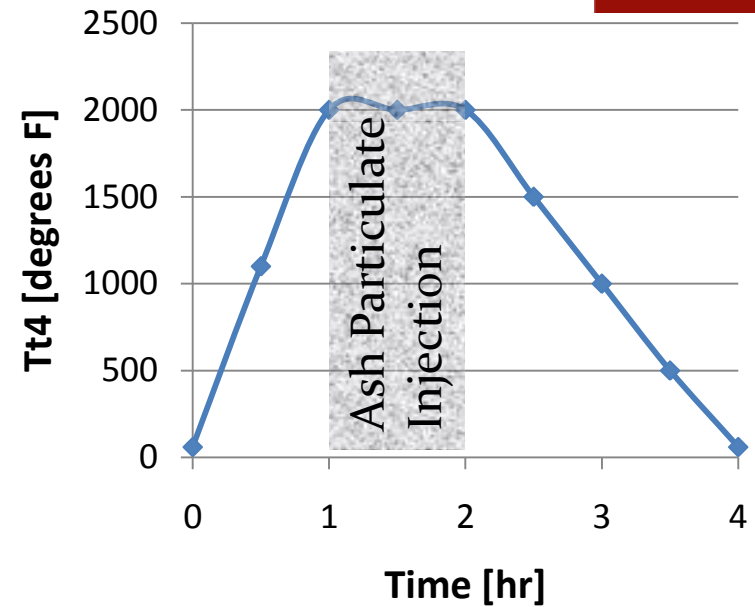
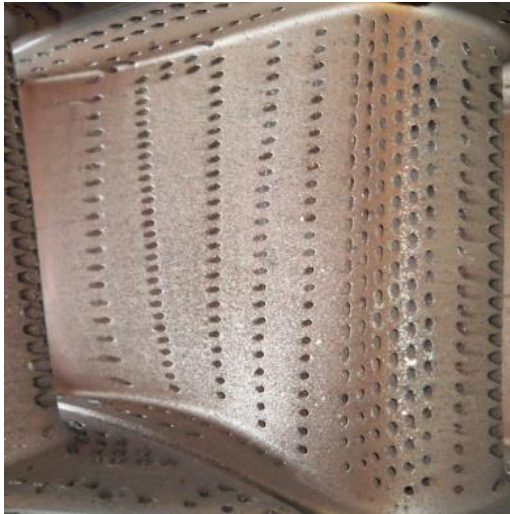


OSU - TuRFR



Typical TuRFR Test Profile

Before



After



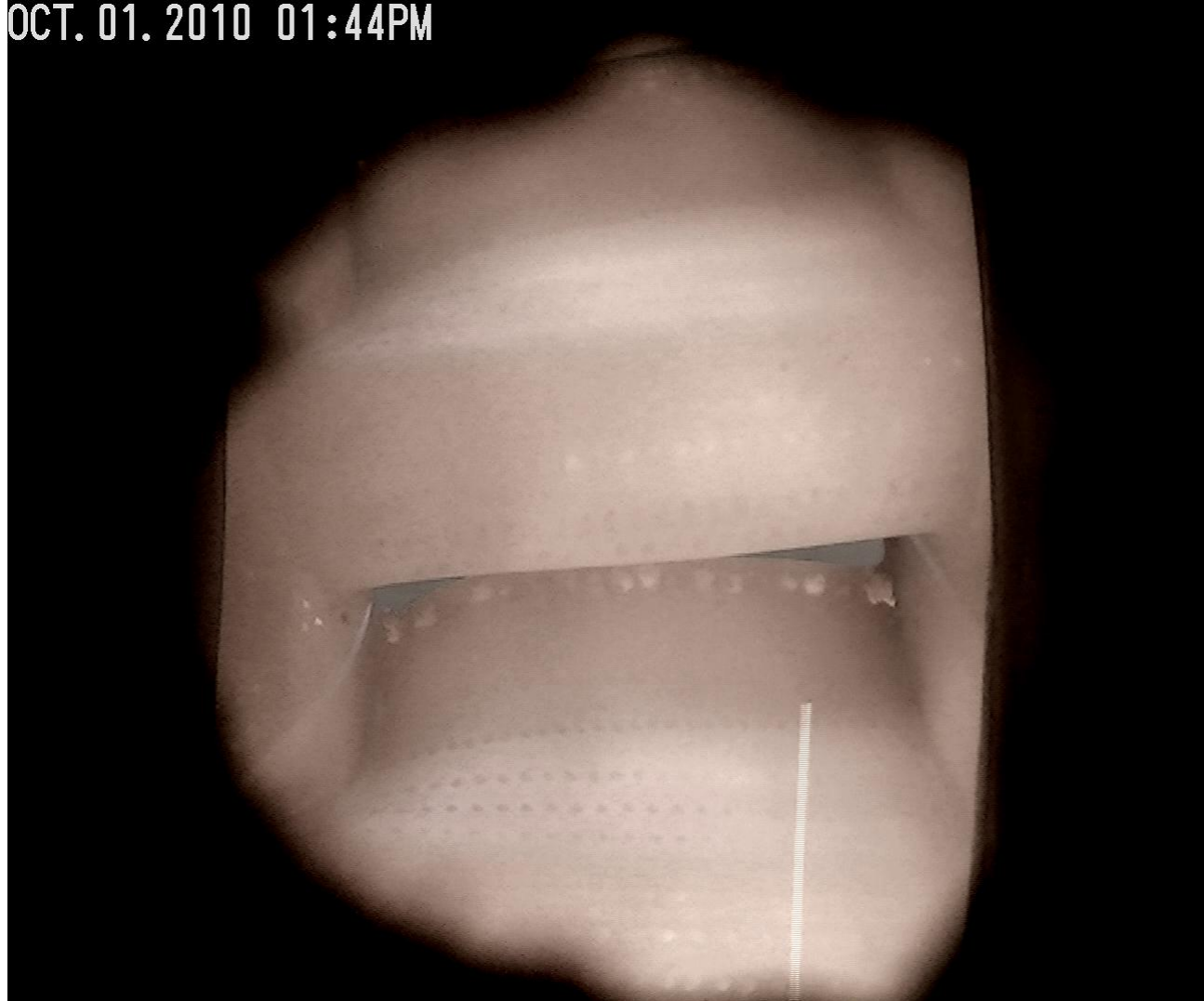
TuRFR Results



Typical Test Run

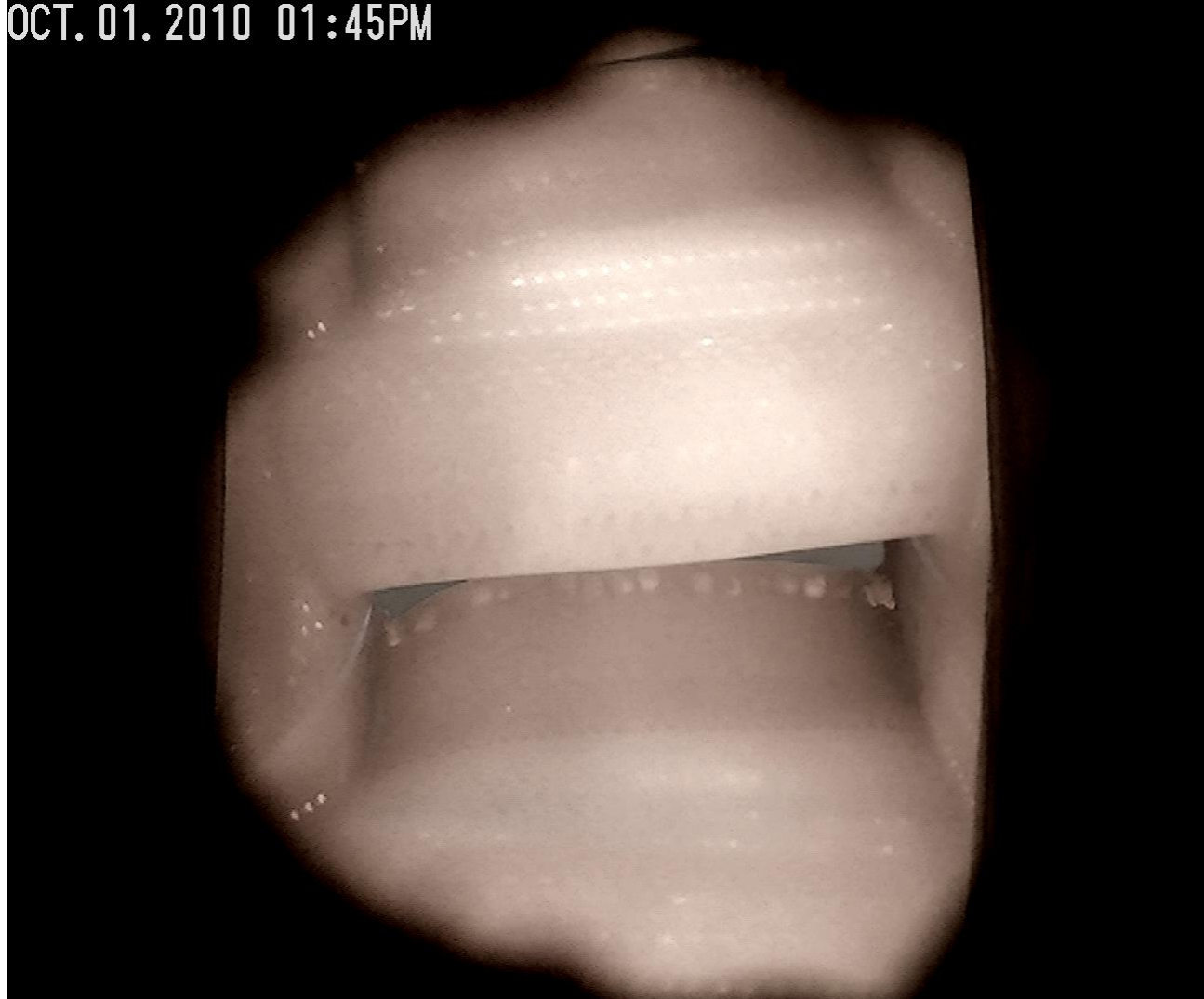


OCT. 01. 2010 01:44PM



Test on 10/1/10
Wyoming (Jim Bridger Power Station)
Sub-Bituminous Ash (from BYU)
Test Conditions: ~1900 F; $M=0.09$; No Cooling
 $t=0$ sec

OCT. 01. 2010 01:45PM



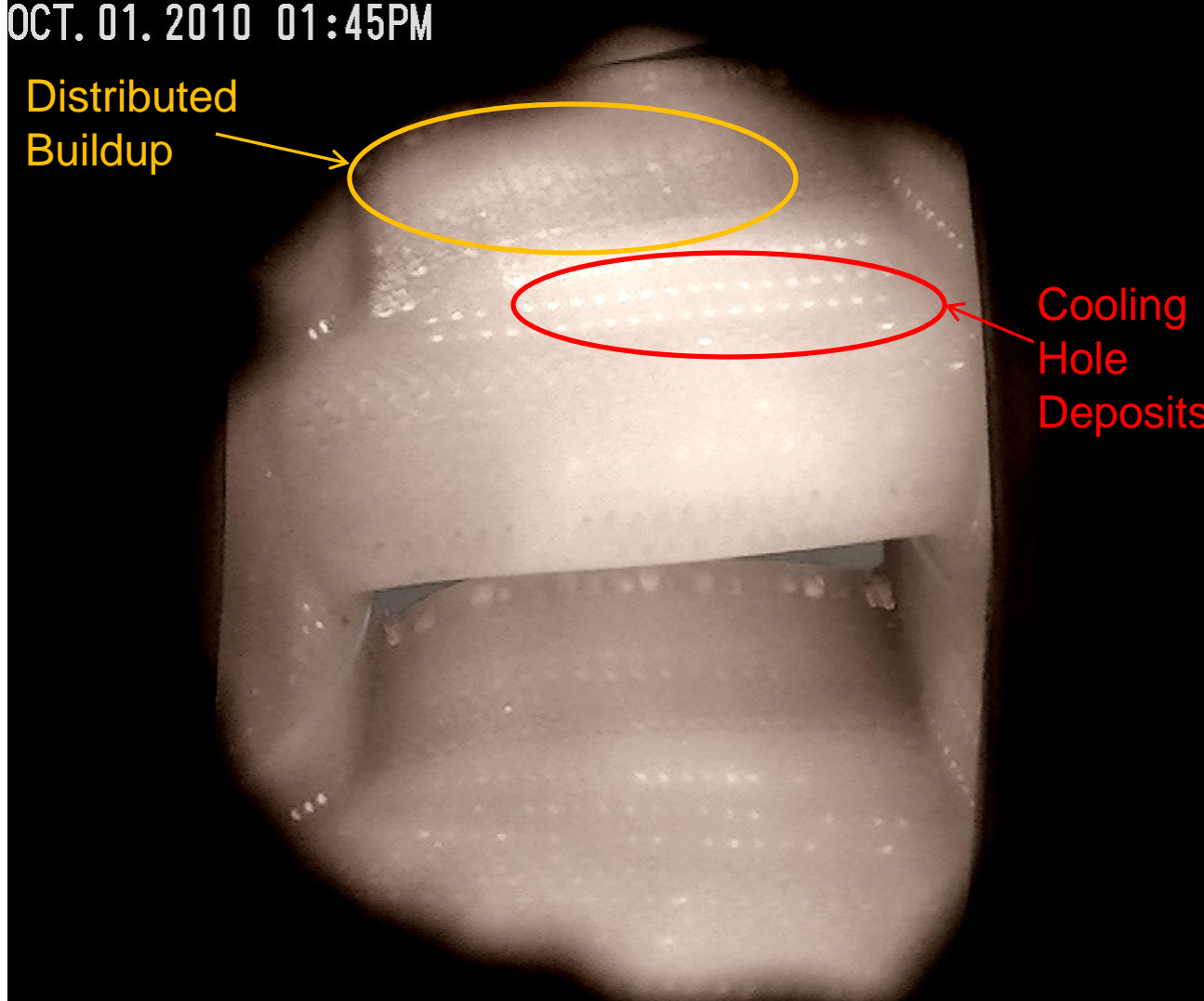
t= +30 sec

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Distributed
Buildup

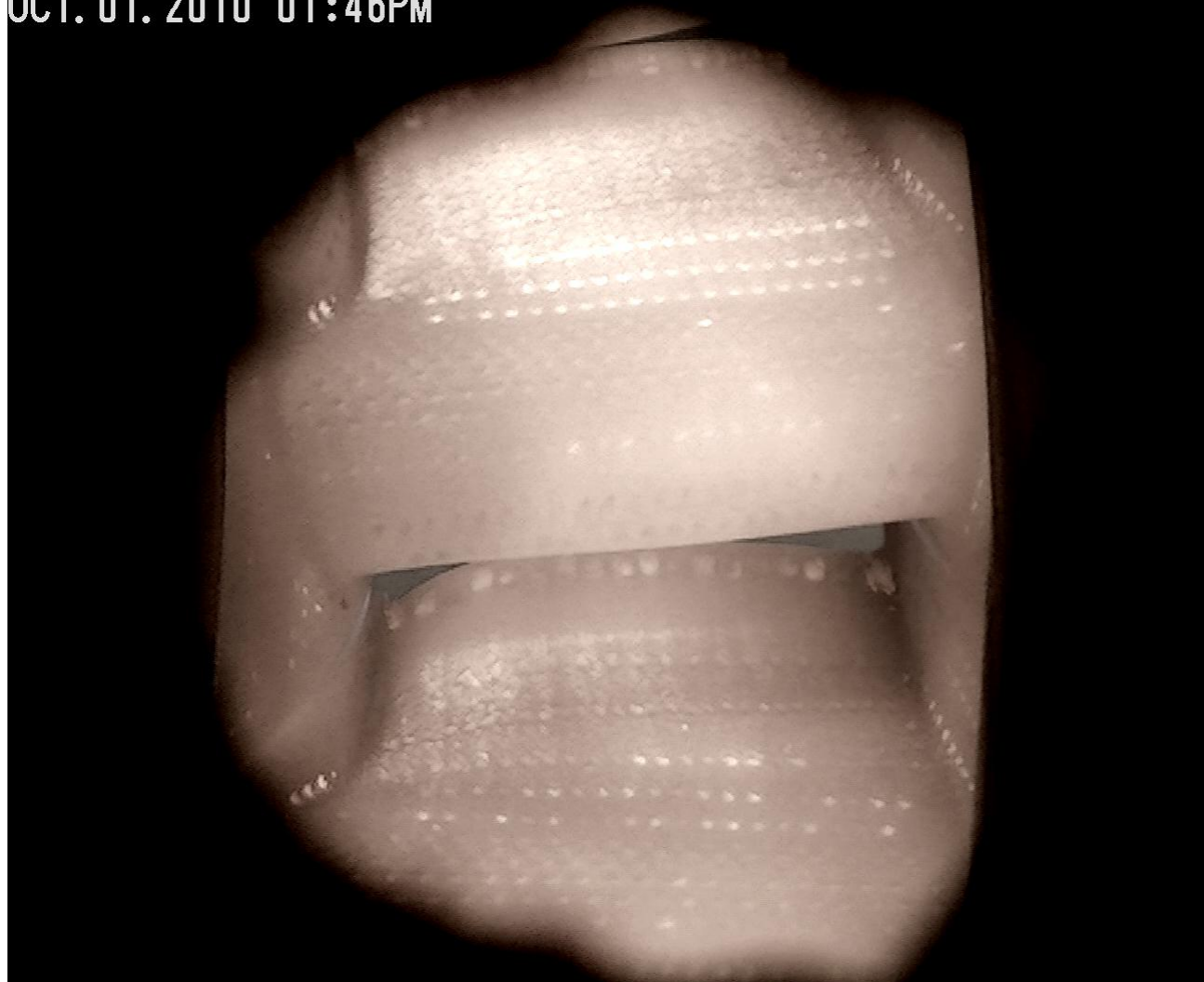


Cooling
Hole
Deposits



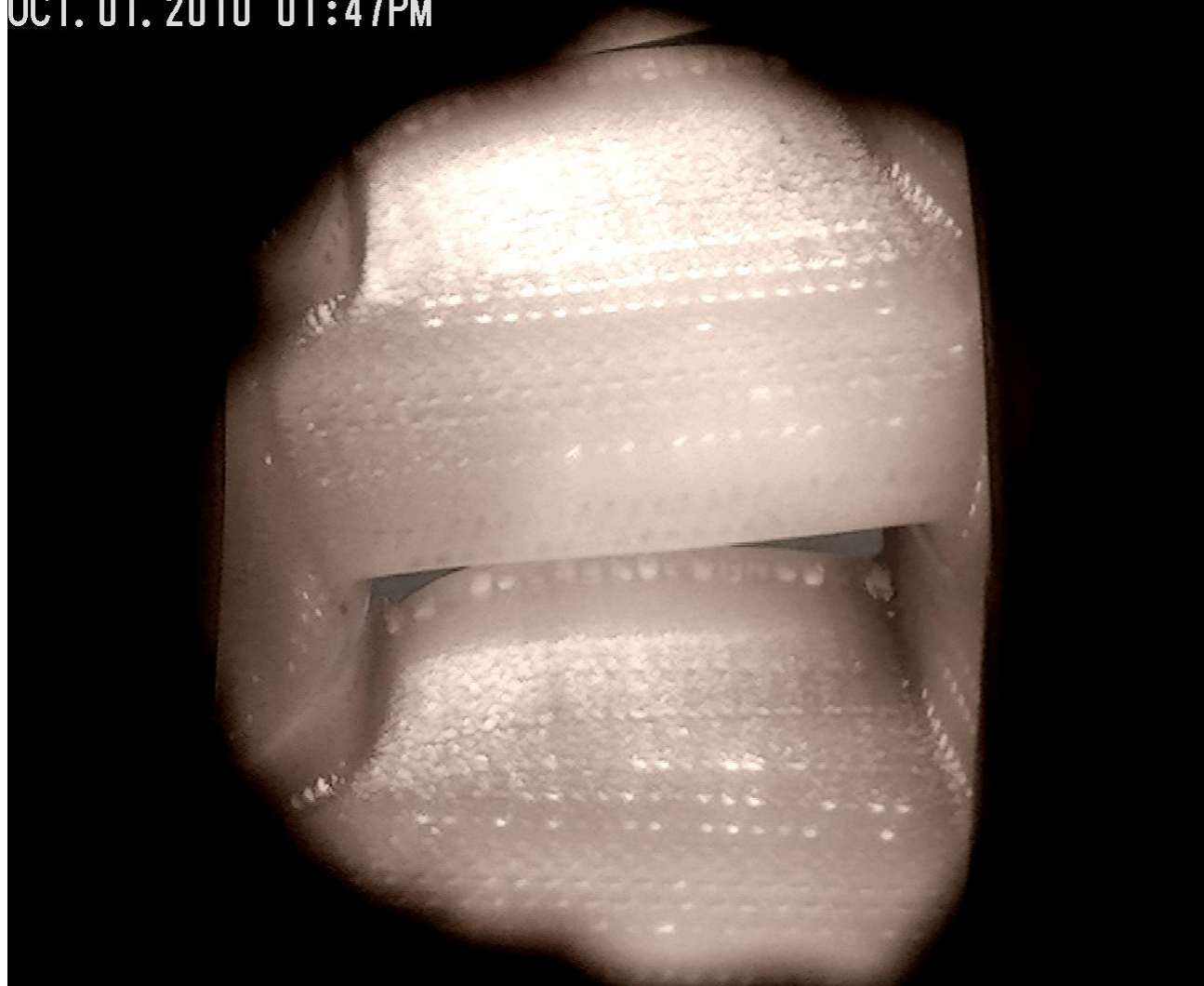
$t = +1$ min

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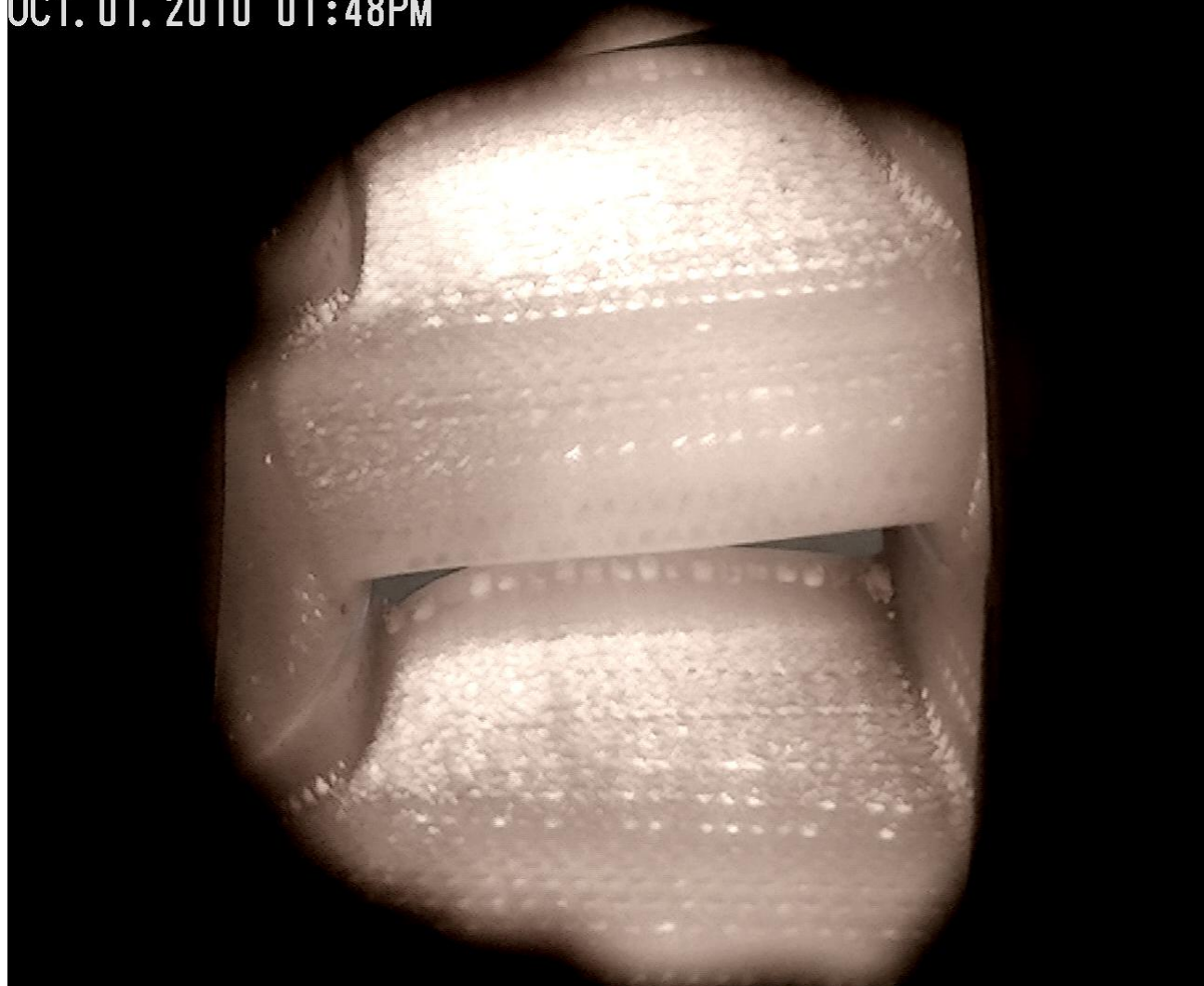
$t = +2 \text{ min}$

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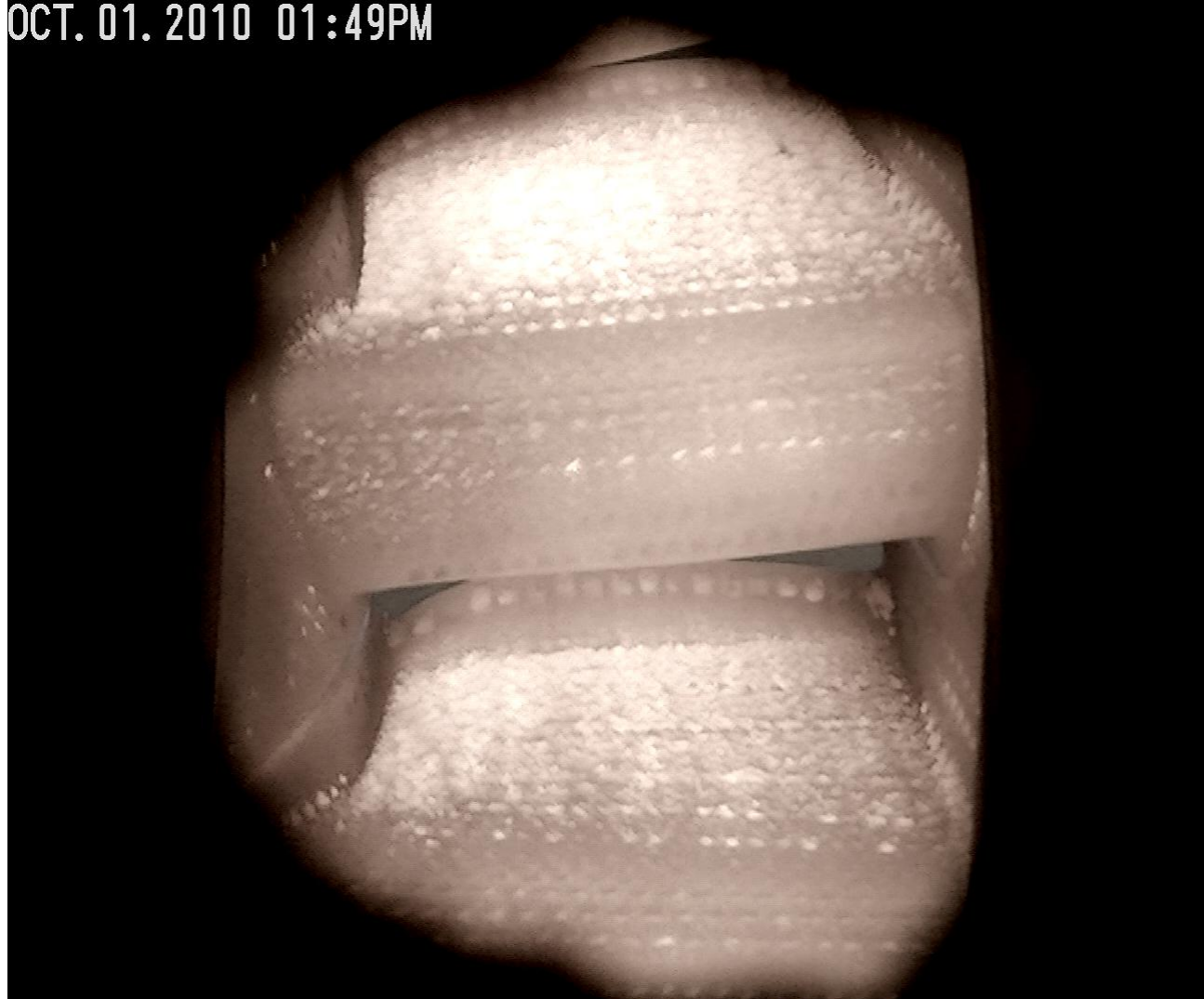
$t = +3 \text{ min}$

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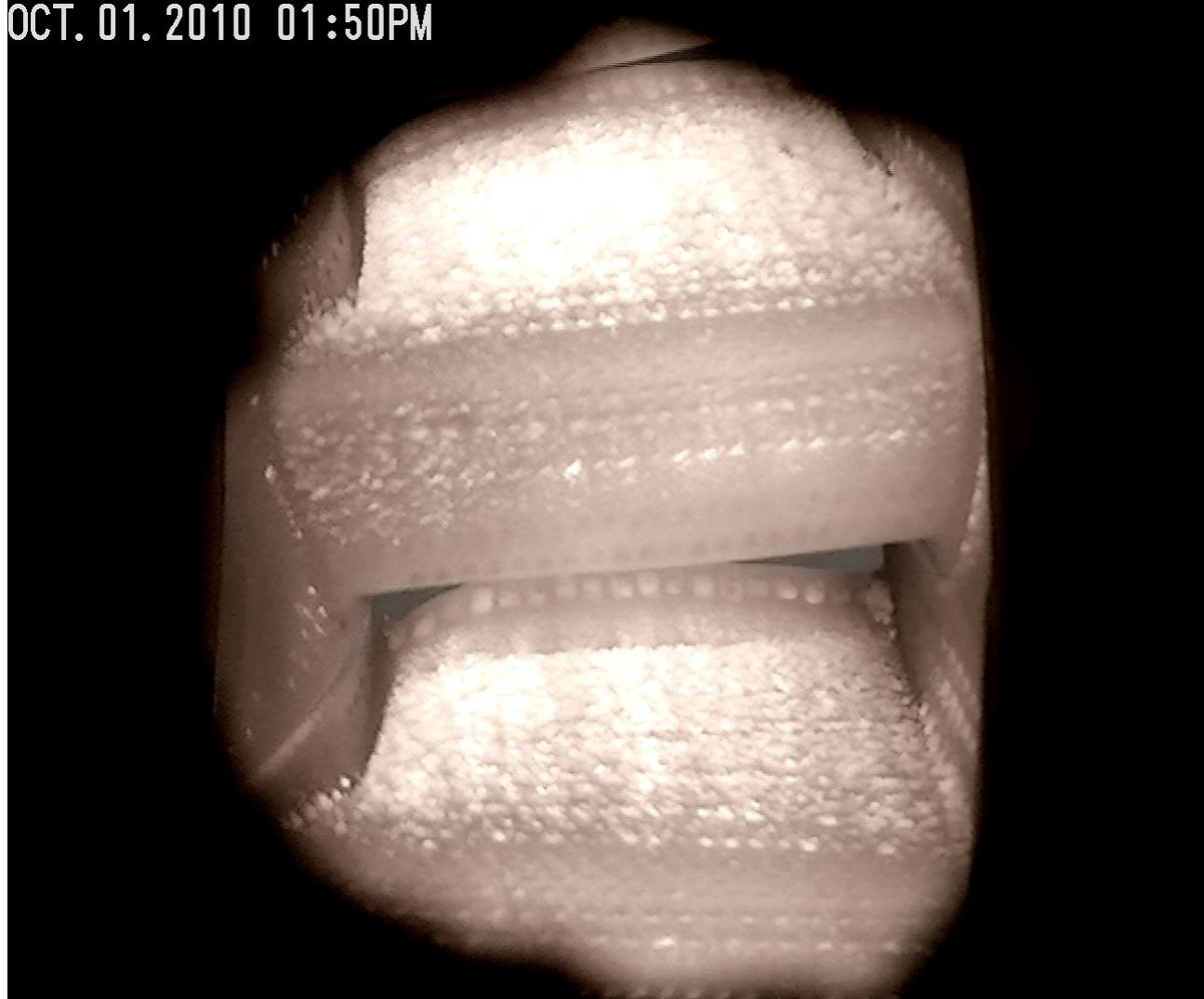
$t=+4$ min

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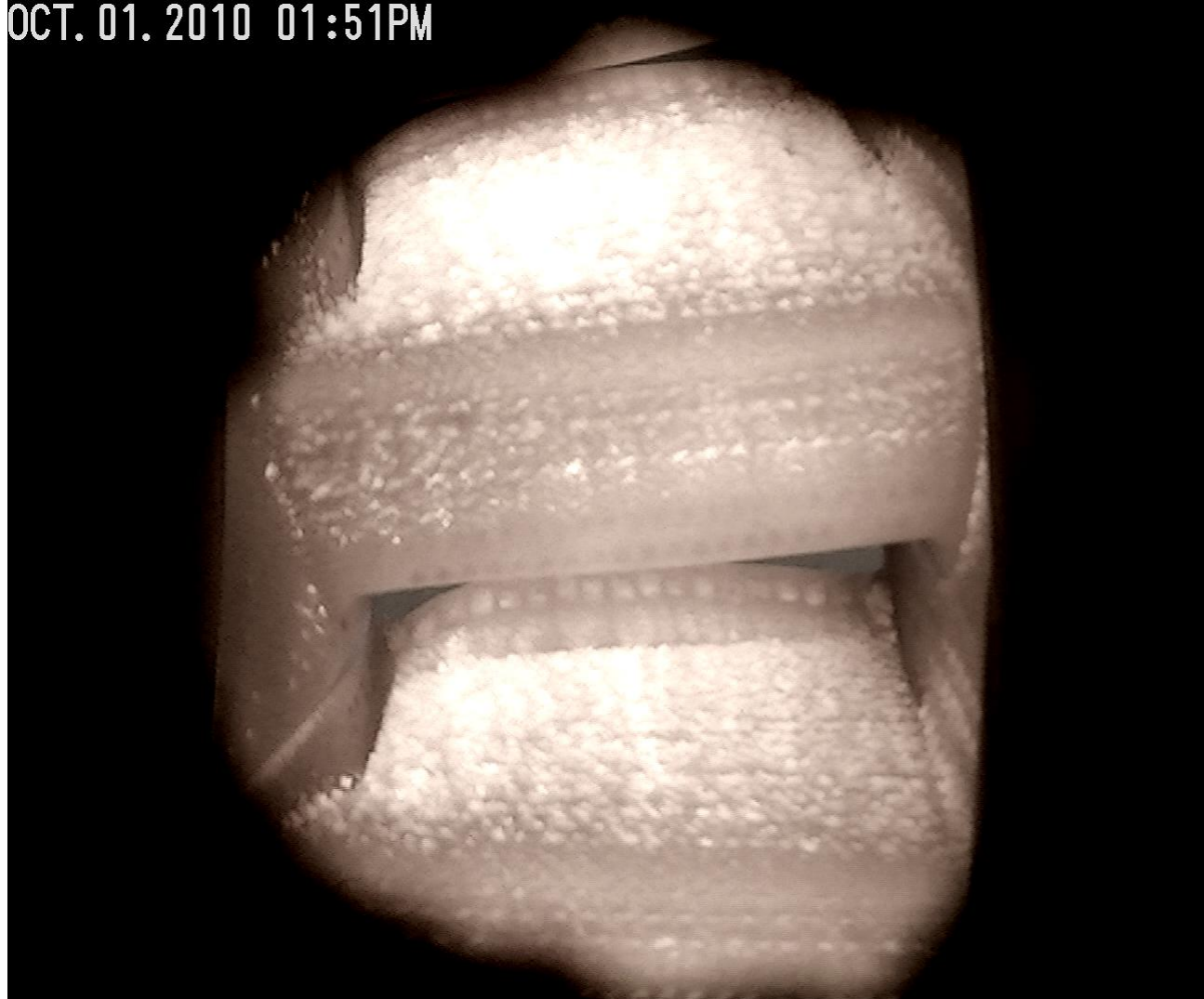
$t = +5 \text{ min}$

OCT. 01. 2010 01:50PM



$t=+6$ min

OCT. 01. 2010 01:51PM



$t=+7$ min

OCT. 01. 2010 01:52PM



$t=+8$ min

OCT. 01. 2010 01:53PM



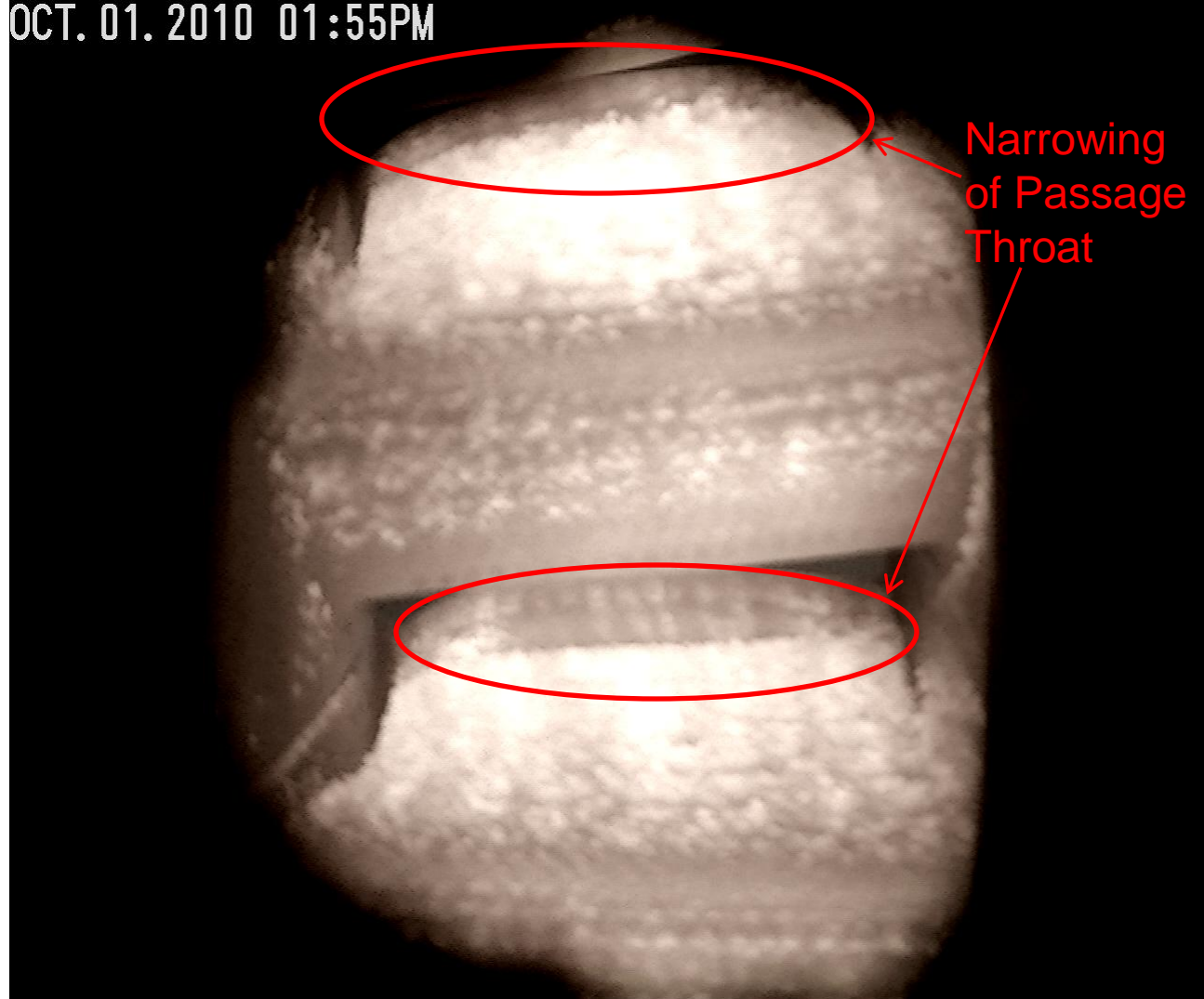
$t=+9$ min

OCT. 01. 2010 01:54PM



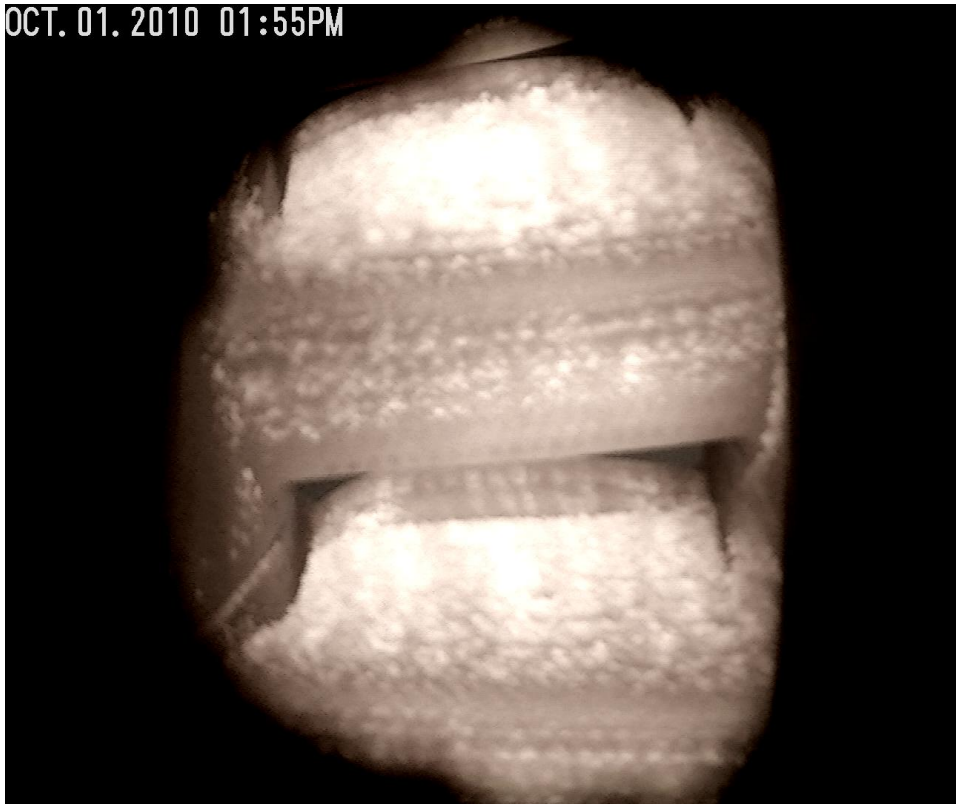
$t=+10$ min

OCT. 01. 2010 01:55PM



$t=+11$ min

OCT. 01. 2010 01:55PM



Test on 10/1/10
Wyoming (Jim Bridger Power Station)
Sub-Bituminous Ash (from BYU)
Test Conditions: ~1900 F; $M=0.09$;
No Film Cooling

Final Deposit Image



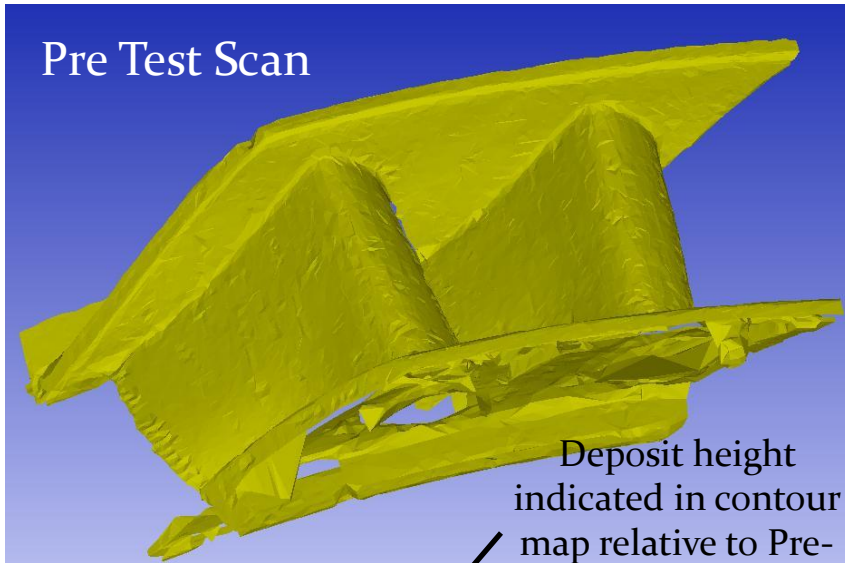
OBSERVATIONS

- Deposit builds from mid-chord on pressure surface forward to leading edge.
- Film cooling holes are sites for deposition initiation (whether or not they are blowing)
- Narrowing of passage throat

Post Test Diagnostics

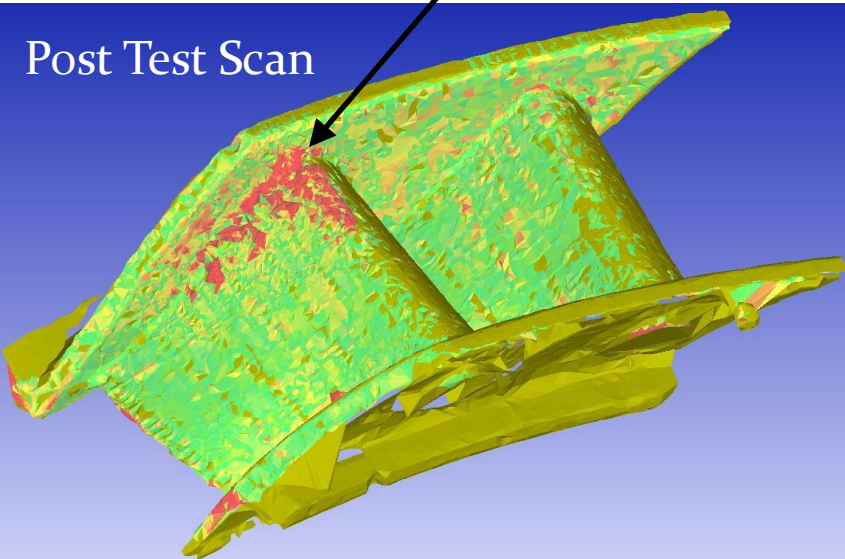
Surface Metrology

Pre Test Scan

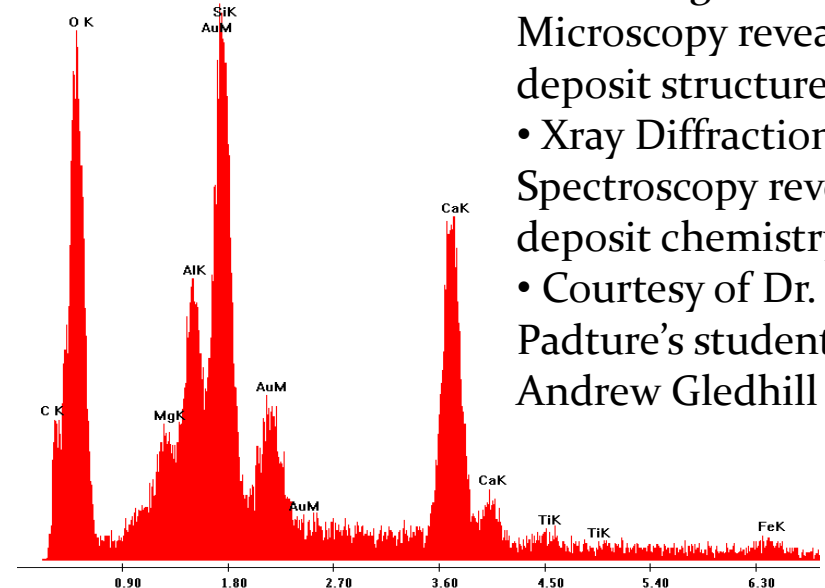
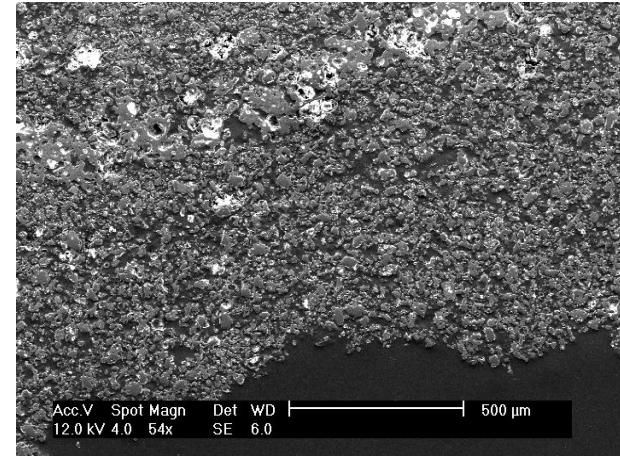


Deposit height
indicated in contour
map relative to Pre-
Test Datum

Post Test Scan



Deposit Microscopy



- Scanning Electron Microscopy reveals deposit structure.
- Xray Diffraction Spectroscopy reveals deposit chemistry.
- Courtesy of Dr. Padture's student, Andrew Gledhill

TuRFR Results

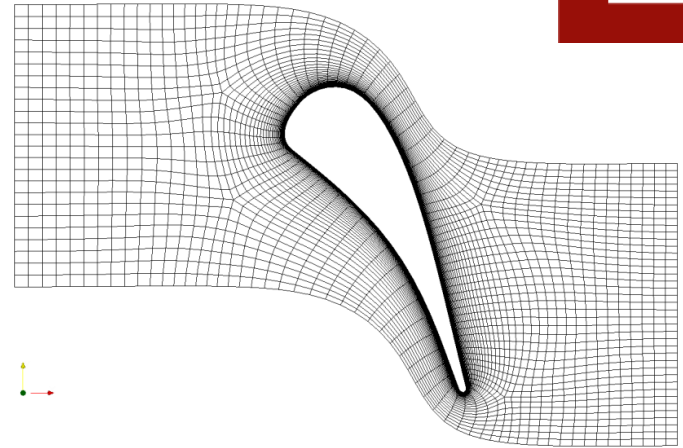
OBSERVATIONS of DEPOSITION MECHANICS

- Suction surface deposit free
- Deposit builds from mid-chord on pressure surface forward to leading edge.
- Deposition is sensitive to ash type and size
- Film cooling holes are sites for deposition initiation (whether or not they are blowing)
- Large deposits are sloughed off the surface during and after testing (e.g. during cooldown)

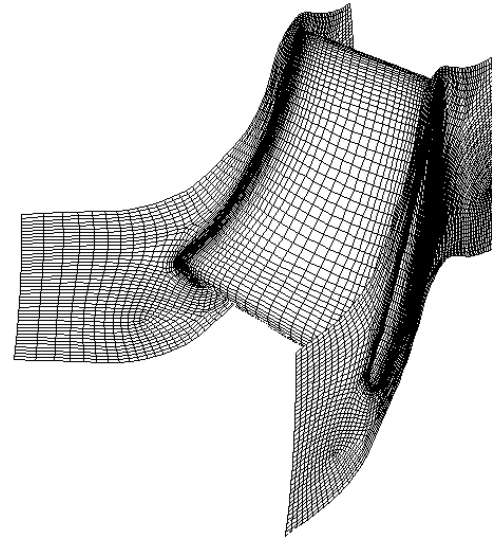
Ash Deposition Modeling

Computational Model

- Flow solution using FLUENT
 - Commercially available
 - Solves discretized flow equations to predict fluid dynamics
 - $k-\omega$ turbulence model
- Deposition Models
 - developed in C language and incorporated as User-Defined Functions in Fluent
- Turbine grid made using GridPro
 - VKI Turbine Vane
 - GE-E³ Turbine Vane



VKI Turbine Vane (2D)



E³ Turbine Vane (3D)

Computational Model

- Two sticking models
 - Critical Viscosity Model (Tafti et al. 2010)

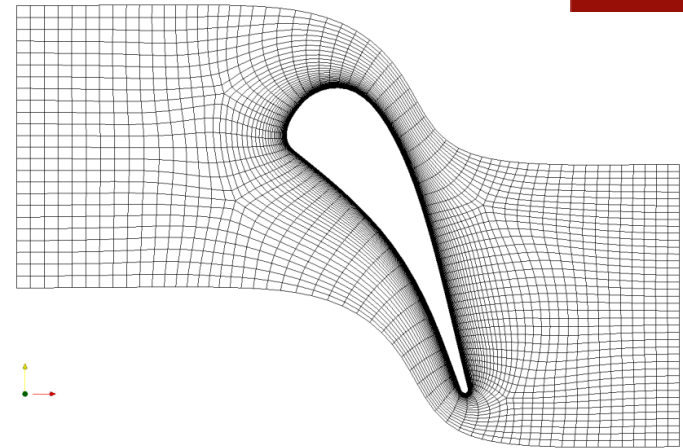
- Sticking probability based on viscosity of particle

$$P_S(T_P) = \frac{\mu_{Crit}}{\mu_{Tp}}$$

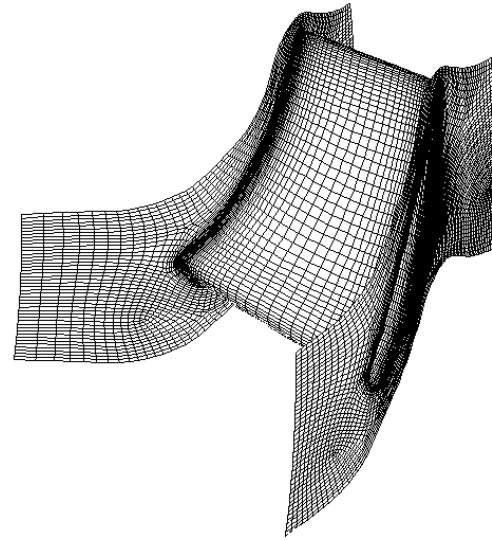
- Critical Velocity Model

(El Batsh, Haselbacher.2002)

- Particles stick if $V_N < V_{CRIT}$
 - Particles rebound if $V_N > V_{CRIT}$
 - $V_{CRIT} = f(T, d, \dots)$

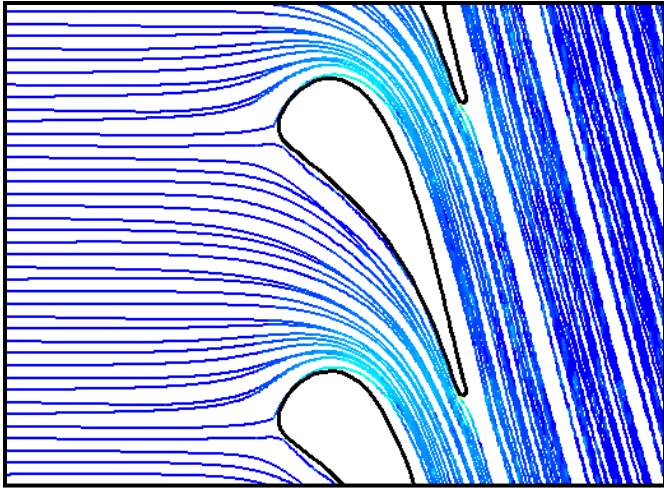


VKI Turbine Vane (2D)

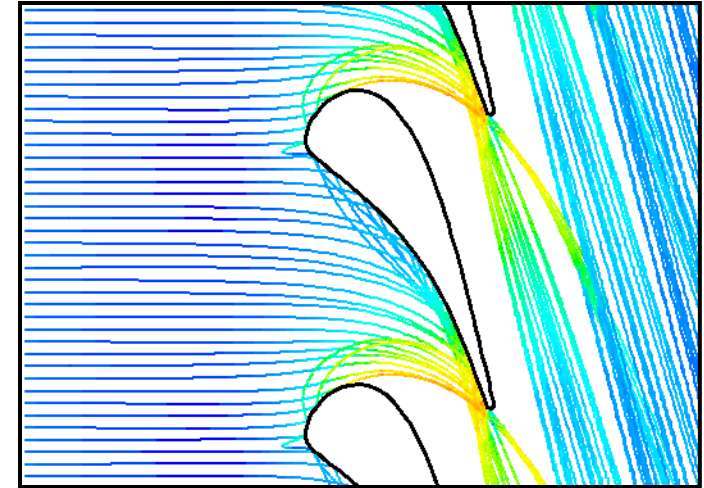


E³ Turbine Vane (3D)

Particle Trajectories

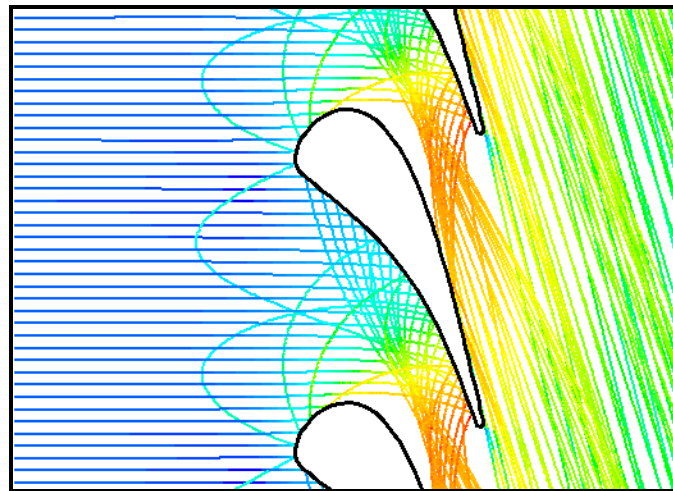


1 μm particles
($St_k = 0.01$)



10 μm particles
($St_k = 1.0$)

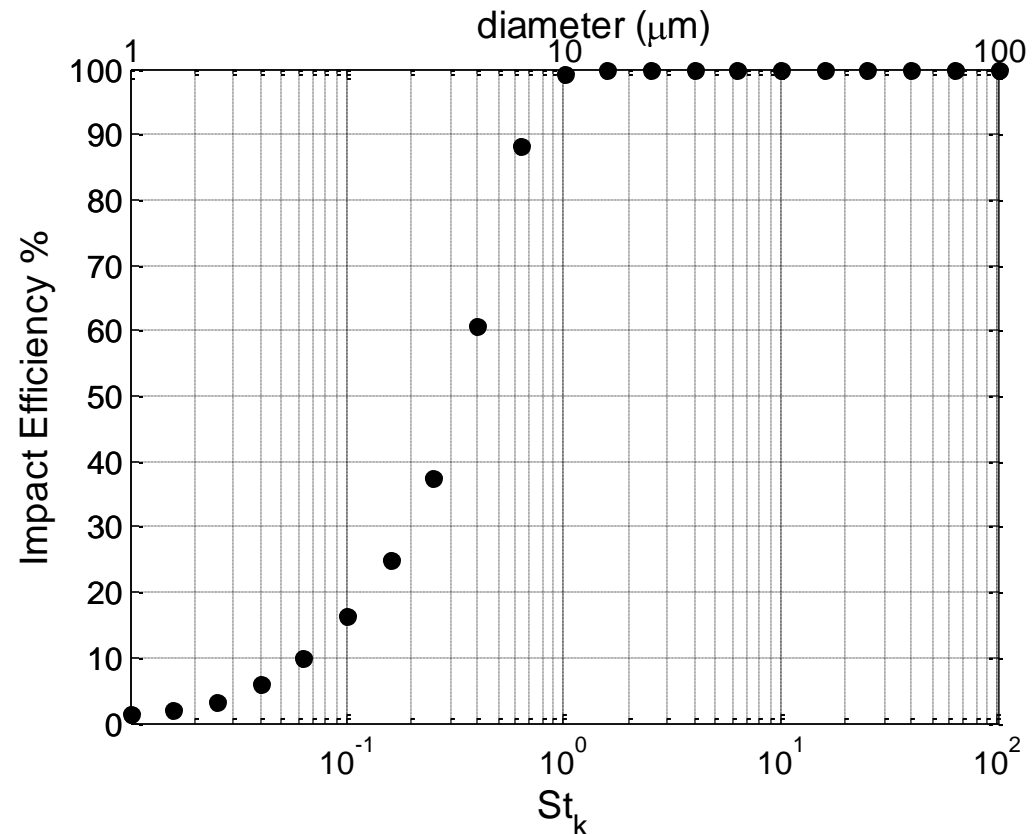
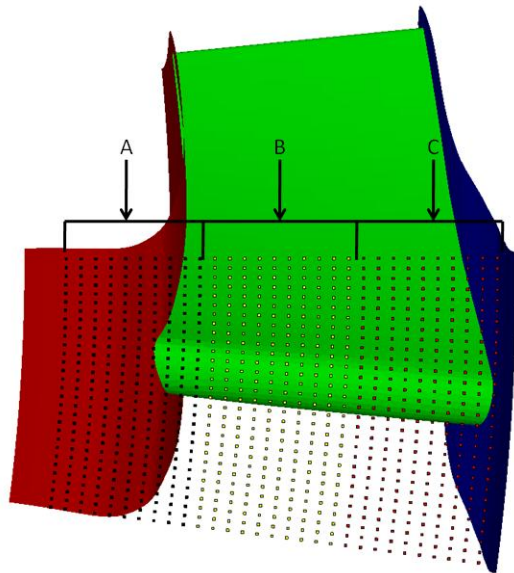
$$St_k = \frac{\rho_p d_p^2 V_i}{18 \mu l_c}$$



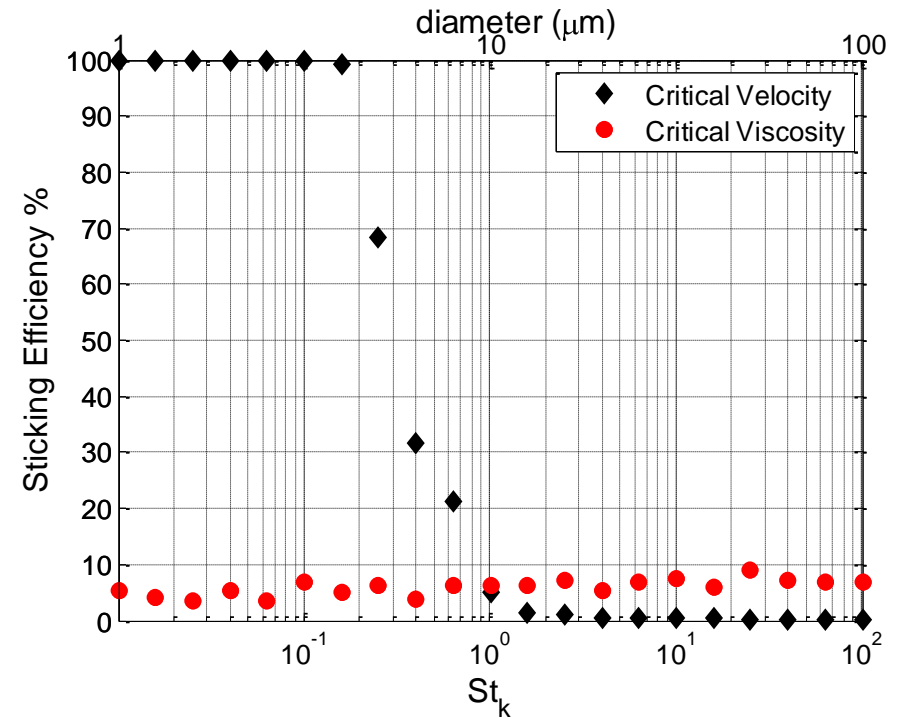
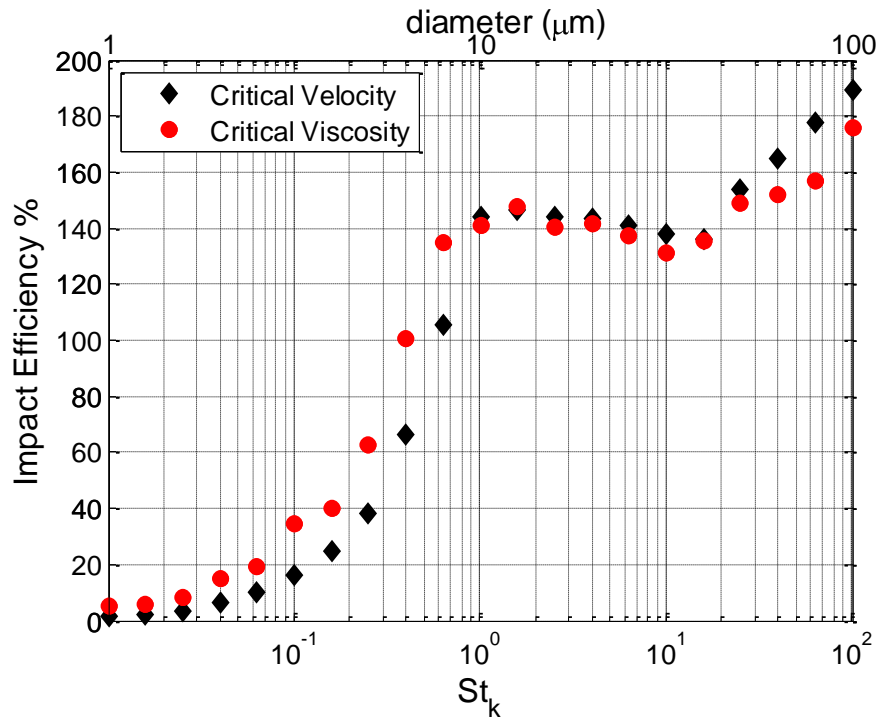
50 μm particles
($St_k = 25$)

Simulation Results-Particle Size Effect

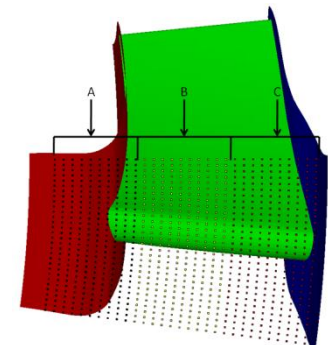
- Small particles less likely to impact surface
- Particles larger than 10 μm ($St_k = 1$) nearly 100% likely to impact surface
- Particles not allowed to rebound or impact multiple times



Simulation Results-Particle Size Effect

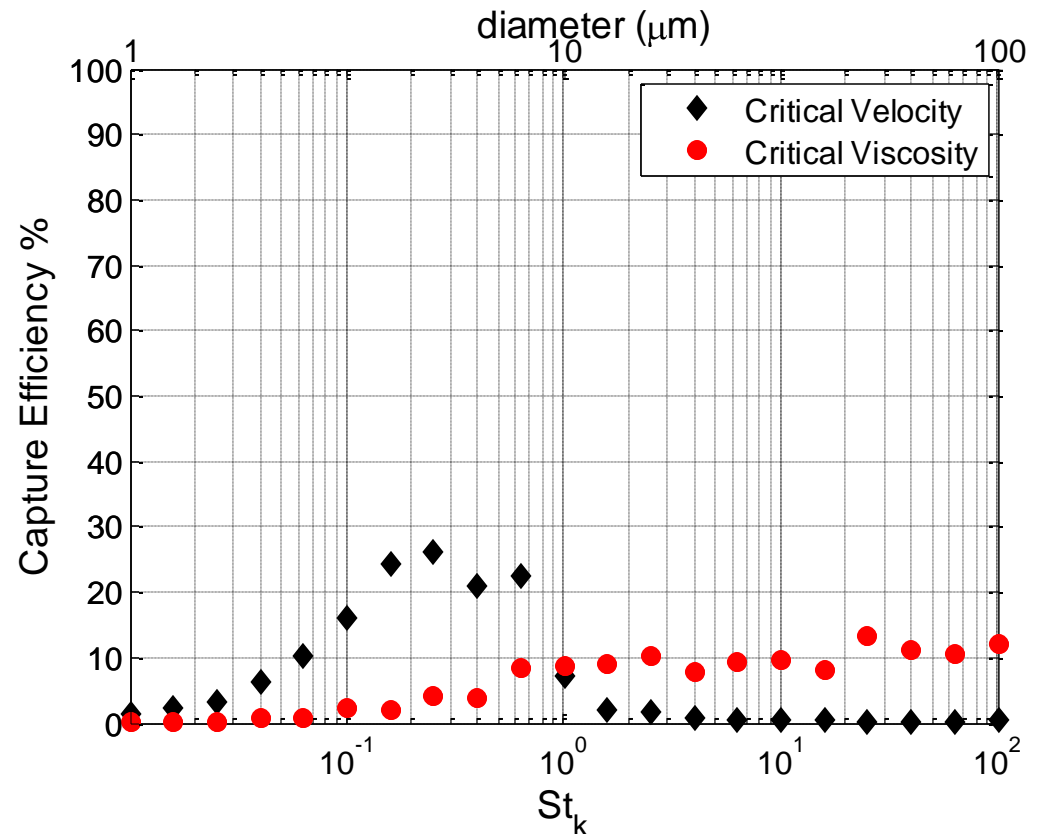
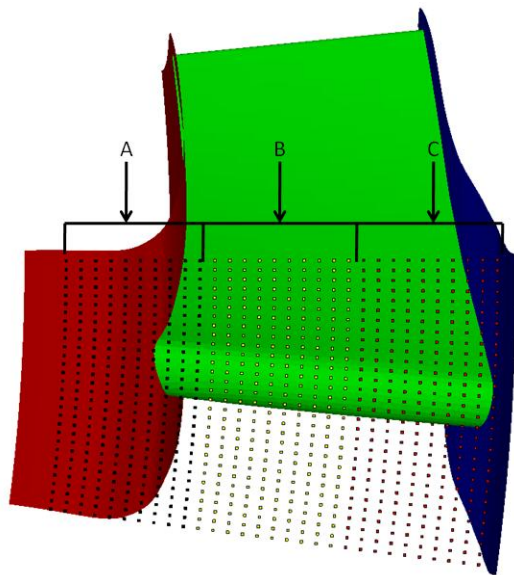


- Multiple impacts allowed
- Impact efficiencies similar for both models
- Large particles impact surface multiple times
- Sticking efficiencies very different for each model



Simulation Results-Particle Size Effect

- Critical Velocity model predicts range of particles likely to stick (St_k from 0.05 to 1.0)
- Critical Viscosity model predicts larger particles more likely to stick



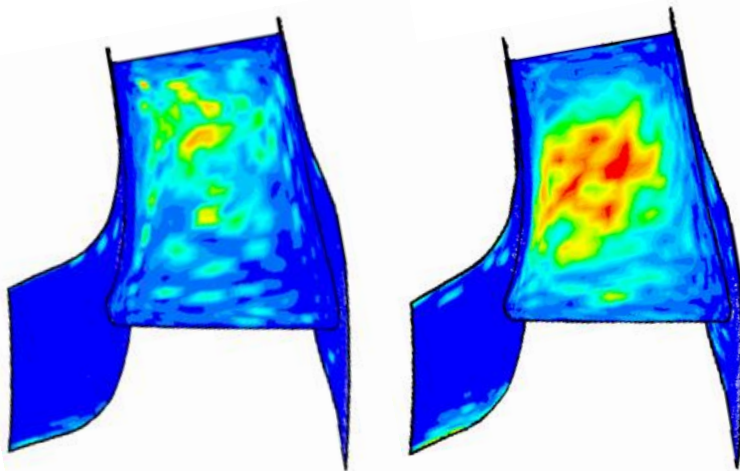
Simulation Results

Comparison of sticking models- JBPS ash

	Total Injected #	Impact Efficiency %	Sticking Efficiency %	Capture Efficiency %
Critical Velocity Model	18900	112	10.6	12.0
Critical Viscosity Model	18900	124	6.2	7.7
Experimental	-	-	-	~20.0

Experimental

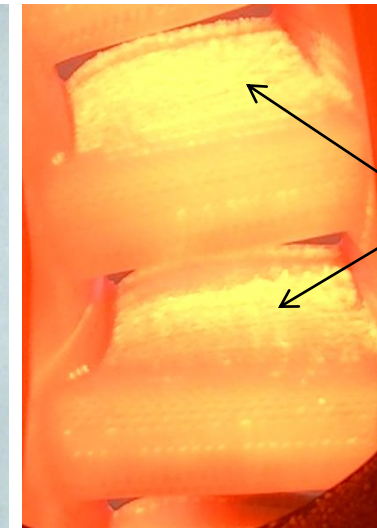
Critical Viscosity Critical Velocity



Post test



Initial Deposit



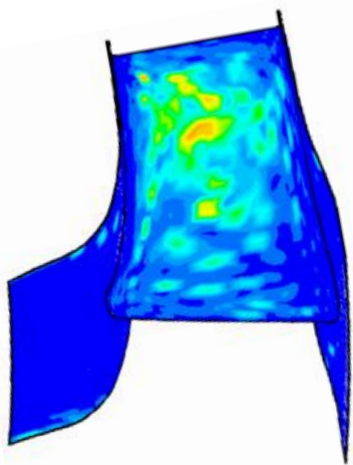
Deposit

Simulation Results

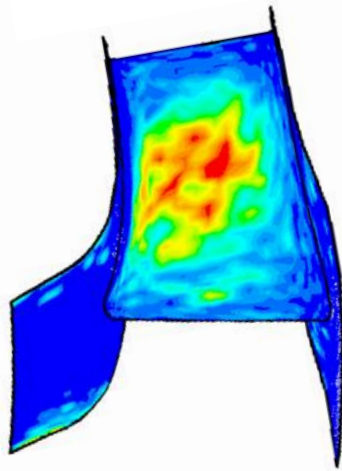
Comparison of sticking models- JBPS ash

- Both models predict deposition on pressure surface at mid chord
- Both models underestimate deposition, especially viscosity model (with defined sticking temperature)
- Model is most accurate at predicting initial deposition

Critical Viscosity



Critical Velocity

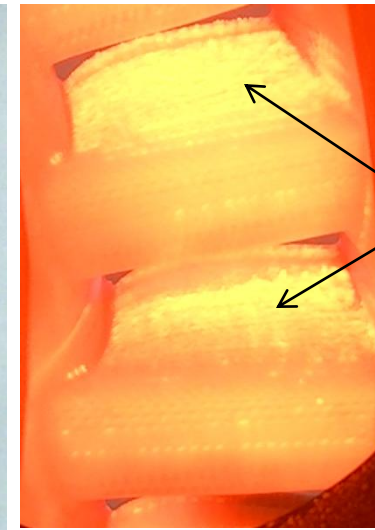


Experimental

Post test



Initial Deposit



Deposit

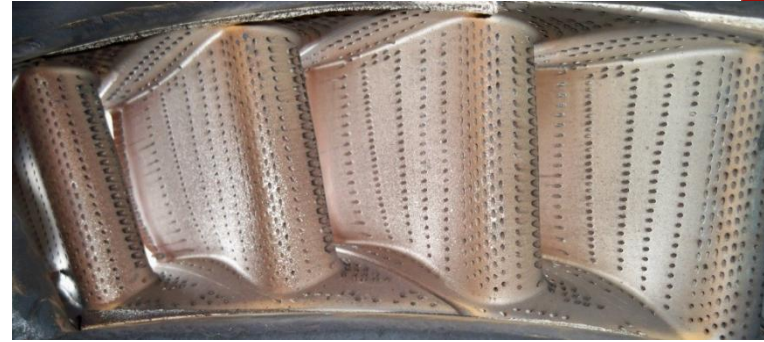
Experimental Results – Four Ash Types

Bituminous

JBPS Sub-bituminous

PRB Sub-bituminous

Lignite



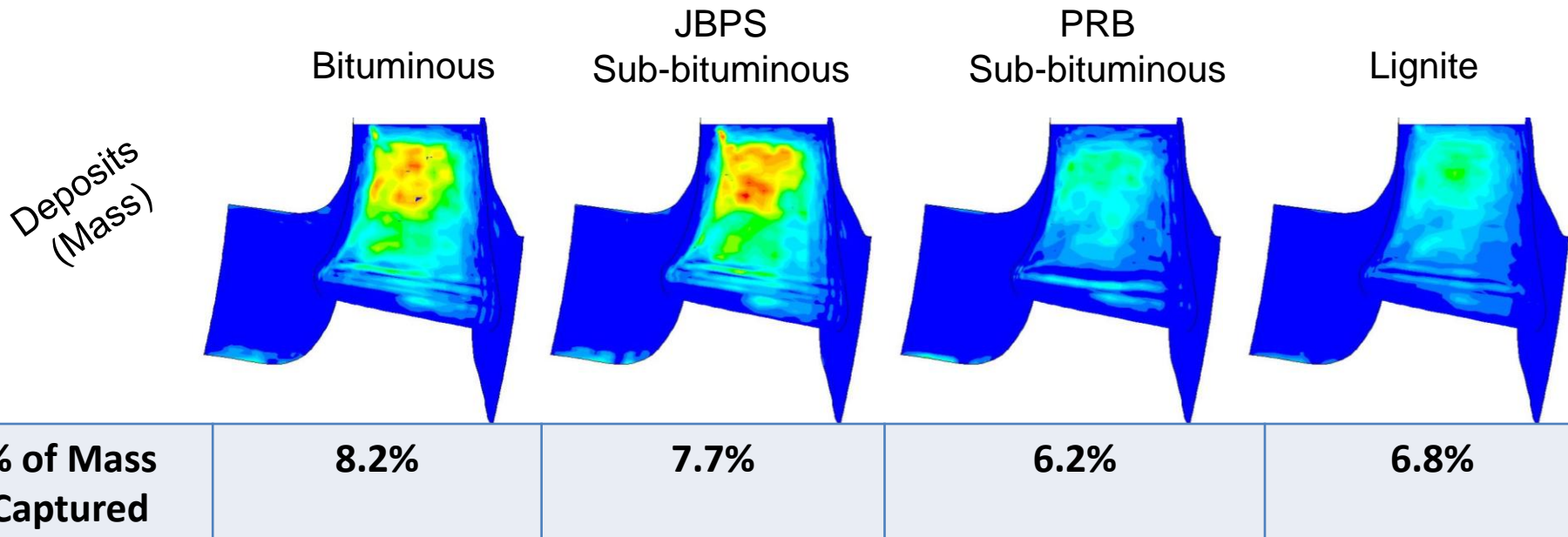
Little Deposition

Much Deposition

Much Deposition

Most Deposition

Deposit Results- Four Ash Types

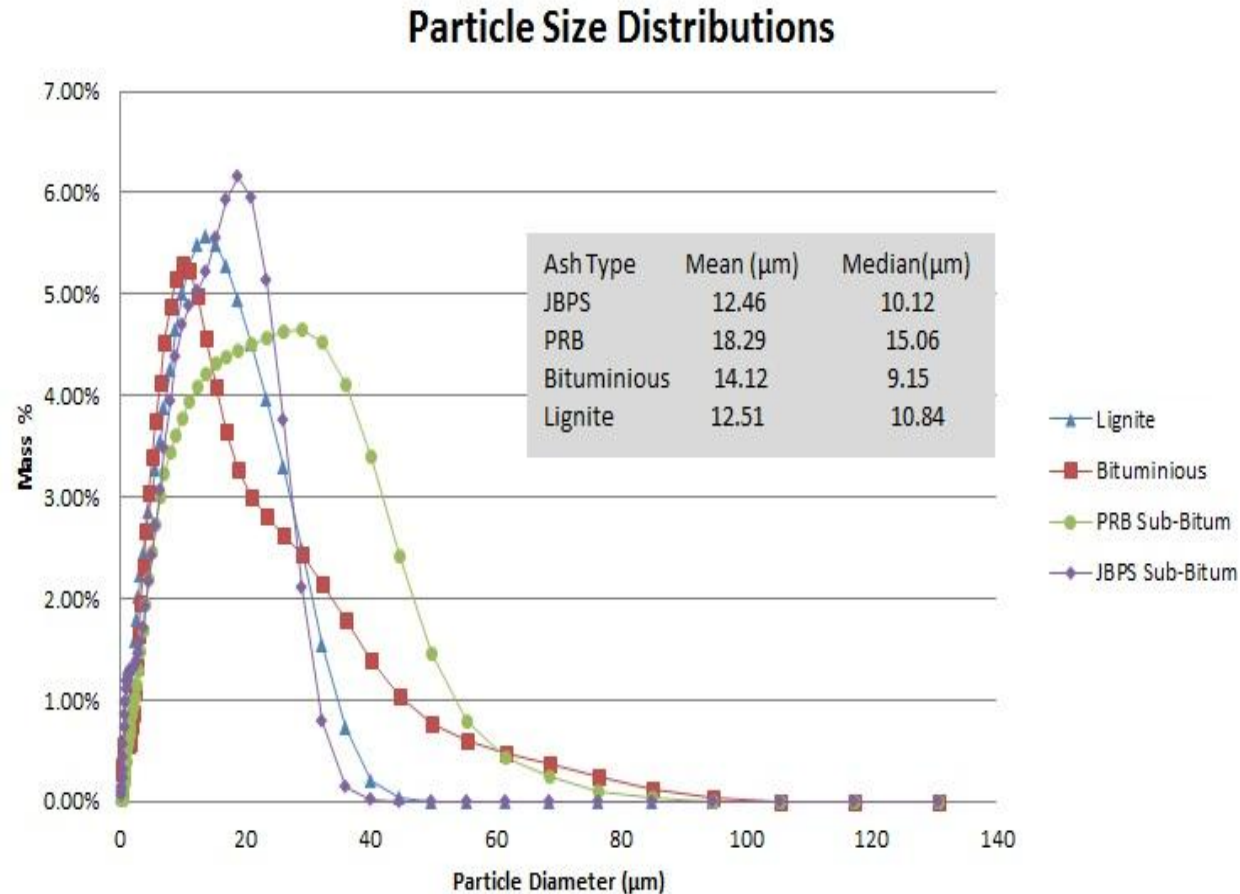


- Numerical simulations did not correspond well with experiments using critical viscosity model
- All predicted similar deposition patterns but with varied magnitudes
- Assumed same sticking temperature for all ash types

Ash Size Measurements

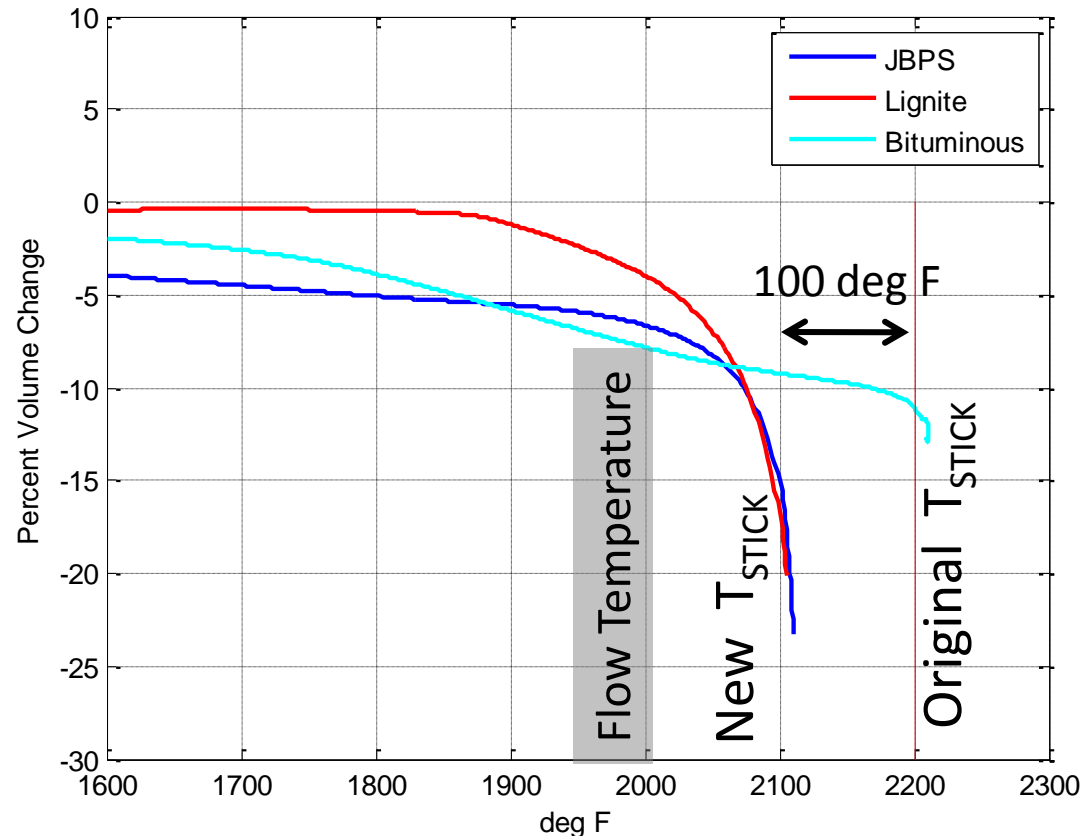
- Similar size distributions, with exception of PRB sub-bituminous
- Larger PRB distribution increases expected distribution in model.
- Verified ash densities

$$St_k = \frac{\rho_p d_p^2 V_i}{18 \mu l_c}$$



Ash Thermal Expansion Measurements

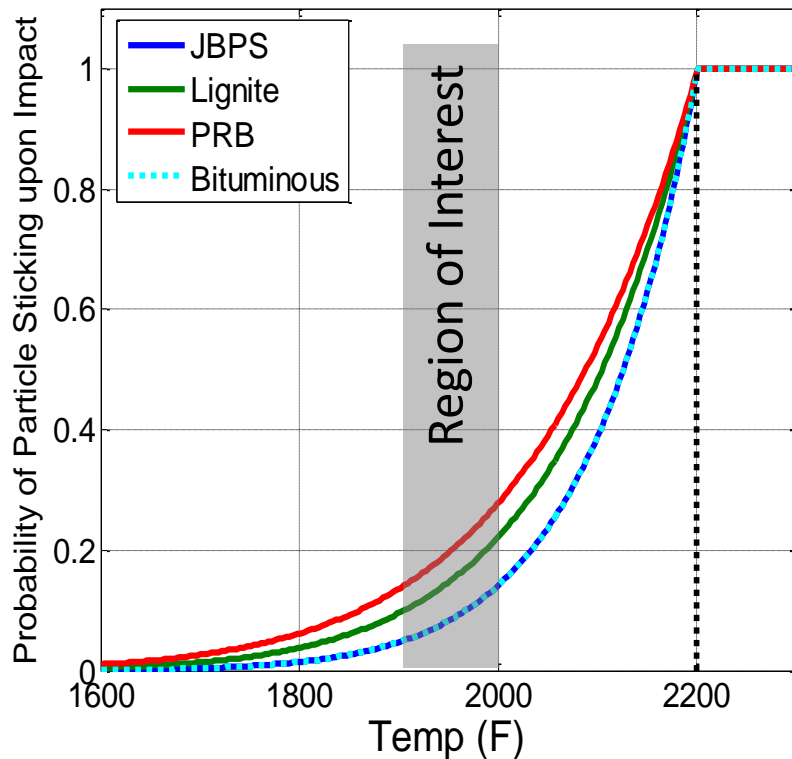
- Graph shows thermal expansion with increasing temperature. As ash particles sinter and melt, the volume of the ash decreases.
- JBPS and Lignite begin melting at 2100 F
- Bituminous begins melting at 2200 F
- These values are used as critical sticking temperatures
- PRB ash failed test



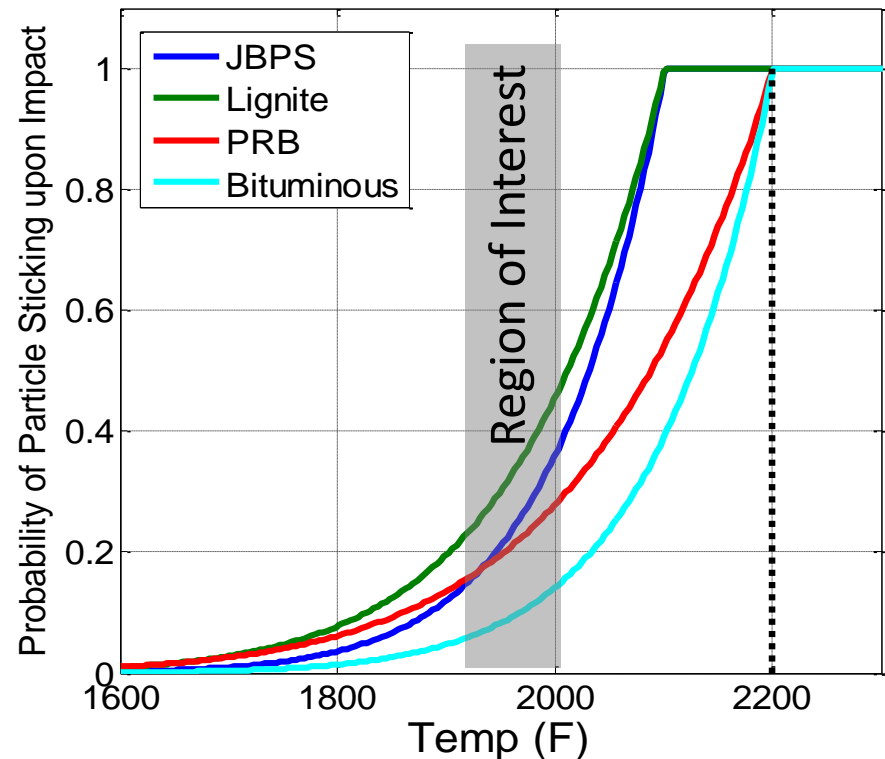
$$P_S(T_P) = \frac{\mu_{Crit}}{\mu_{Tp}}$$

Sensitivity to Ash Composition

- Sticking Probabilities for Critical Viscosity model
- Adjusted sticking temperature better agrees with experimental results

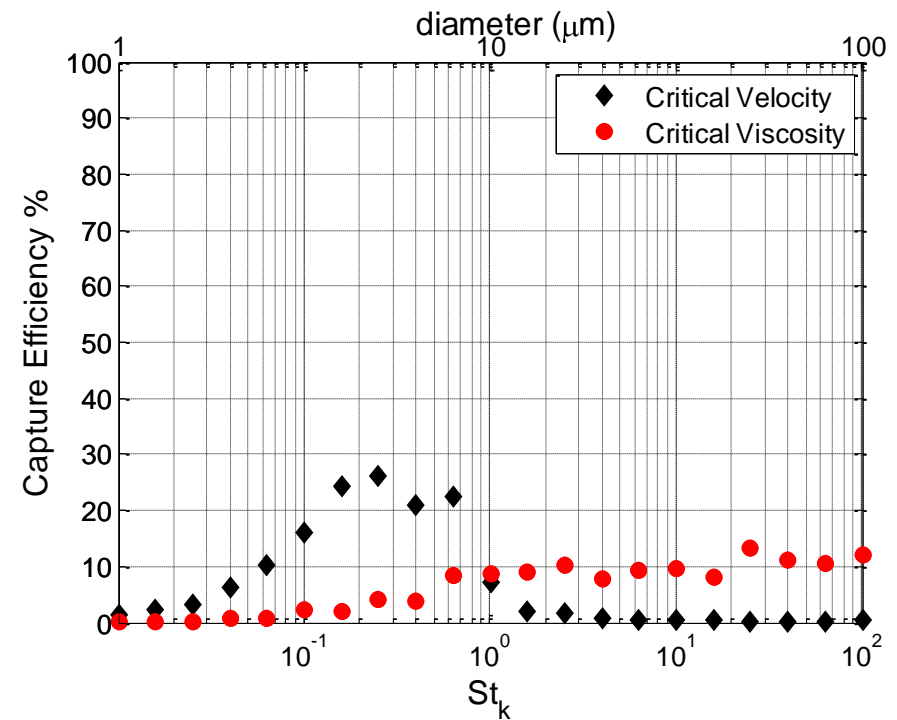
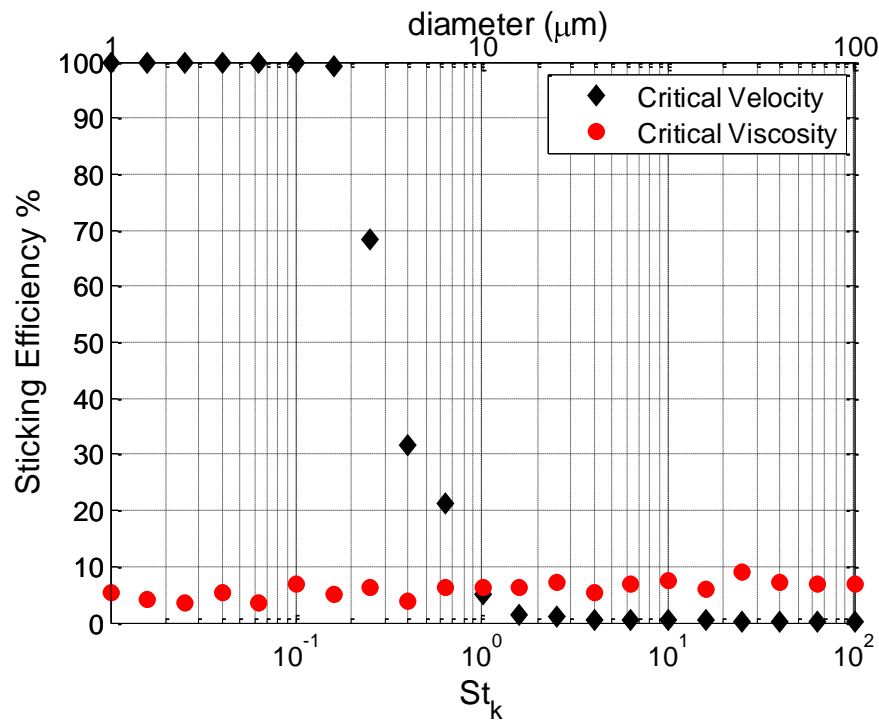


Old



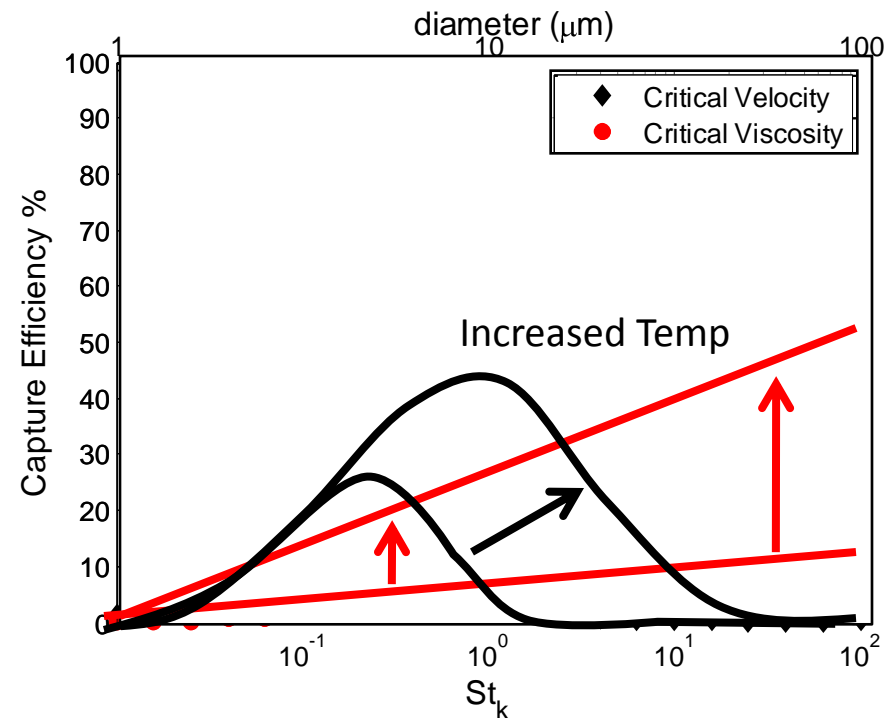
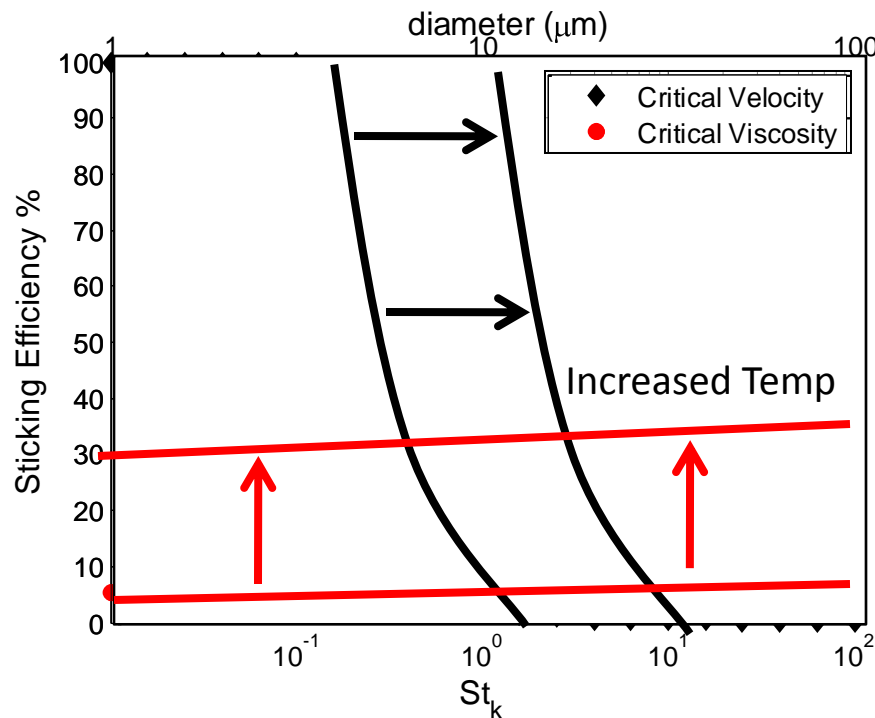
New

Sensitivity to Ash Composition



- Plots only valid for assumed $\mu(T)$ [critical viscosity] or $E(T)$ [critical velocity] relationship.

Simulation Results – Temperature Sensitivity

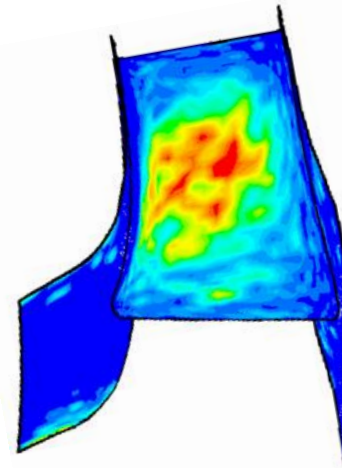
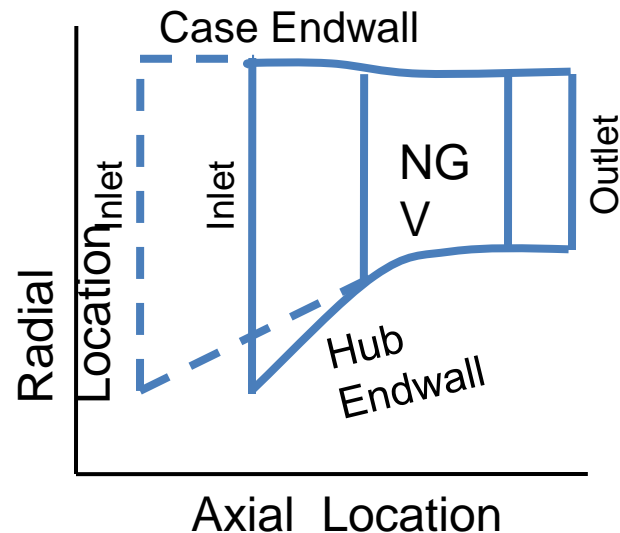


- Higher fluid temperatures, or lower critical sticking temperature, increases probability of sticking and deposit
- Viscosity model sensitive to critical sticking temperature
- Viscosity model more closely matches experiment

$$St_k = \frac{\rho_p d_p^2 V_i}{18 \mu l_c}$$

Endwall Modifications

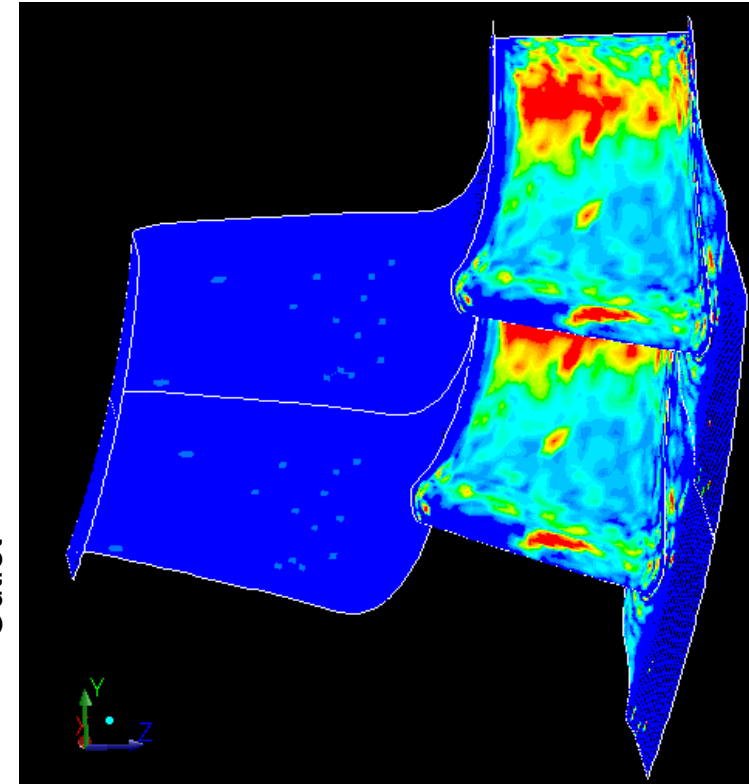
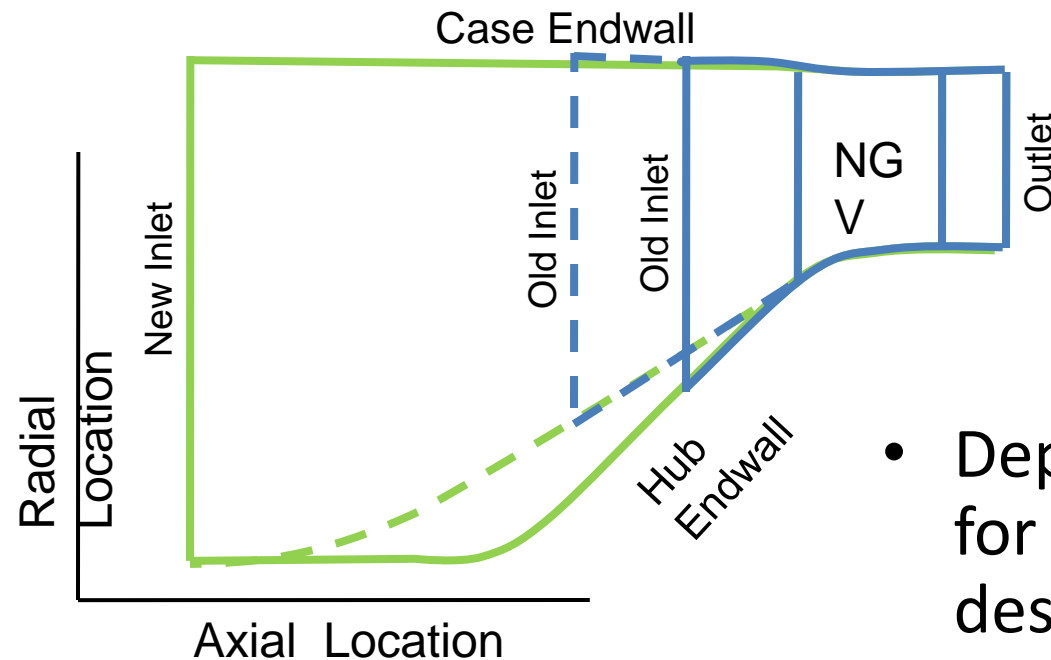
- Particles deposited preferentially toward hub endwall
- Hub endwall was redesigned with less severe inlet angle



- Results showed slight difference in deposition rates
- Difference likely due to extended inlet boundary condition affecting passage flow fields, rather than endwall modification

Endwall Modifications

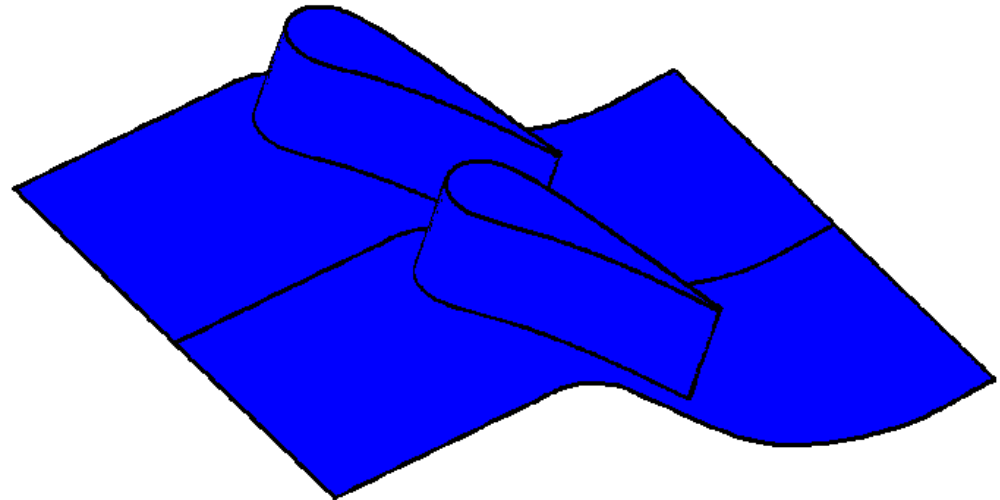
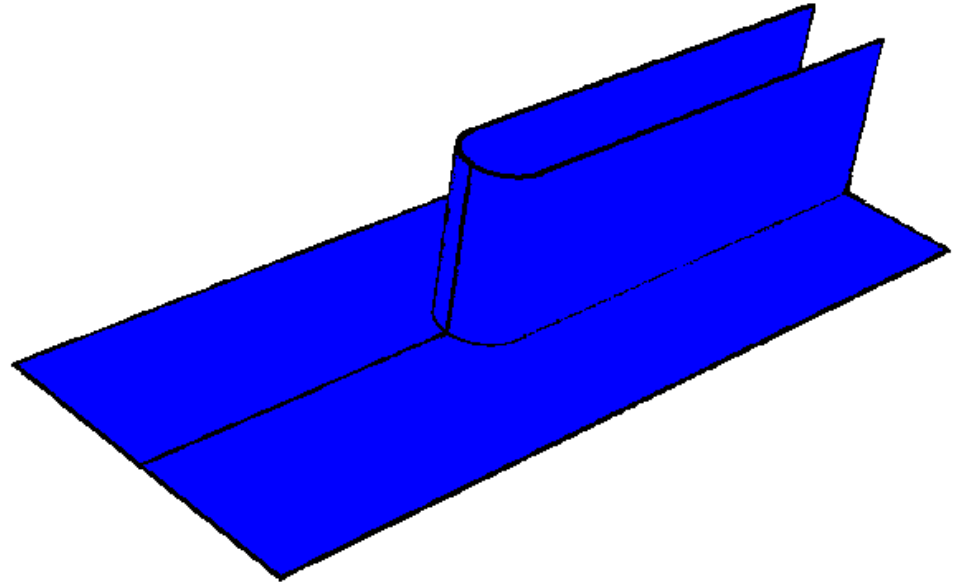
- Endwall extended upstream
- Inlet planes identical
- Allowed for more realistic turbulent dispersion of particles



- Deposition rates identical for both extended endwall designs

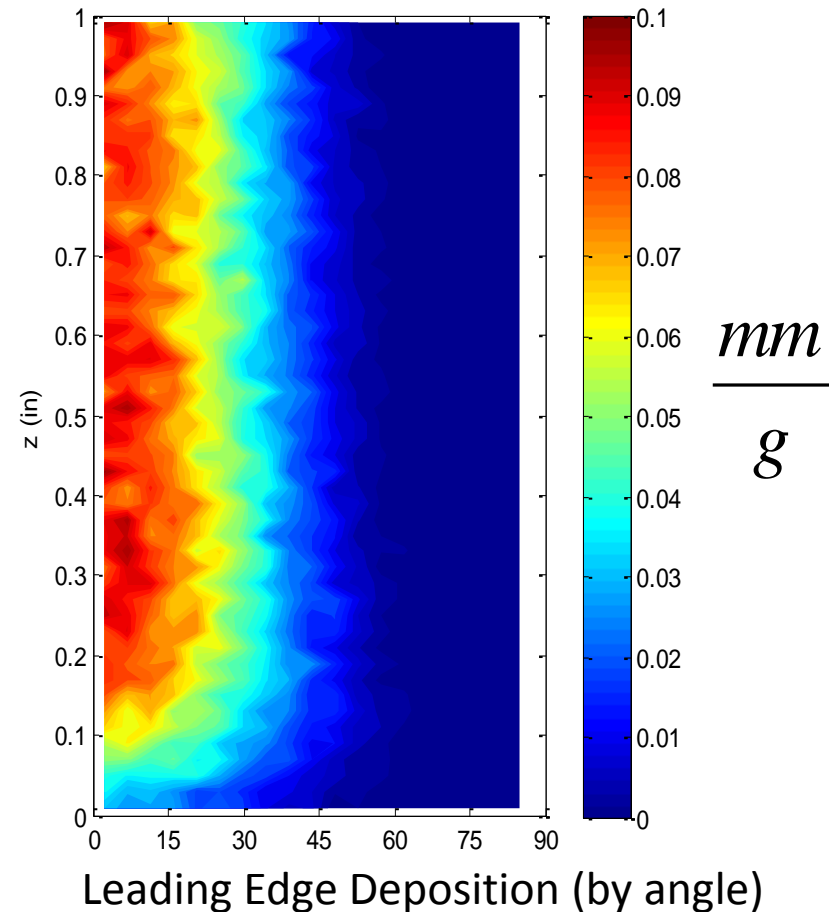
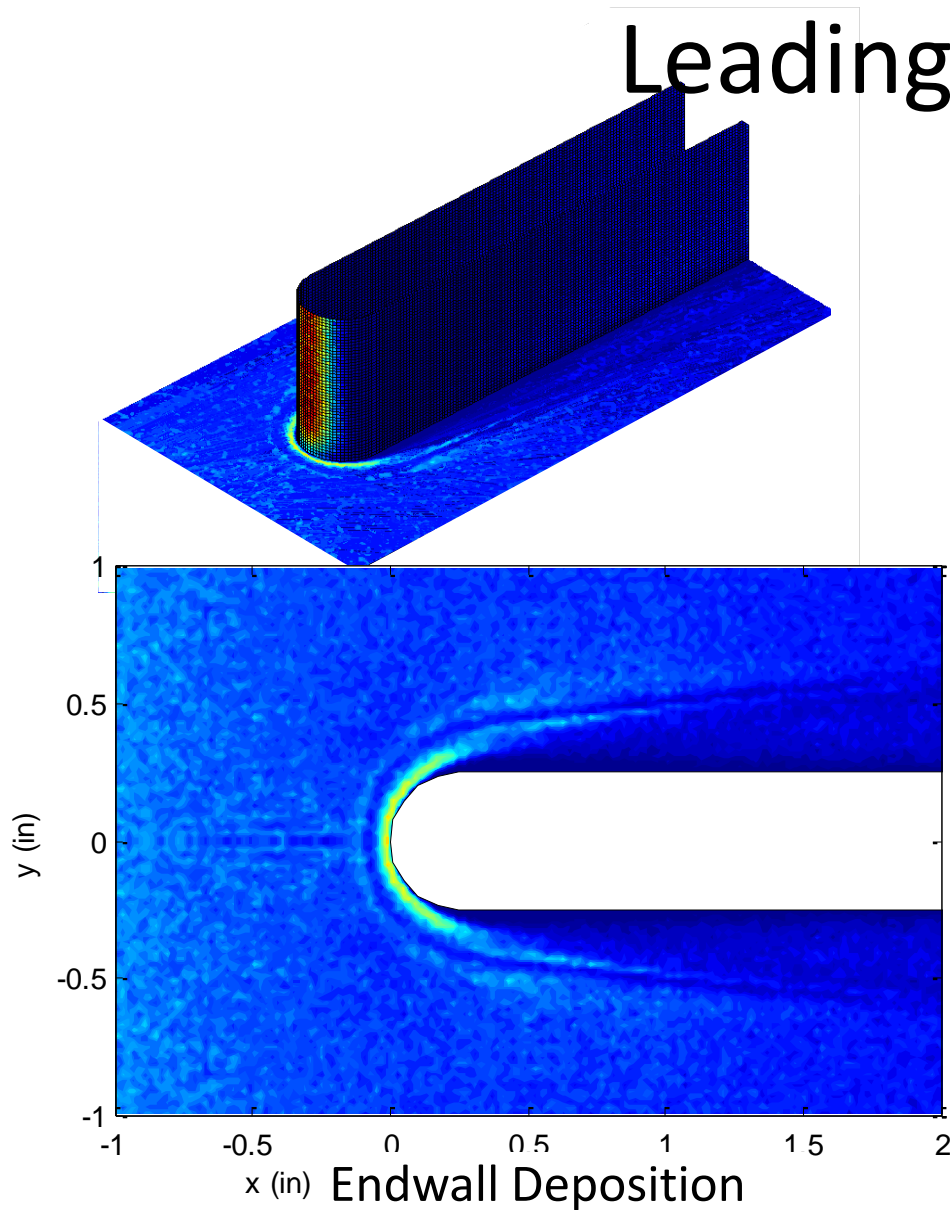
Flat Endwall Studies

- Investigate how particulate deposits on endwalls.
- Ran tests using a flat plate with a cylindrical leading edge
- Rolls Royce high pressure vane with flat endwalls



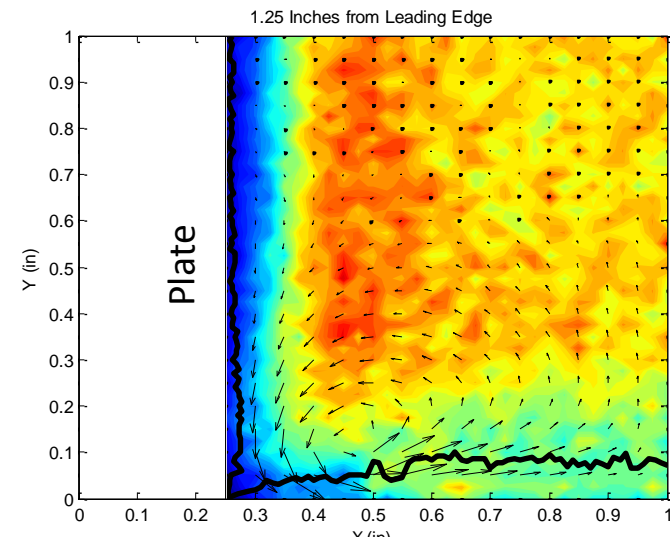
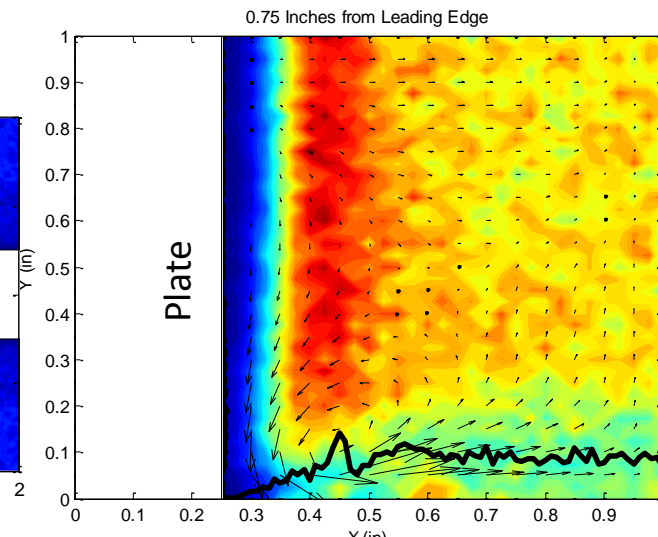
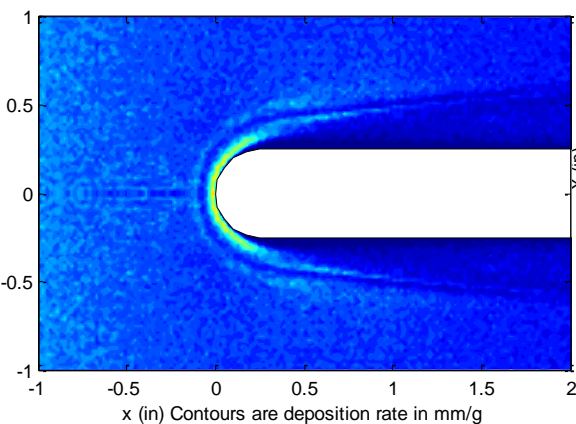
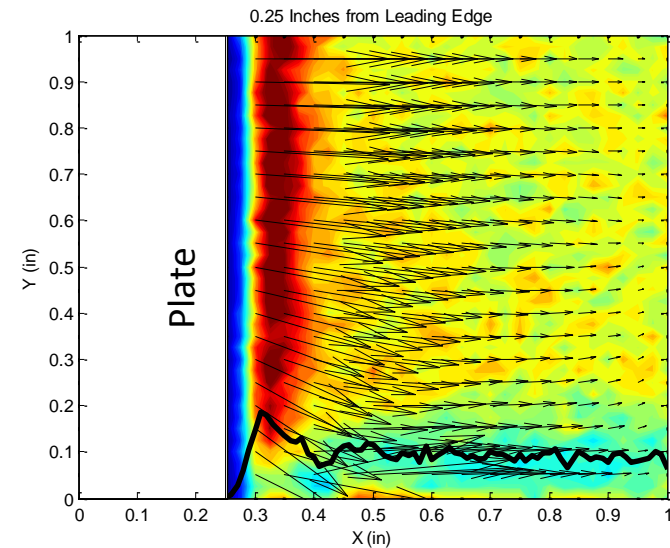
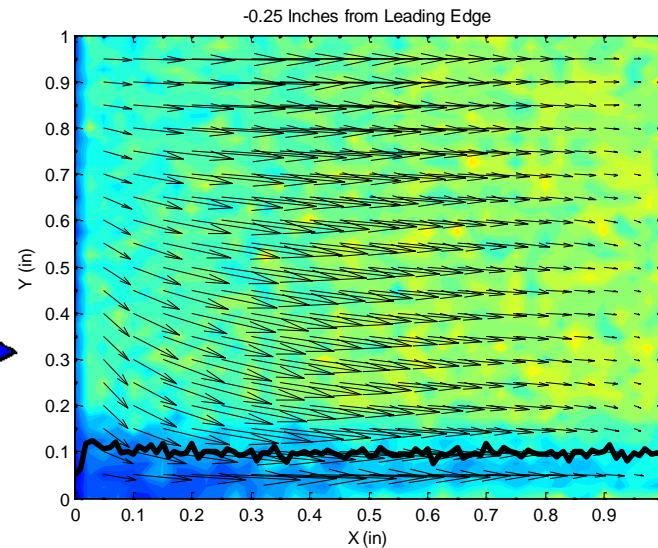
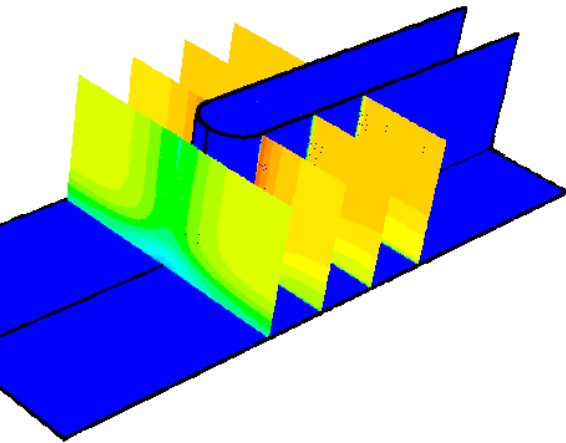
Flat Endwall Studies – Flat Plate

Leading Edge



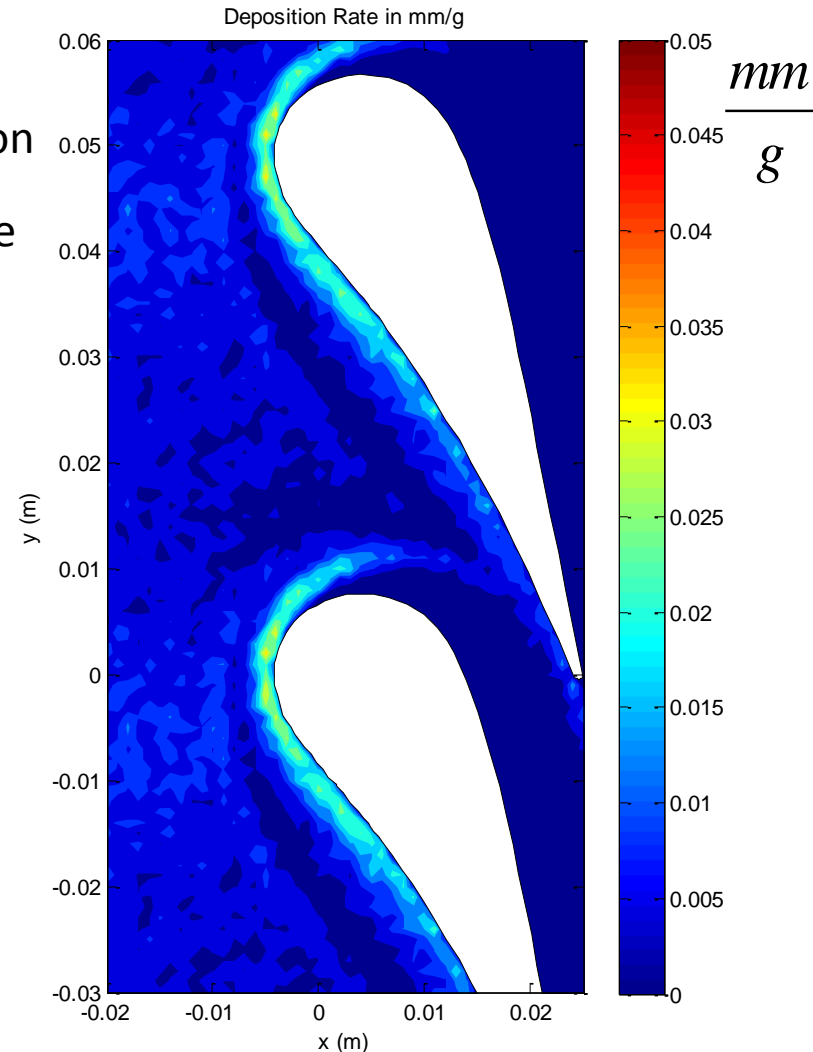
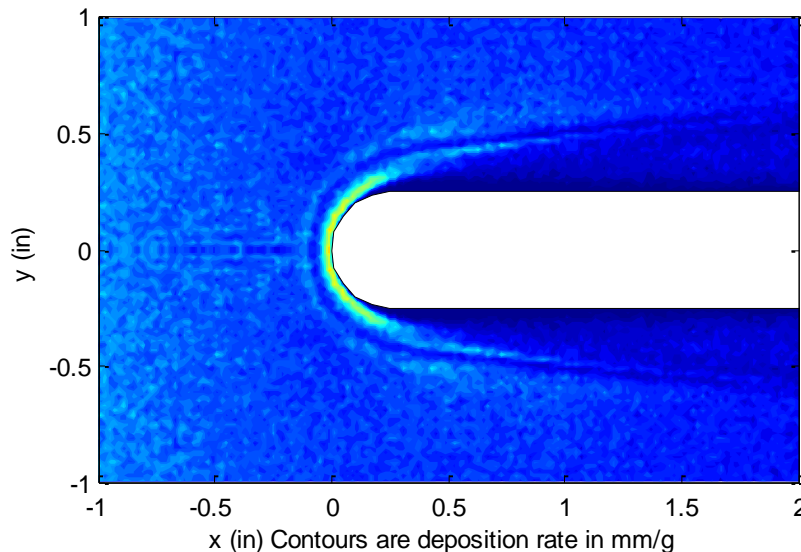
5 micron Particles, $St_k = 0.25$ $Re = 12.7K$

Flat Endwall Studies – Particle Concentration



Flat Endwall Studies – Vane Endwall

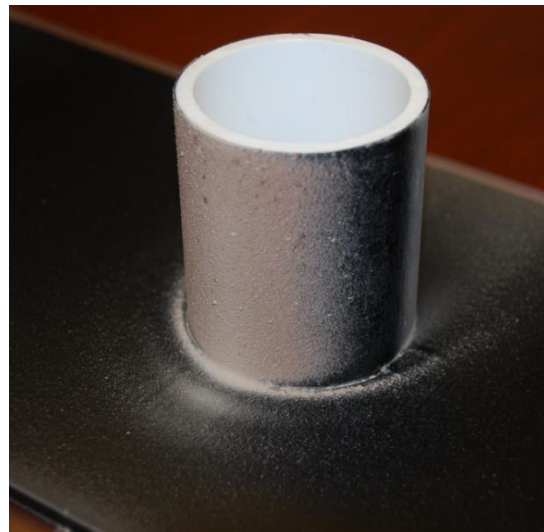
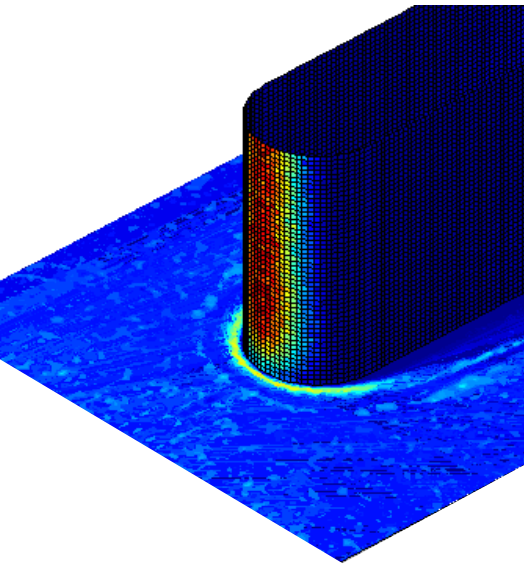
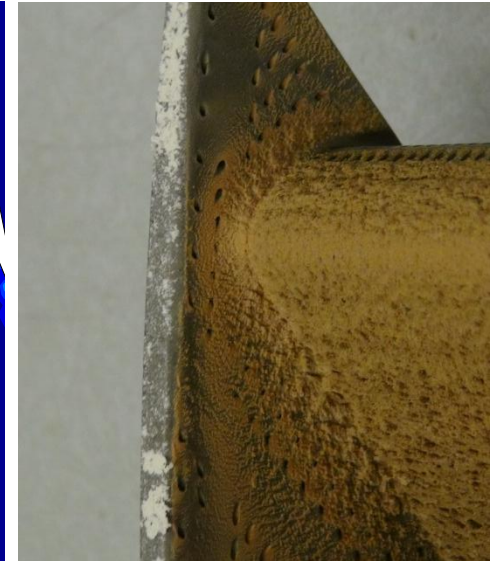
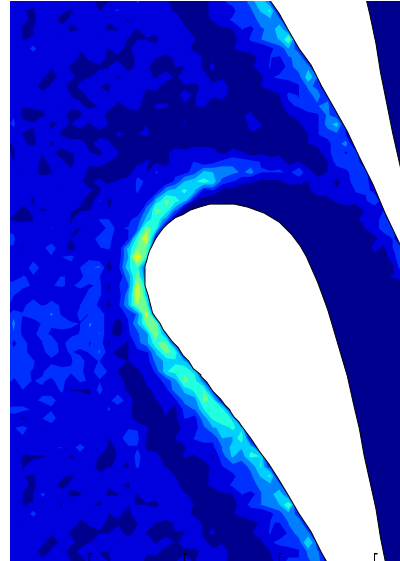
- Largest Deposition rates near leading edge.
- Even these deposition rates on endwalls are generally less than half of the magnitude of deposition on leading edges (or pressure surface).
- Larger particles responsible for deposition on vane surfaces. Alternatively, smaller particles are more susceptible to deposit on endwalls.
- Vane deposition dominated by inertial impaction. Endwall deposition due to turbulent dispersion, secondary flows, or particle concentration.



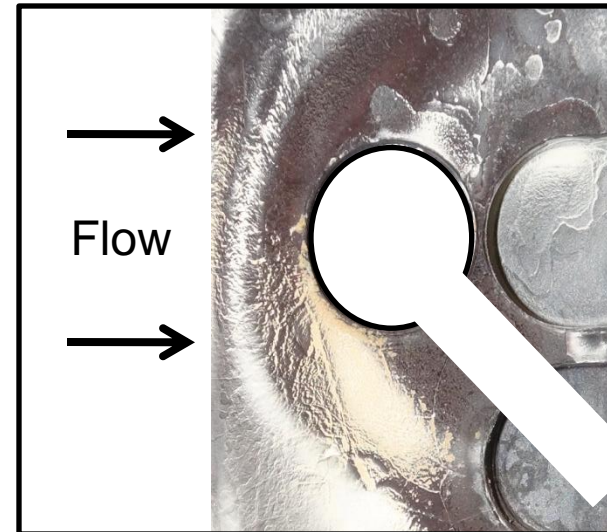
5 micron Particles, $St_k = 0.25$ $Re = 12.7K$

Flat Endwall Studies – Comparison with Experiments

- Leading edge endwall deposition and secondary flow patterns observable in experiments, especially with low St_k number
- Endwall deposition minor compared to leading edge.

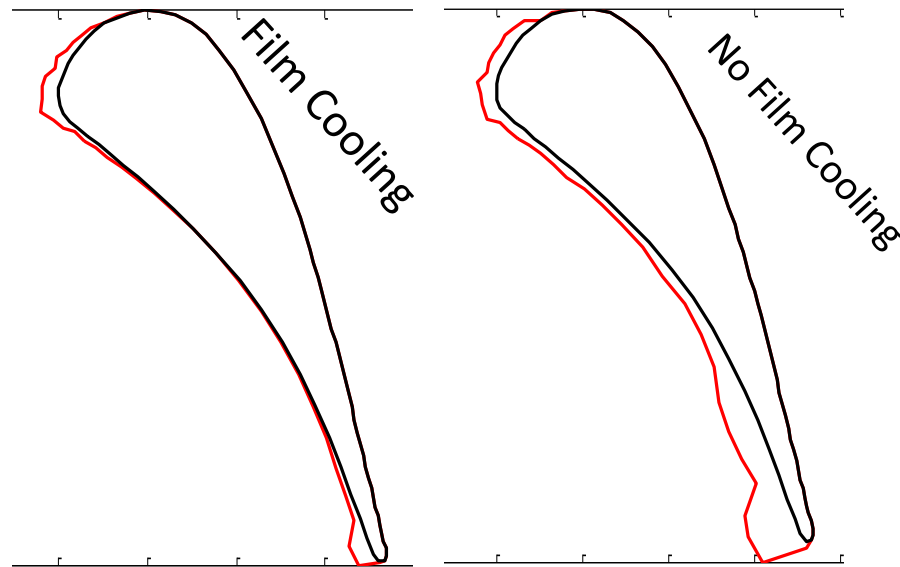


Lawson, Thole (2011)

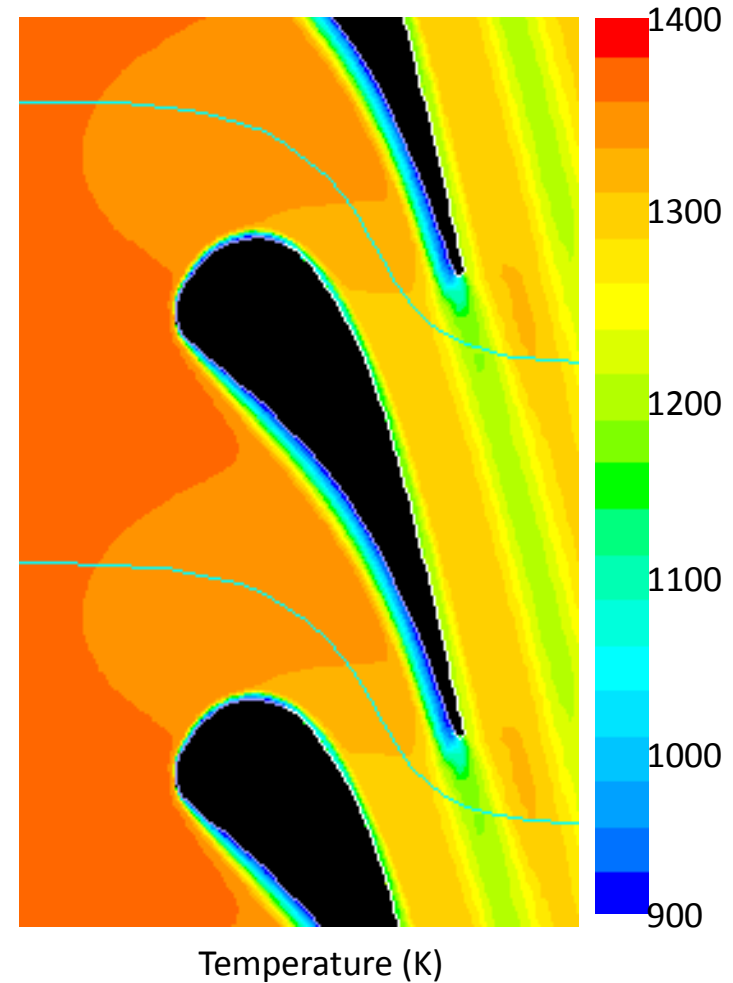


Faired Cylinder TuRFR test

Film Cooling –Transpiration Cooling



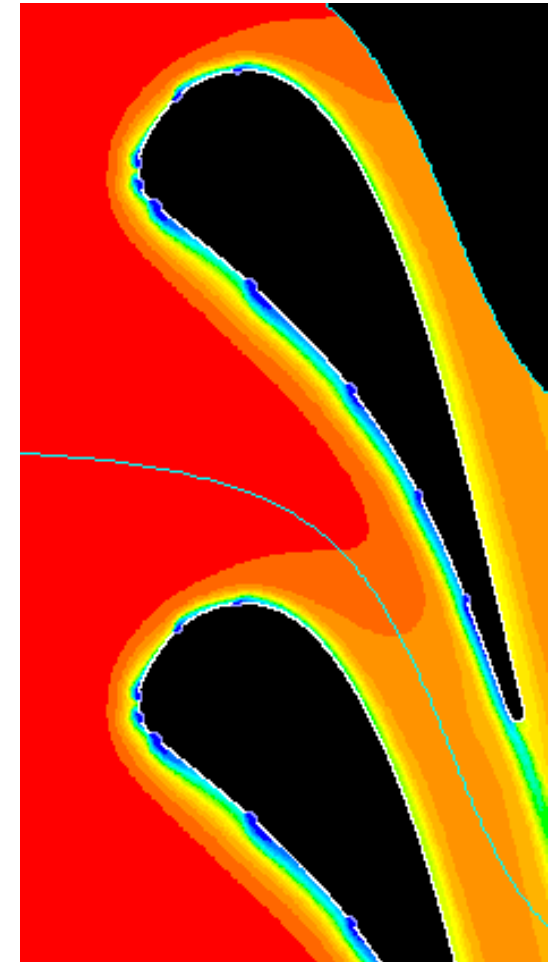
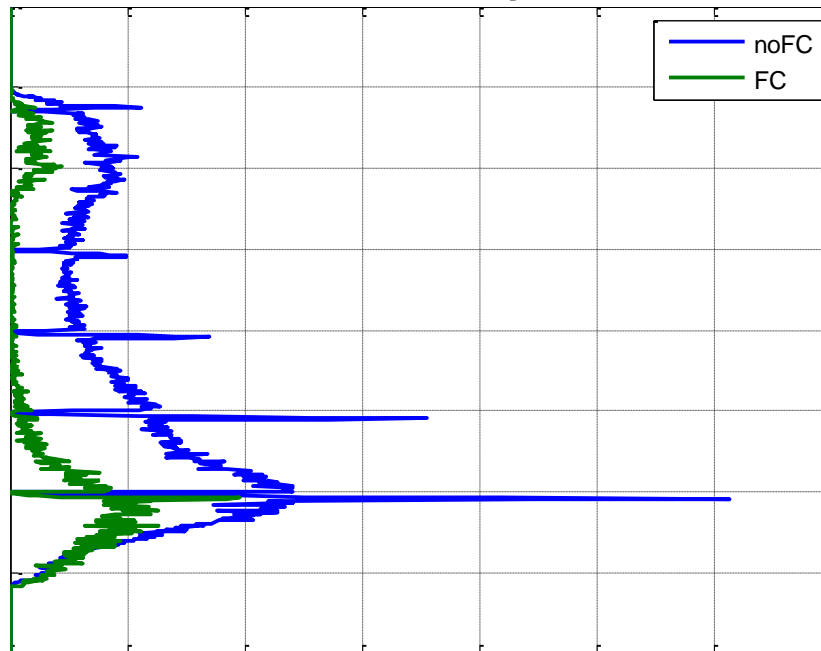
- Film cooling reduced deposition by 80% for 6 μm diameter particles
- Reduced trailing edge deposition for small St_k particles, but not as much as is seen in experiments.
- Little change for large St_k



Film Cooling – Slit Cooling

- Similar results as transpiration cooling
- Leeward side of film cooling “holes” are sites for much larger deposition rates.
- Average deposition not significantly changed in hole locale
- Same limitations as transpiration cooling.

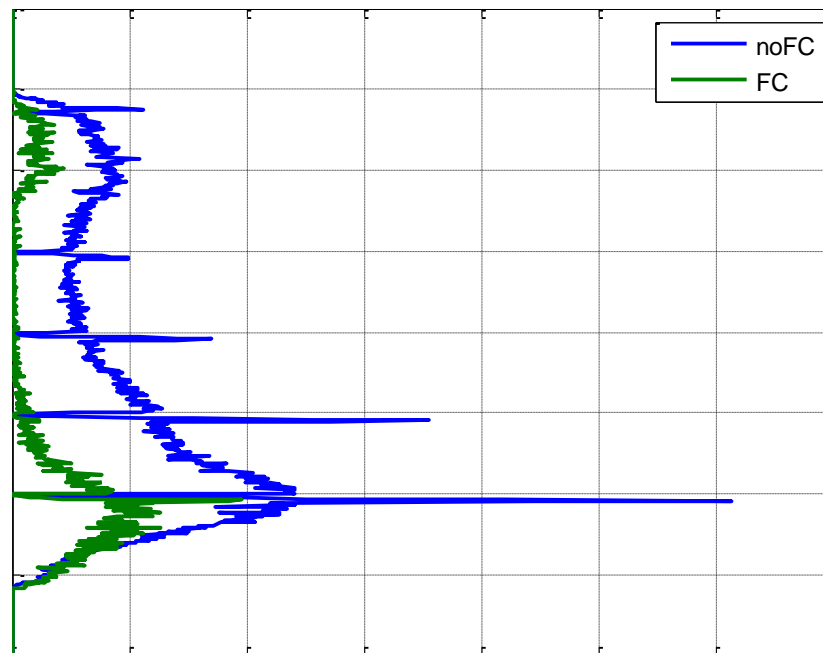
Deposit
histogram on
tangential
direction
10 μm



Film Cooling – Slit Cooling

- Experimental tests indicate that deposition begins in film cooling holes, even when film cooling is present

Deposit
histogram on
tangential
direction
10 μm



Effect of Stokes Number and Film Cooling on Deposition

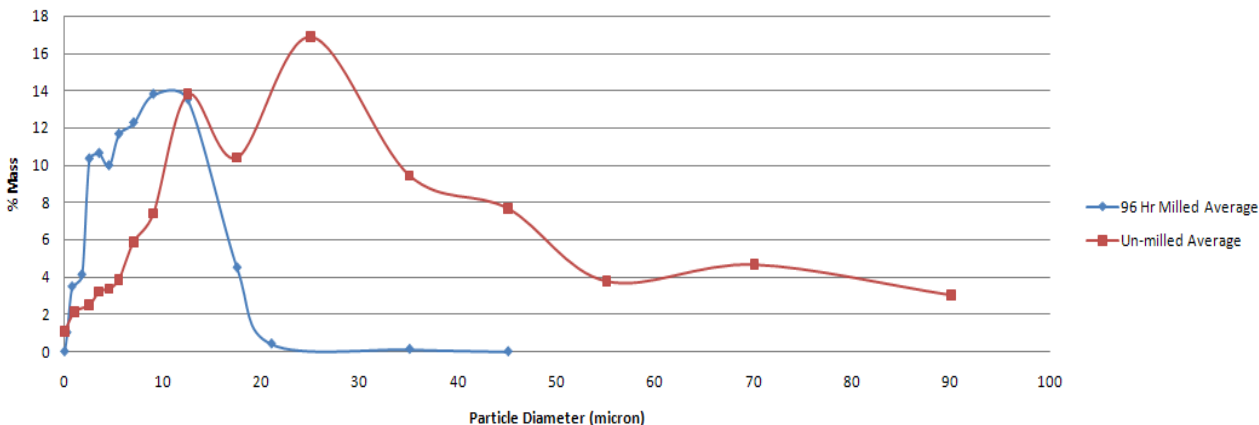
Experimental Data

Ash Characterization

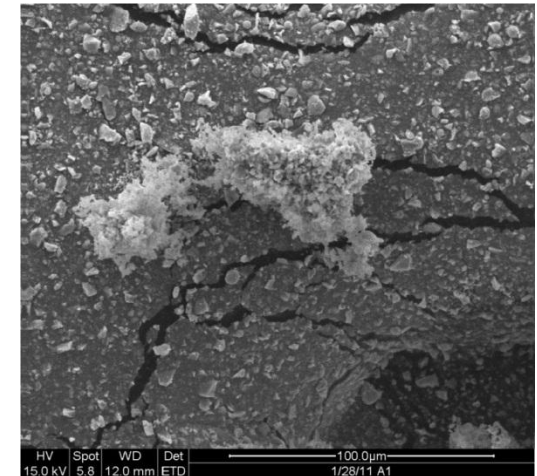
- Sub-bituminous ash from Jim Bridger Power Station (JBPS)
- Ash is milled for 96 hours to decrease particle diameter

Statistical Values for JBPS Fly Ash (microns)

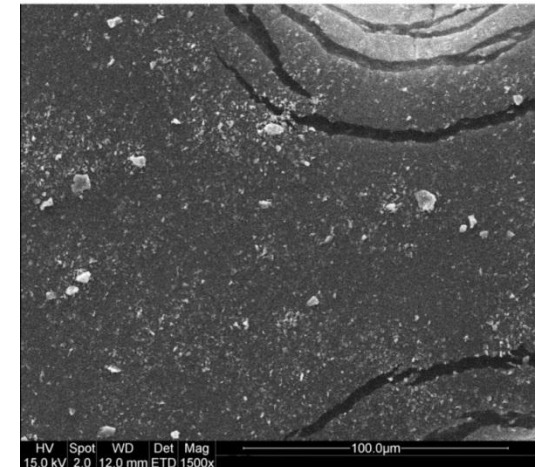
	96 hours	Un-milled
Mean Diameter	6.8	24.4
Modal Diameter	8.2	27.0
Median Diameter	5.9	17.6



Averaged Particle Size Distributions for JBPS Coal Fly Ash



(a)



(b)

Micrographs :

- a) Un-milled
- b) 96 Hr. Milled JBPS

TuRFR Test Results

JBPS Ash Condition	Un-milled	96 Hr. Mill	Un-milled	96 Hr. Mill
Inlet Mass Flow (kg/s)	0.38	0.39	0.39	0.39
Mach Number	0.084	0.082	0.076	0.082
Inlet Temperature (°C)	1084	1082	1083	1080
Exit Reynolds Number	229500	236500	245000	239700
Film Cooling %	7.9%	10.1%	0.0%	0.0%
Density Ratio	2.22	1.86	N/A	N/A
Total Particulate Mass (g)	300	251	292	329
Test Time (hrs.)	0.49	0.46	0.49	0.44
Particle Concentration (ppmw-hr)	221	178	205	234
Stokes Number	4.52	0.34	4.01	0.34

TuRFR Test Results

Uncooled

Cooled

$St \cong 4$



(a)



(b)

$St \cong 0.3$



(c)

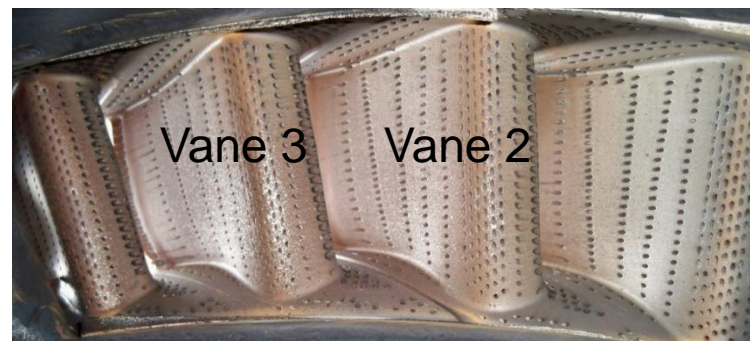


(d)

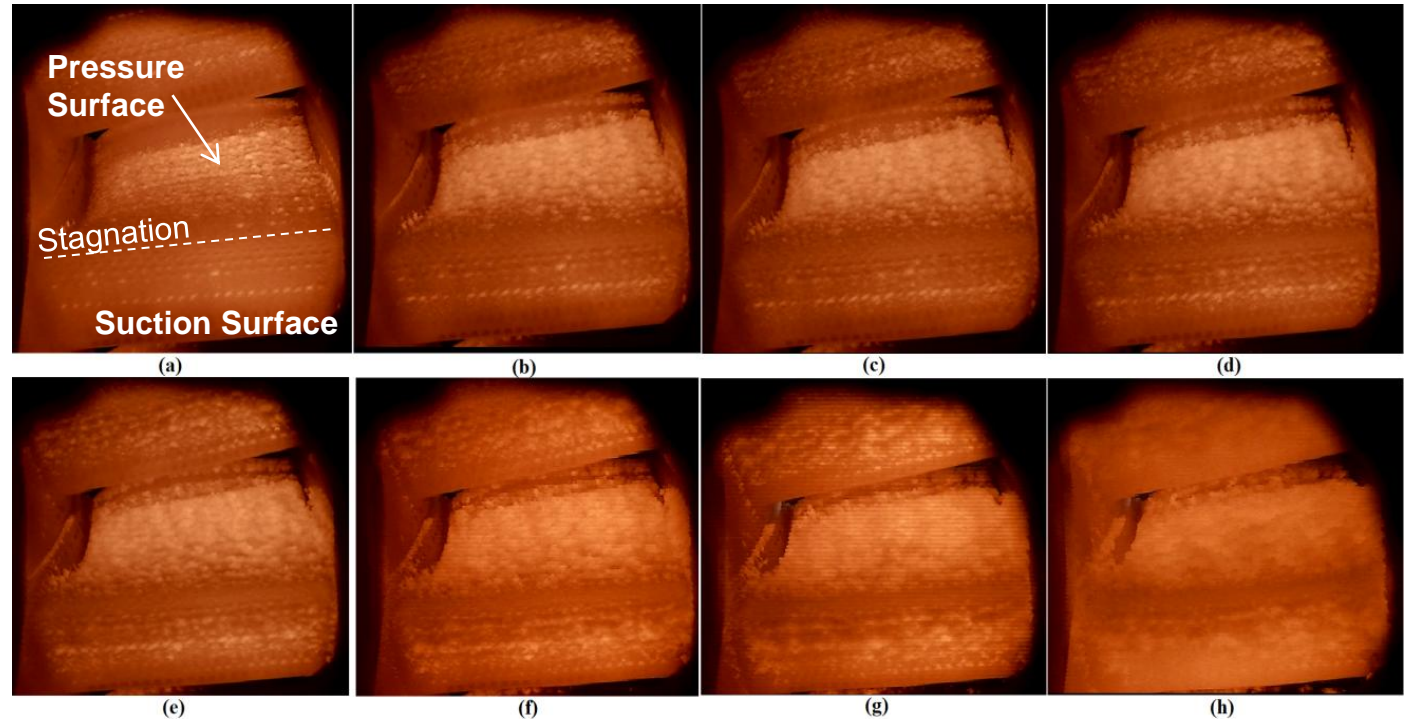
Capture Efficiency

Vane 3	Large Stokes No Film Cooling	Large Stokes with Film Cooling	Small Stokes No Film Cooling	Small Stokes with Film Cooling
1 st Set of Tests	27.0%	12.2%	9.5%	-
2 nd Set of Tests	31.2%	11.3%	5.8%	3.7%

Vane 2	Large Stokes No Film Cooling	Large Stokes with Film Cooling	Small Stokes No Film Cooling	Small Stokes with Film Cooling
1 st Set of Tests	19.9%	7.8%	6.6%	-
2 nd Set of Tests	22.1%	10.5%	5.1%	2.9%



No Cooling, Large Stokes

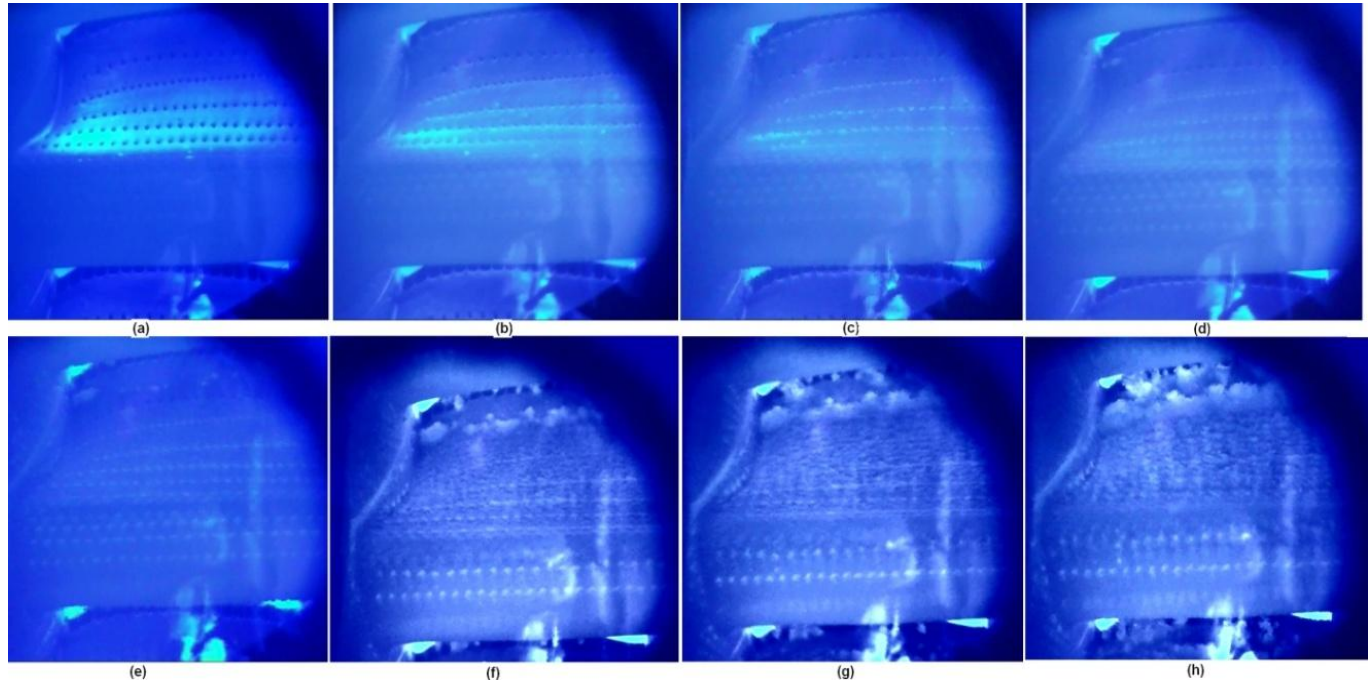


CFM 56 doublet

Development of JBPS Deposit, Large Stokes Number, without Film Cooling, 292 g injected: (a) + 1 min (b) +2 min (c) +3 min (d) +5 min (e) +7 min (f) +13 min (g) +20 min (h) +30 min

- Deposits begin to develop past mid-chord on the pressure surface and move forward toward the leading edge.
- Same progression on the suction side
 - Deposits on suction side only occur close to the leading edge

No Cooling, Small Stokes



CFM 56 doublet

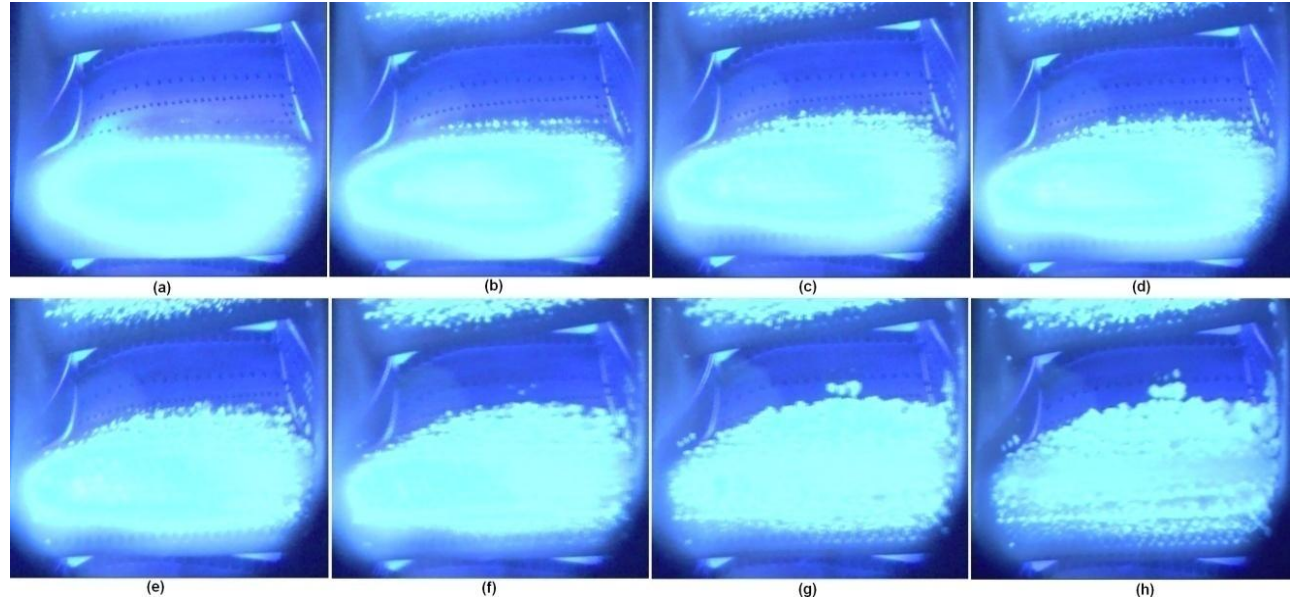
Development of JBPS Deposit, Small Stokes Number, without Film Cooling, 329 g injected: (a) + 1 min (b) +2 min (c) +4 min (d) +5 min (e) +7 min (f) +13 min (g) +20 min (h) +30 min

- Deposits develop in and around film cooling holes first.
- Higher accumulation is seen on the pressure surface towards the trailing edge.
- Much less overall deposit accumulation when compared to the large Stokes test.

9% Cooling, Large Stokes



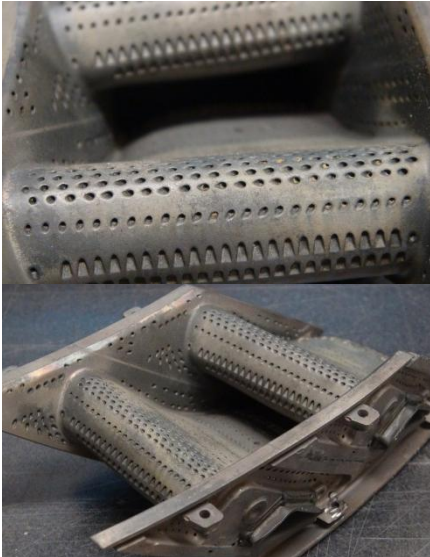
CFM 56 doublet



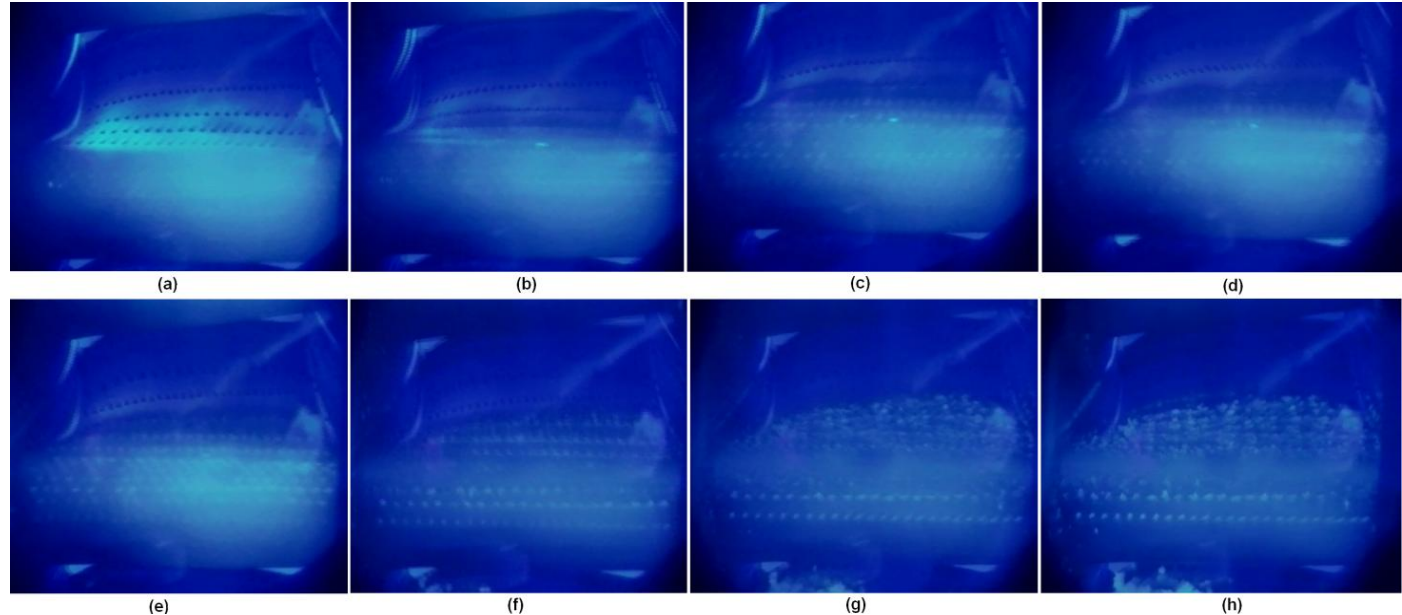
Development of JBPS Deposit, Large Stokes Number, with Film Cooling, 300g injected: (a) + 1 min (b) +2 min (c) +4 min (d) +5 min (e) +7 min (f) +13 min (g) +20 min (h) +30 min

- Deposition with film cooling begins on the leading edge, inside film cooling holes and remains more concentrated towards the leading edge.
 - No deposition past 60% chord
- Thick deposits present due to the large diameter of the particles

9% Cooling, Small Stokes



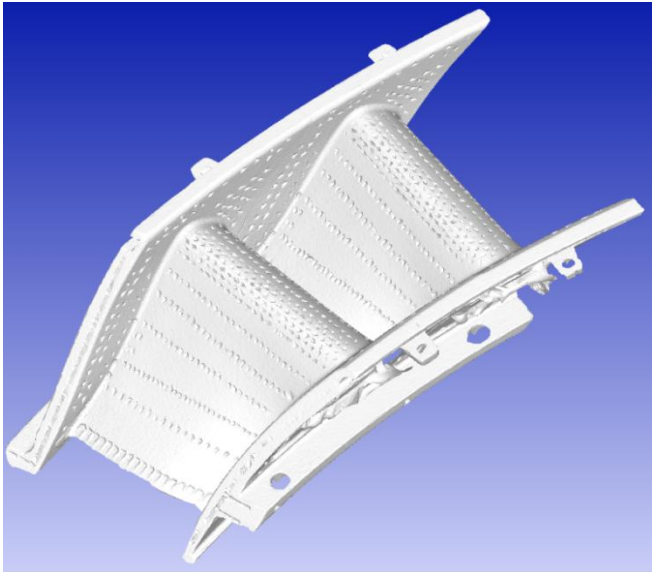
CFM 56 doublet



Development of JBPS Deposit, Small Stokes Number, with Film Cooling, 251g injected: (a) + 1 min (b) +2 min (c) +3 min (d) +5 min (e) +7 min (f) +13 min (g) +20 min (h) +30 min

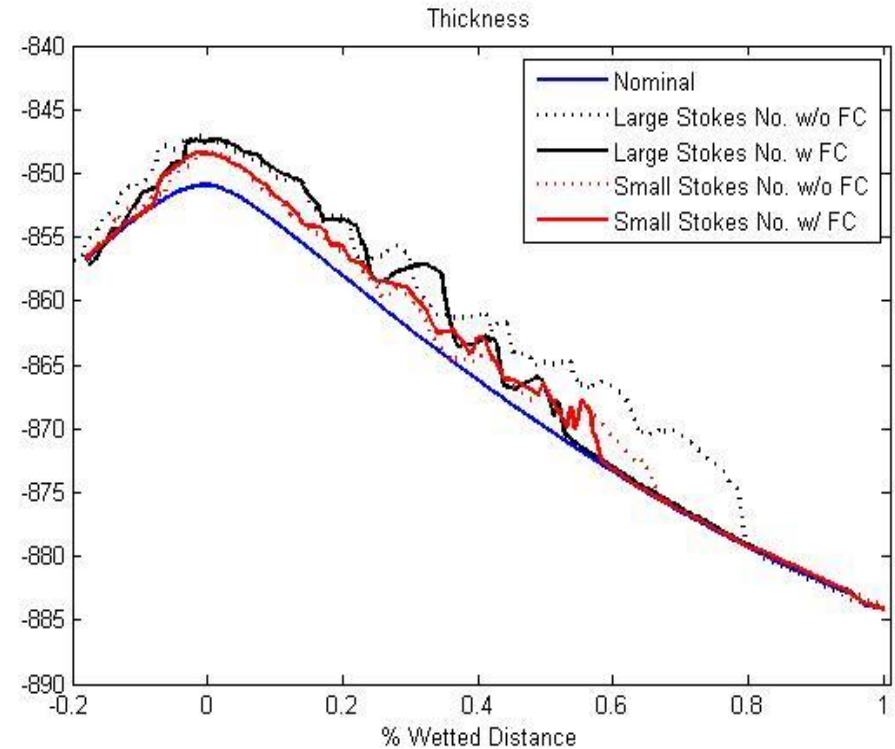
- Deposition begins inside film cooling holes towards the leading edge
 - No deposition past 60% chord
- Thinner deposits than seen for the large Stokes number test.

Surface Metrology



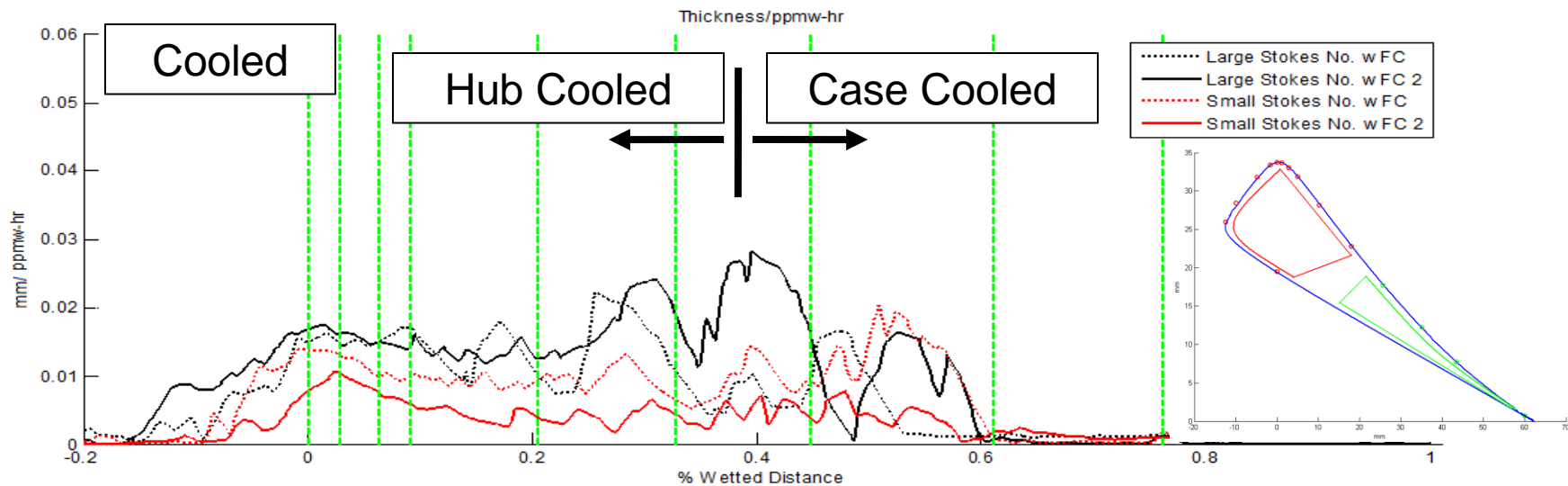
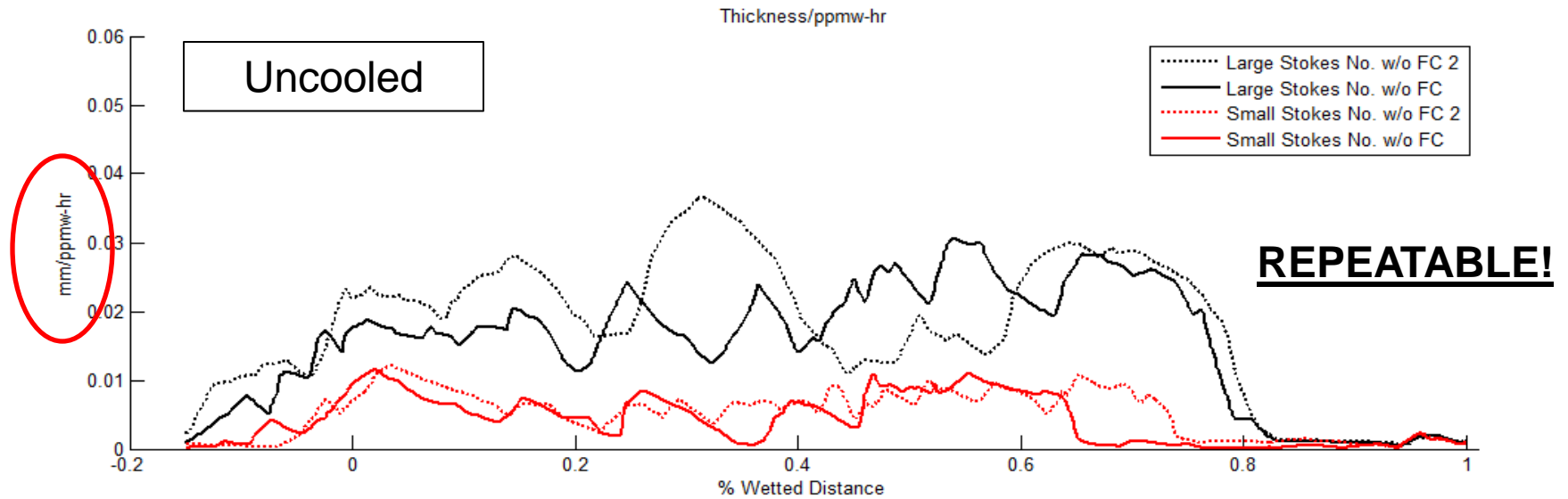
Laser Scan of Nozzle Guide Vane Doublet

- CMM Optical laser scan before and after deposition to acquire deposit thickness distribution

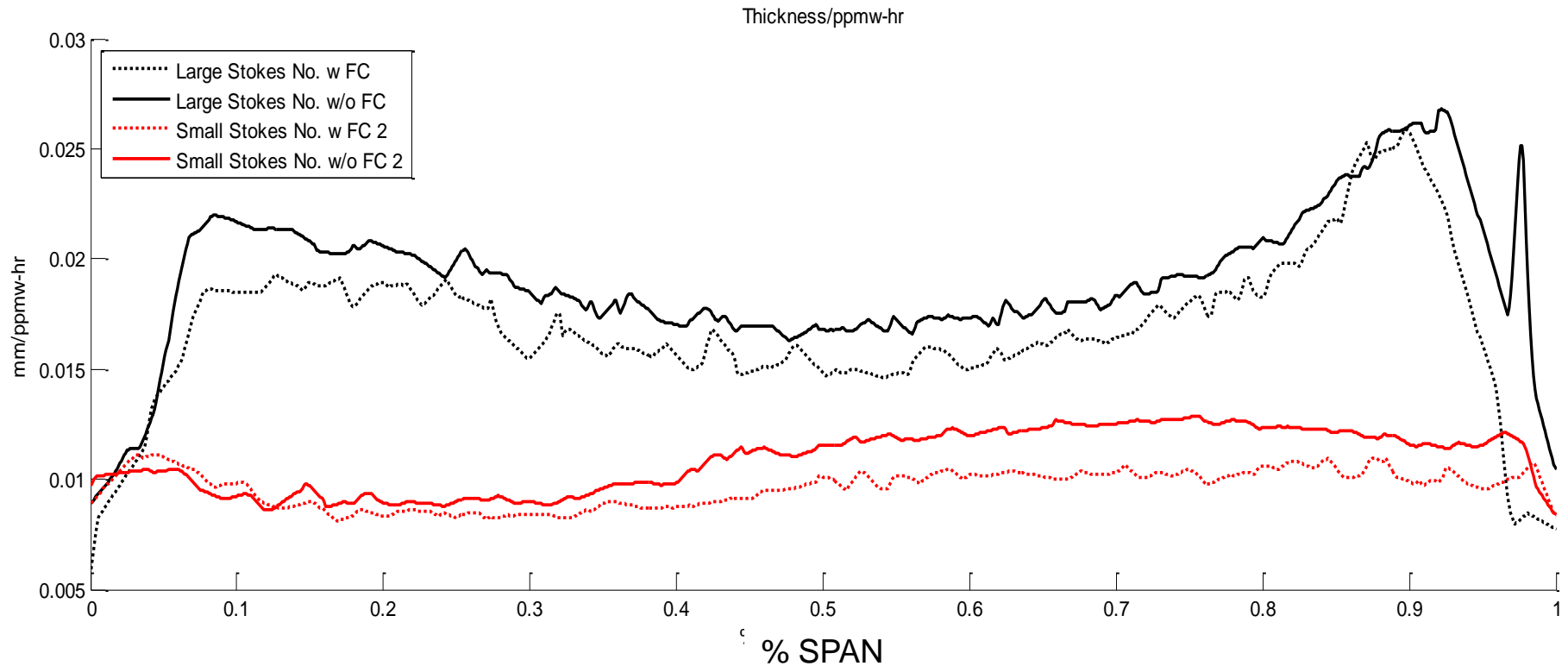


Mid-span Trace of Deposit Thickness with Nominal Vane

Mid-span Thickness Distribution

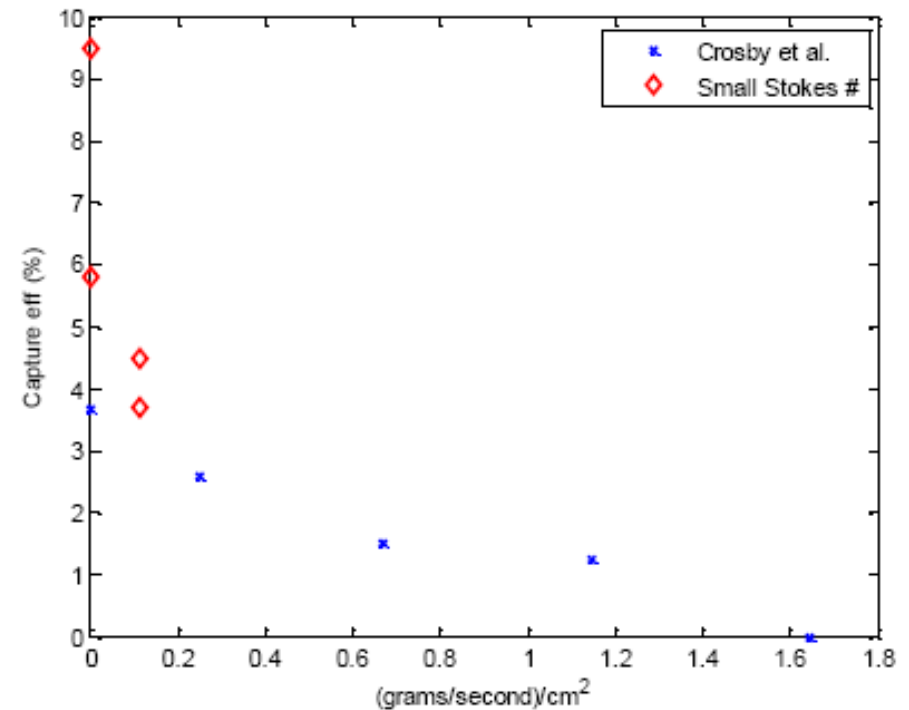
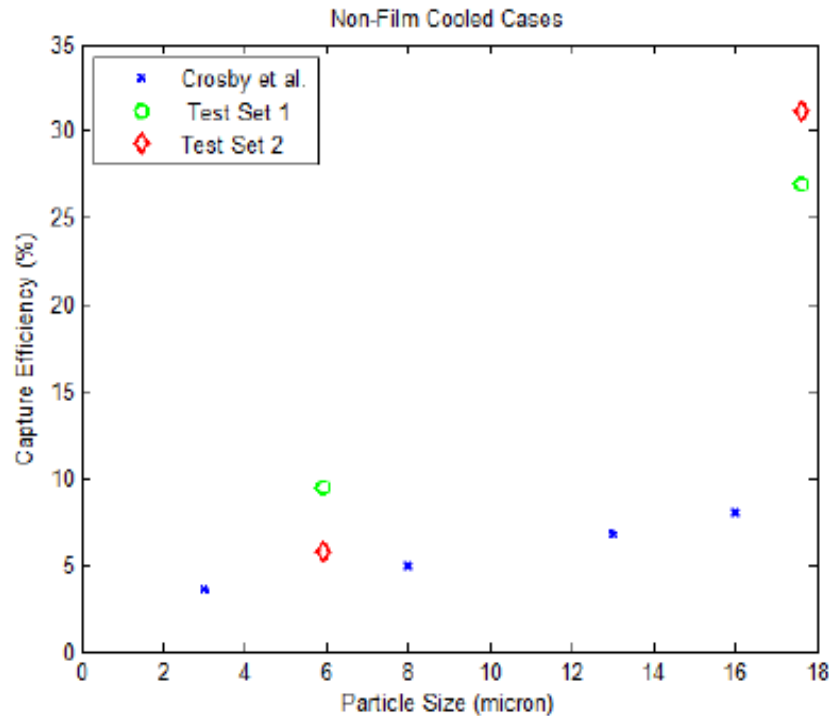


LE Thickness Distribution



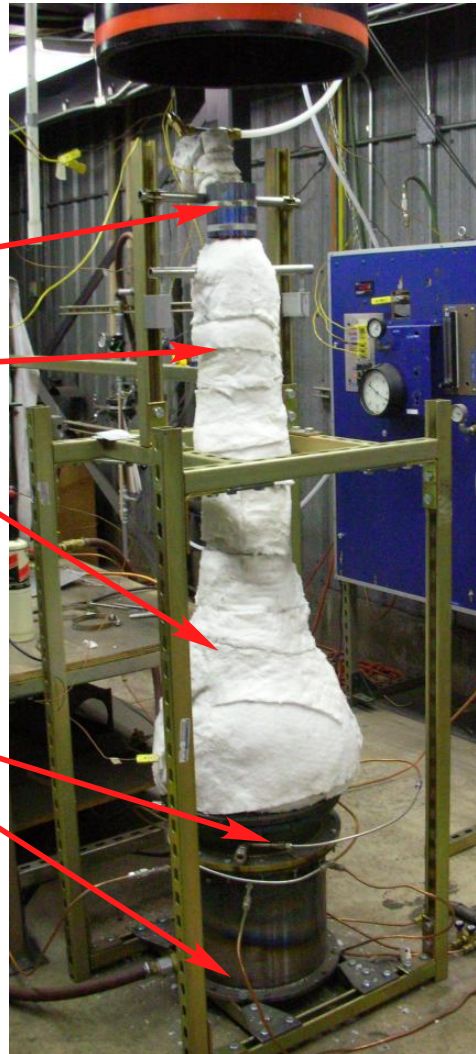
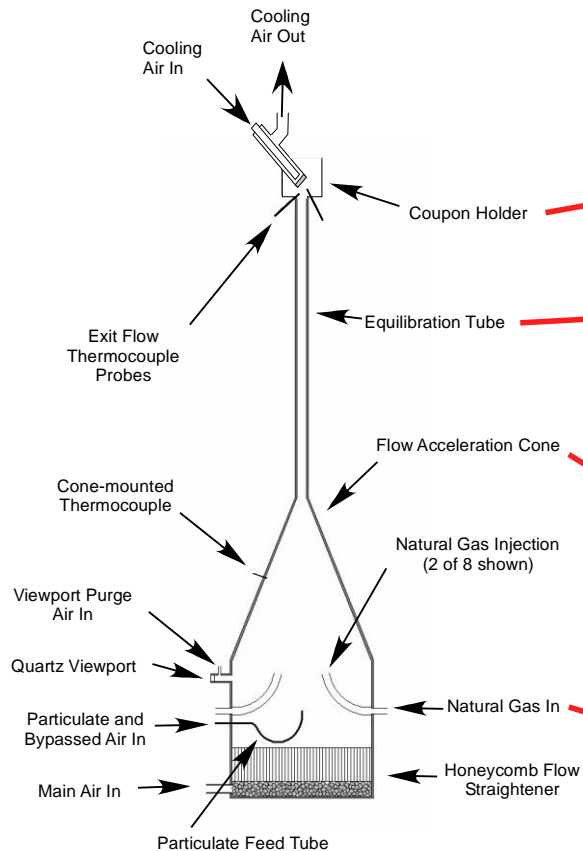
- 10-20% reduction due to cooling
- 60-70% reduction due to Stokes
- Large Stokes more susceptible to secondary flows near endwall

Comparison with TADF



BYU Results/Progress

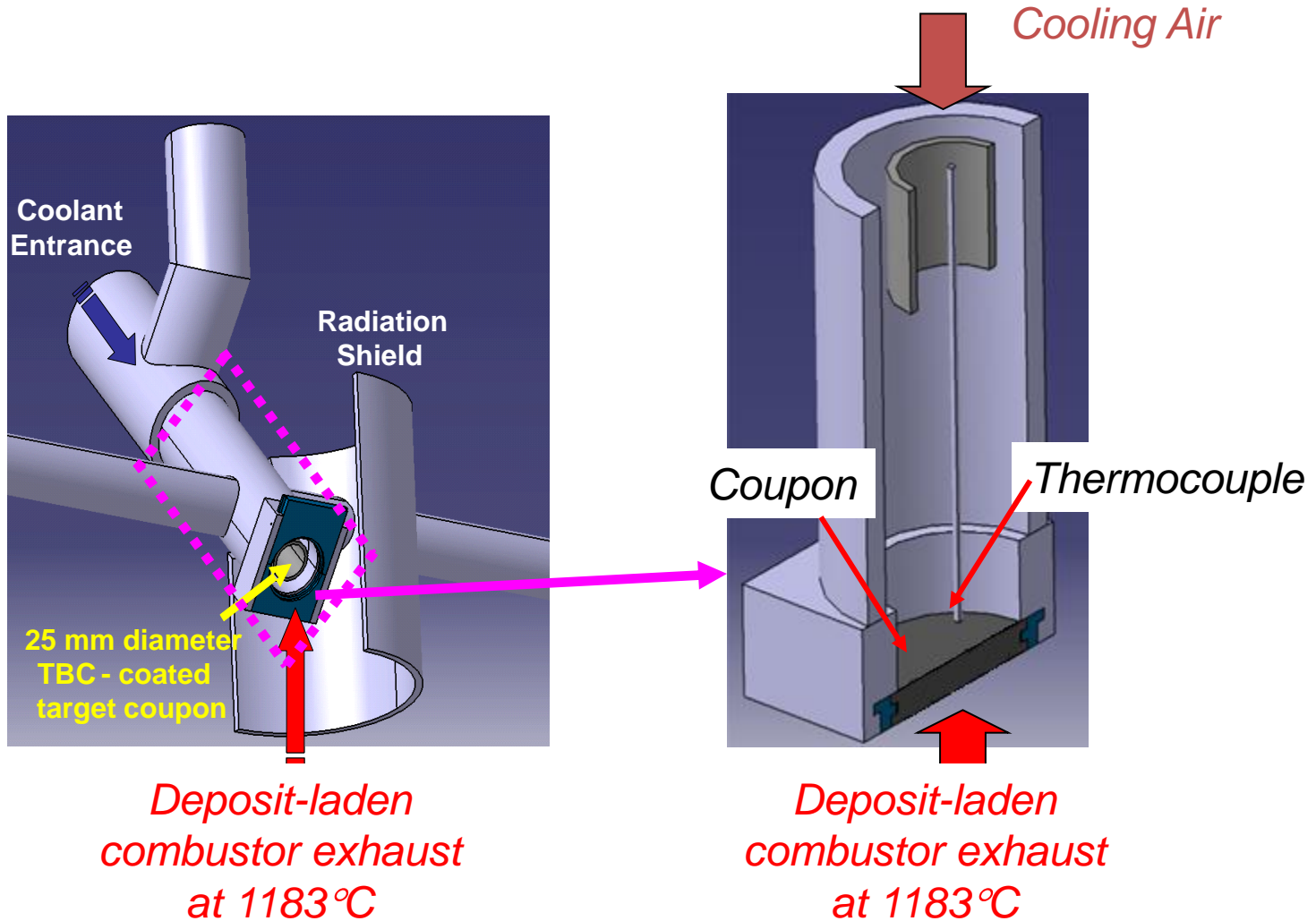
BYU Deposition Facility (TADF)



- Design Parameters to match: temp, velocity, angle, materials, particle size, chemistry, and concentration
- Inconel construction allows max jet temperature of 1200°C
- Exit velocities up to 300m/s – deposition by inertial impaction
- Target coupons supplied from industry
- Capability for impingement and film cooling
- Match net particle throughput

**8000 hrs × 0.1 ppmw ≈
4 hrs × 200 ppmw**

BYU Coupon Holder



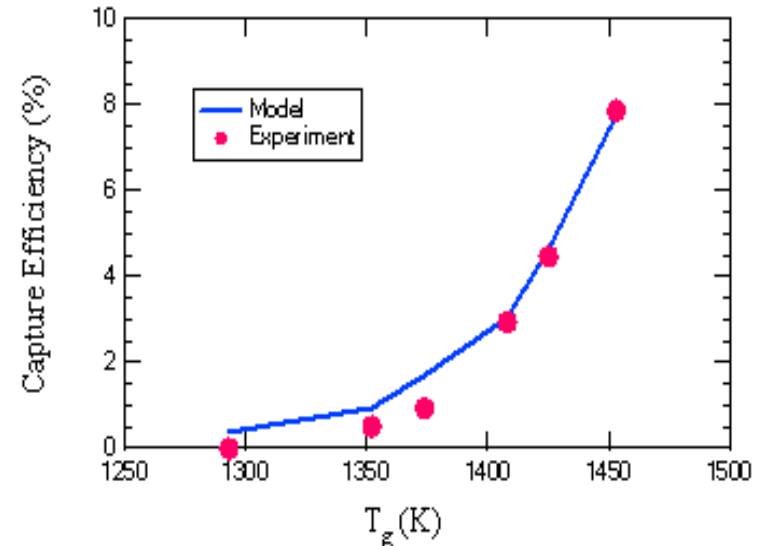
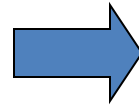
Coupons generally held at 45° angle to flow

BYU – Previous Testing



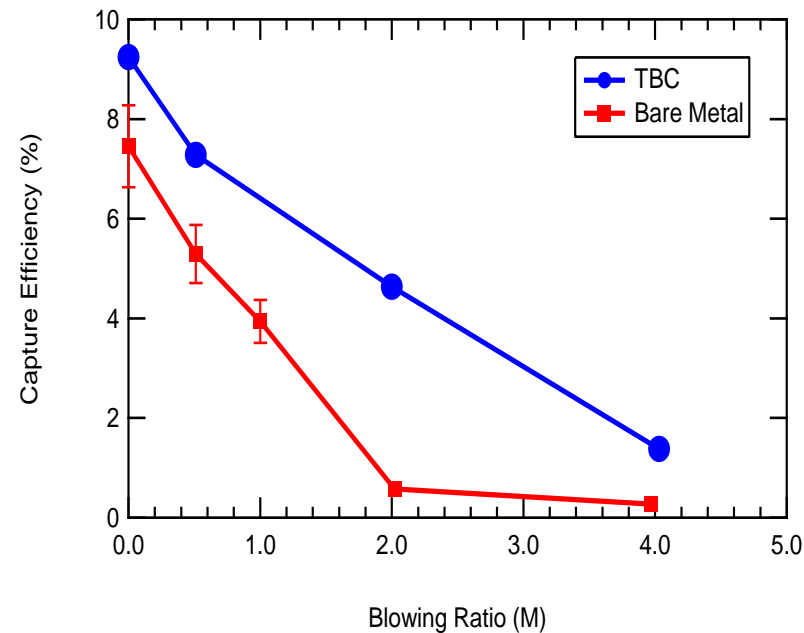
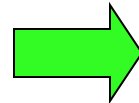
Deposition vs. Temperature

- Deposition increases with gas exit temperature
- Insulated tests conducted up to 1150°C (i.e., no cooling)
- No deposition below ~950°C



Deposition vs. Cooling

- Deposition decreases with increasing coupon cooling
 - backside cooling
 - film cooling on surface

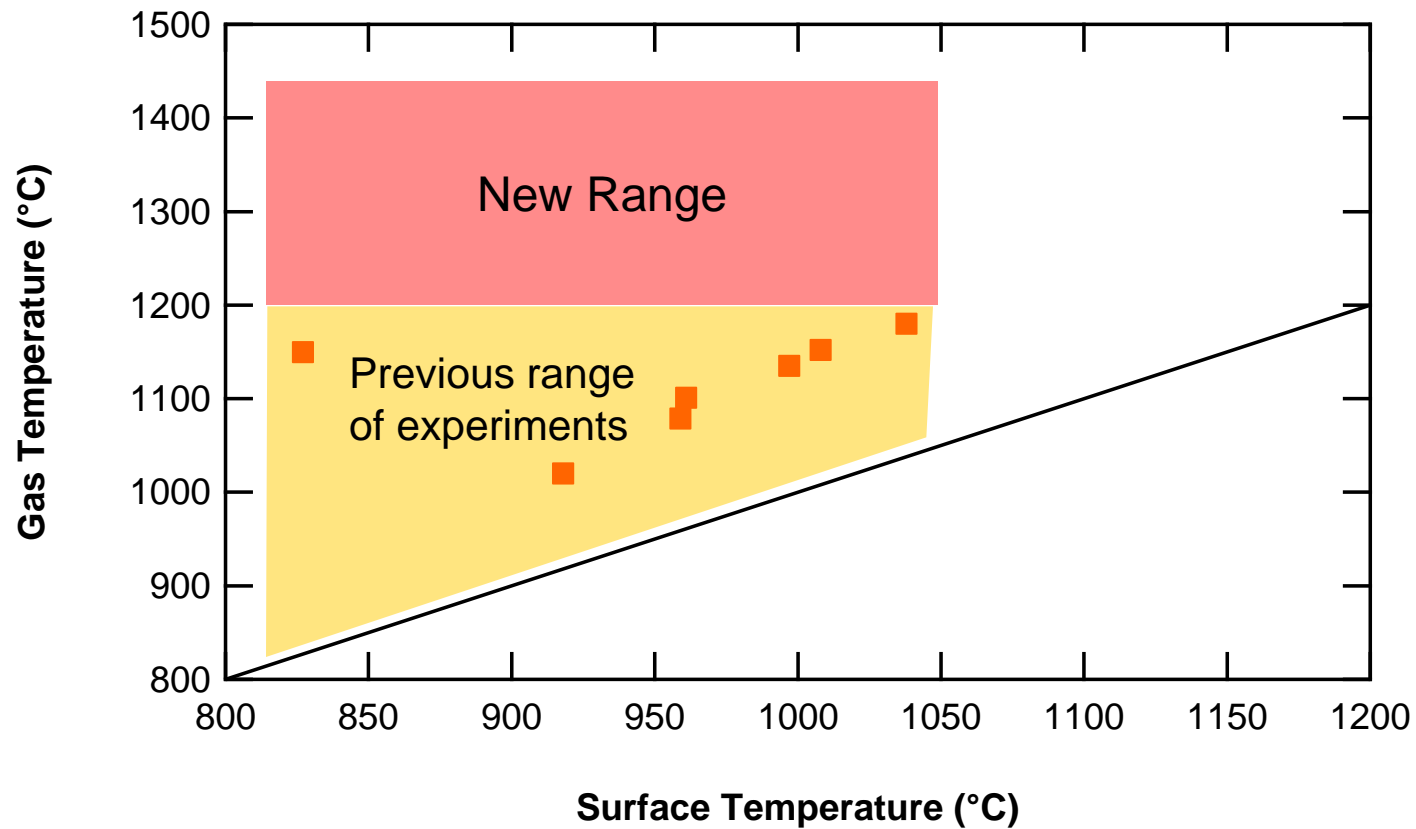




Current Research

- High temperature deposition testing at BYU
 - Modified deposition facility to operate at gas temperature (T_g) up to 1400 °C
- Why test at high temperatures?
 - Previous tests were performed at a T_g of 1150 °C
 - Modern H-class turbines operate at T_g closer to 1400 °C
 - 1150 °C is near the threshold T_g of 950 °C as reported by Crosby et al. (2008)
 - Testing at higher temperatures will distinguish between the effects of gas temperature (T_g) and surface temperature (T_s)

Temperature Range



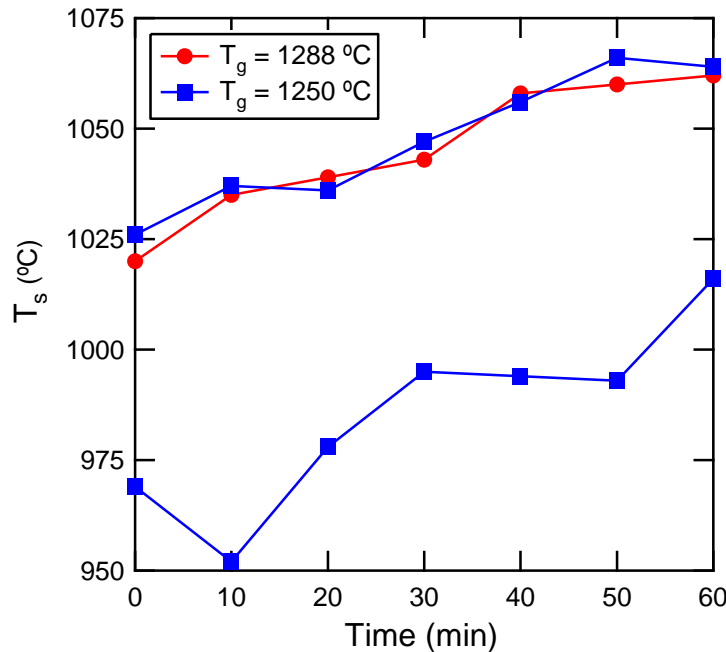


Current High Temperature Testing

- Two main test series are underway, but not yet completed
 1. Maintain the initial surface temperature (T_s) and vary the gas temperature (T_g) from 1250°C to 1400°C
 2. Maintain T_g at 1400°C and vary T_s
- Objective
 - Better understand influence of T_g and T_s individually on particle deposition
 - Accurately model particle deposition based on T_g and T_s



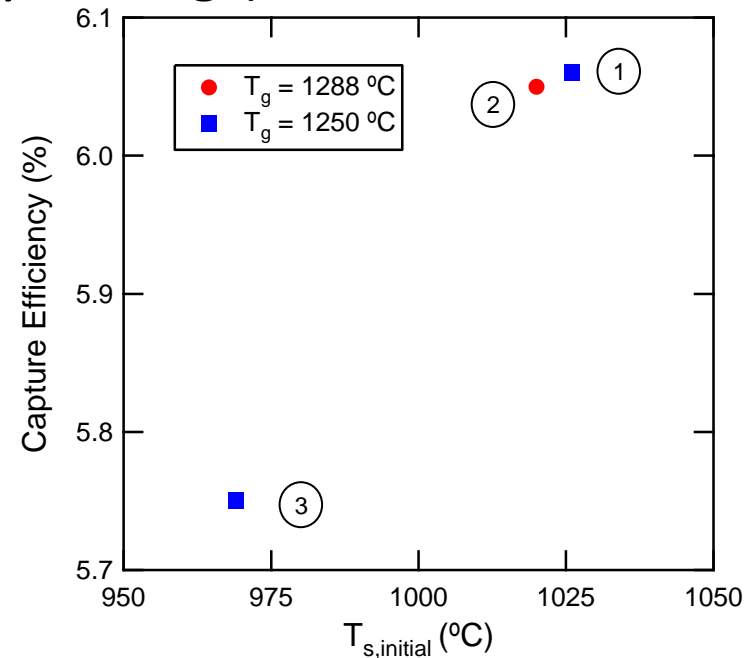
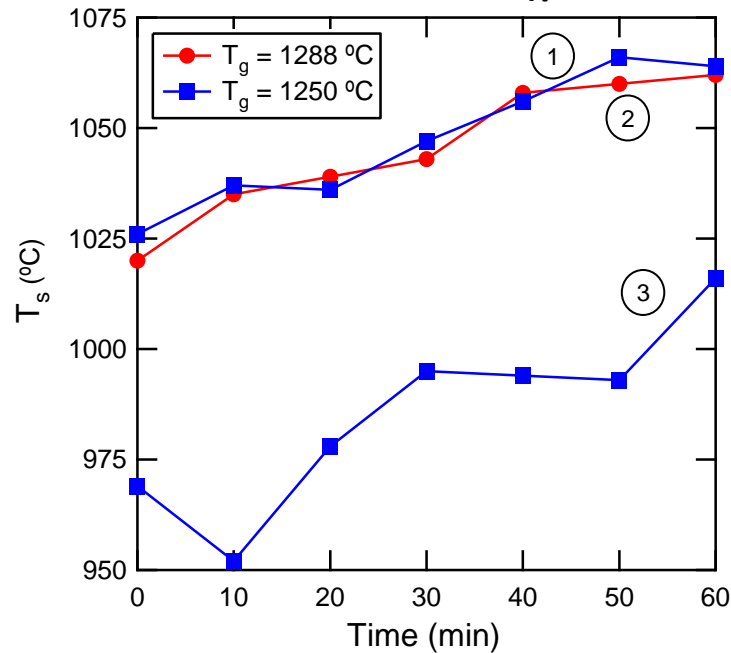
Results: Deposition vs. T_s



- 5 tests have been completed, but T_s data are only available for 3 tests
- At increased time, deposit grows and surface temperature increases
- Two tests had different T_g , but started with similar $T_{s,initial}$
 - T_s profiles of these two tests were similar
- Third test performed
 - High T_g (1288°C) but lower T_s due to coolant escaping on edges of coupon

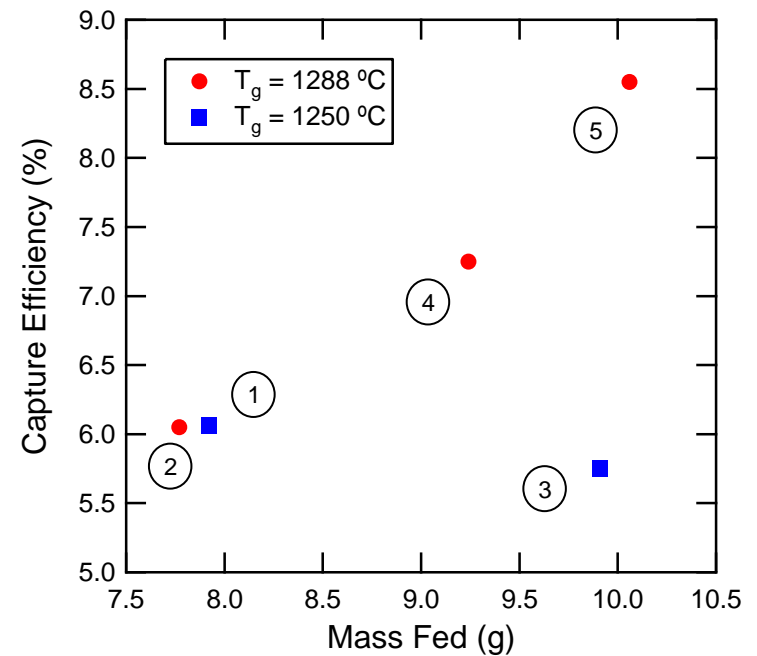
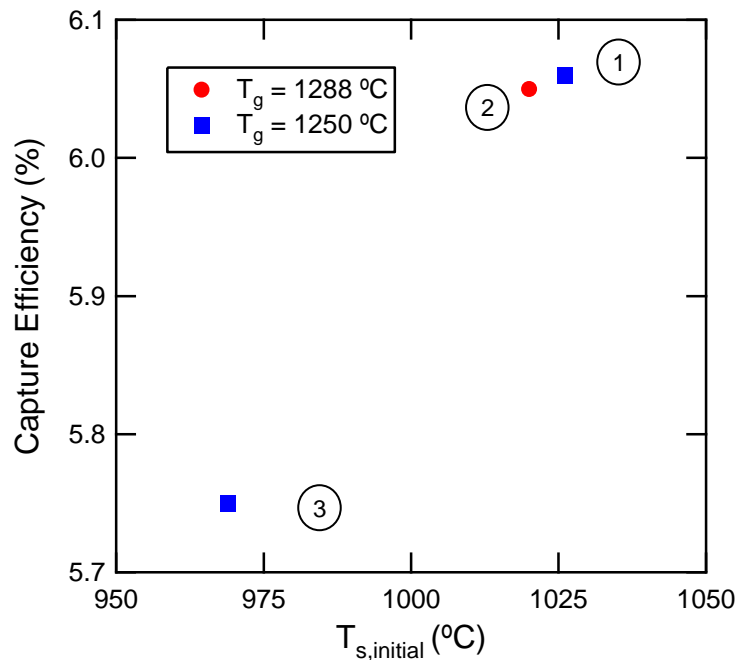
Results: Deposition vs. T_s

(preliminary findings)



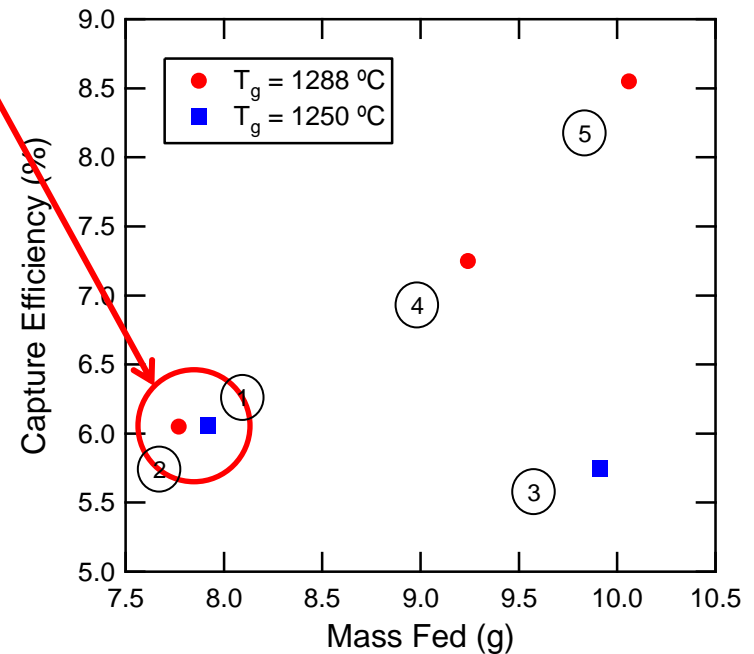
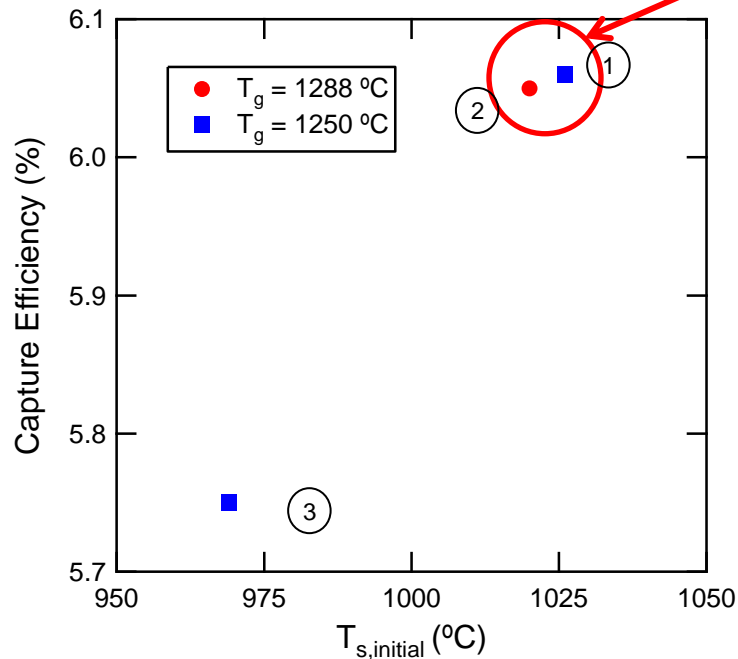
- Compare points 1 and 2
 - Different T_g , similar T_s
 - Similar capture efficiencies!
- Compare points 1 and 3
 - Same T_g , different T_s
 - Different capture efficiencies!

Capture Efficiency vs T_s and Particle Loading



Capture efficiency is seen to increase with both $T_{s,initial}$ and mass of ash particulate in gas stream

Capture Efficiency vs T_s and Particle Loading

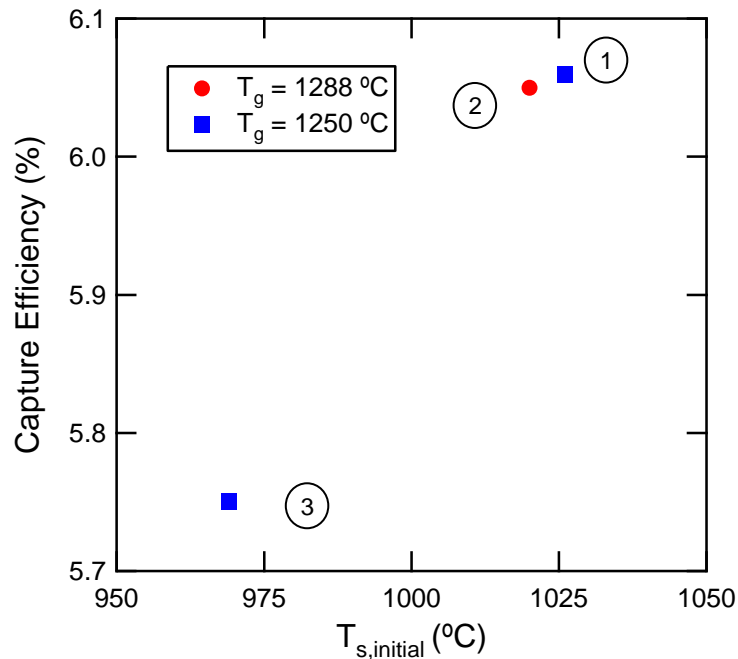


Compare tests 1 and 2

- Different T_g , but similar $T_{s,initial}$
- Mass throughput similar
- Result: **similar capture efficiencies**

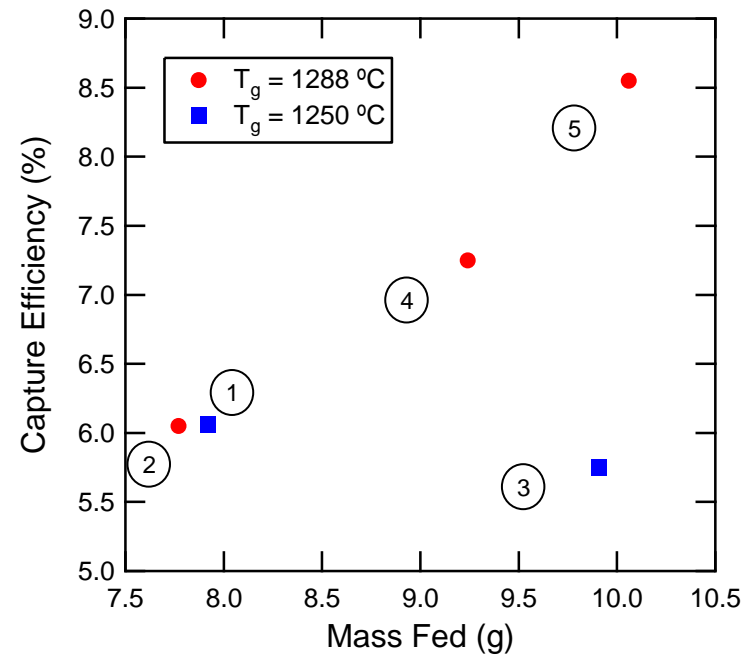


Capture Efficiency vs $T_{s,initial}$ and Particle Loading



Compare points 2, 4, and 5:

- For $T_g = 1288$ °C, capture efficiency is seen to increase with mass throughput



Compare points 1 and 3:

- For $T_g = 1250$ °C, the test with lower $T_{s,initial}$ resulted in a lower capture efficiency despite the higher mass throughput



Conclusions

Tests have been completed at the lower end of the desired range of T_g

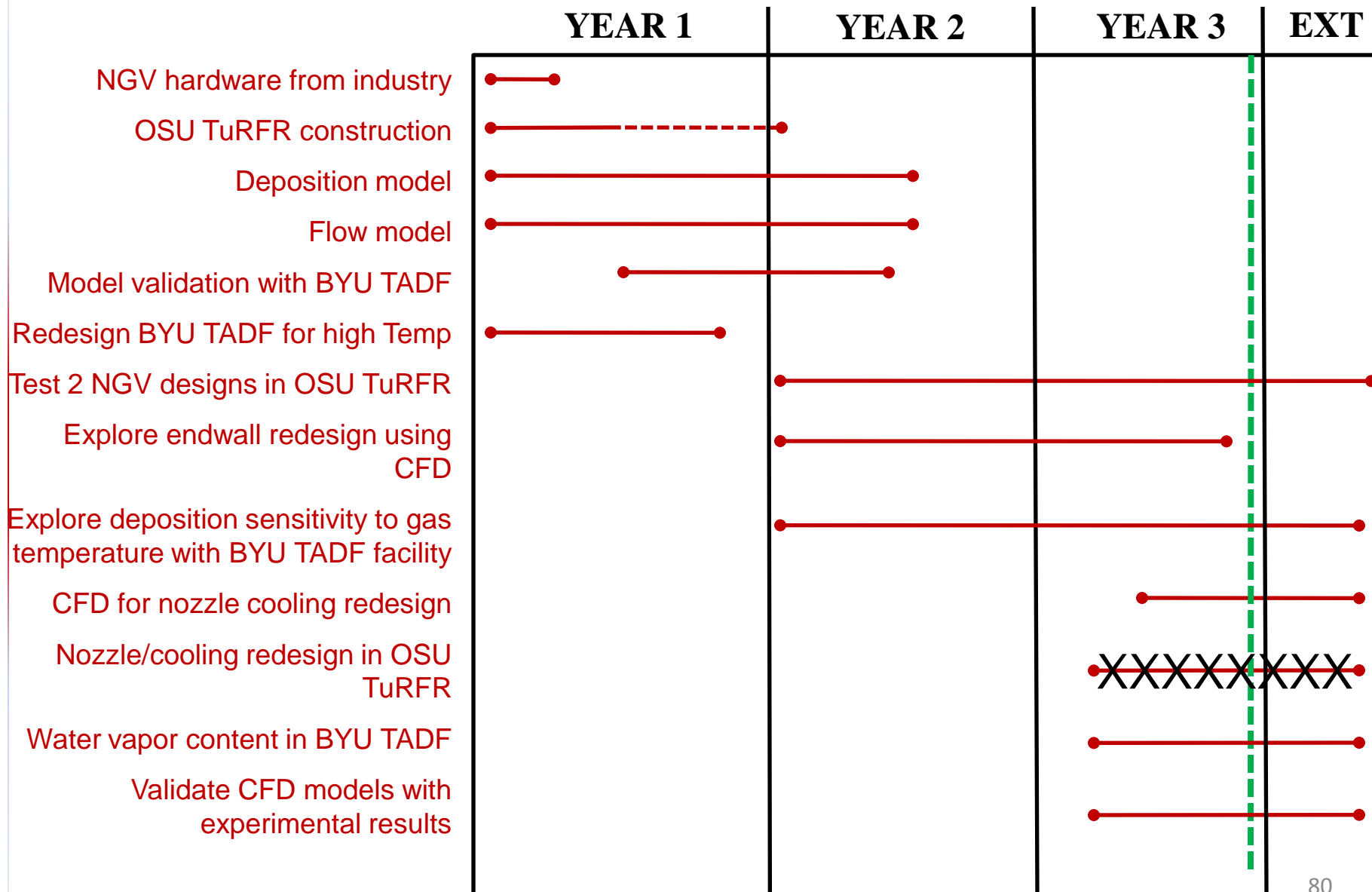
- An increase in surface temperature led to an increase in capture efficiency
- For the tests in which T_g and T_s were known, variation in T_g did not have an effect on capture efficiency
- Tests with similar $T_{s,initial}$ but different T_g developed in a similar manner
- Tests with similar T_g and increasing particle loading had increased capture efficiency
- T_s seems to have a greater effect than particle loading on capture efficiency



BYU - Future Work

- Complete current temperature test series (i.e., with backside cooling)
- Perform film cooling test series at high gas temperatures (up to 1400°C)
- Perform tests with higher water concentration (up to 15 mol%)

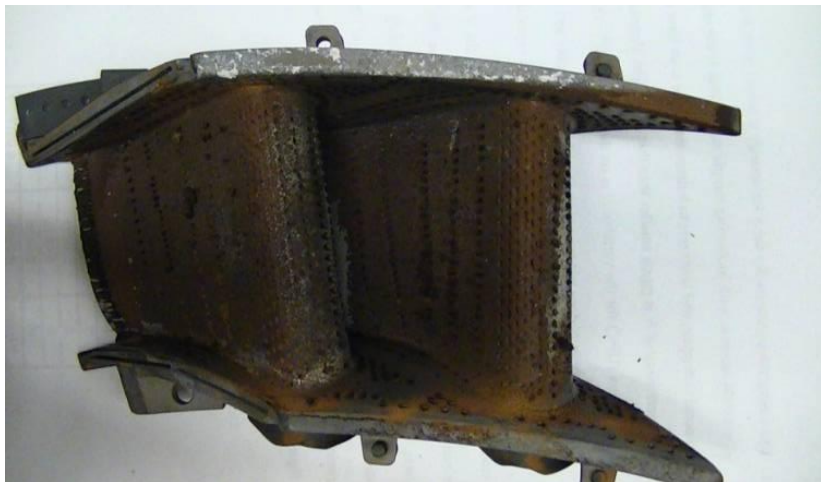
Program Schedule



QUESTIONS?



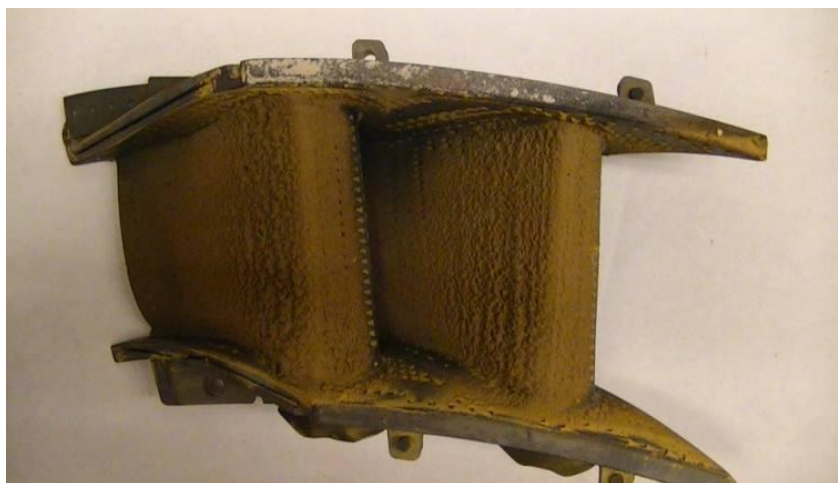
Results with Different Ash



Bituminous Coal Fly Ash
- Limited deposition up to 2050F – High Iron



Lignite Coal Fly Ash
-Thickest deposits at lowest temperature (1900F)

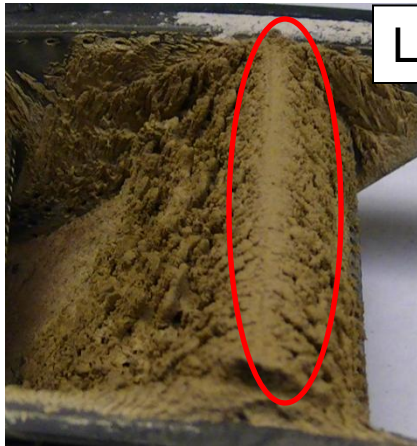


PRB Subbituminous Coal Fly Ash
- Significant deposition above 1920F – High Calcium



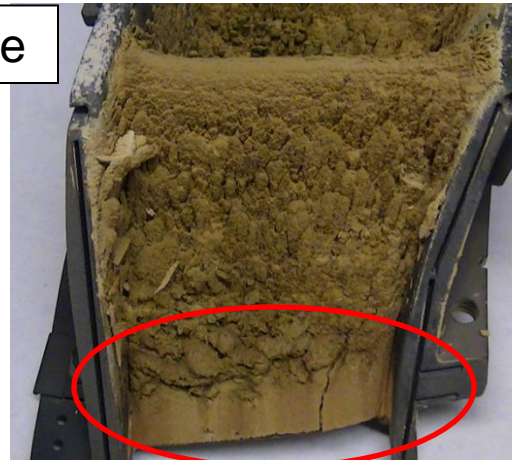
"BYU" Subbituminous Coal Fly Ash
- Comparable deposits to PRB – High Silicon

Results with Different Ash



Lignite

Leading edge stagnation line well defined



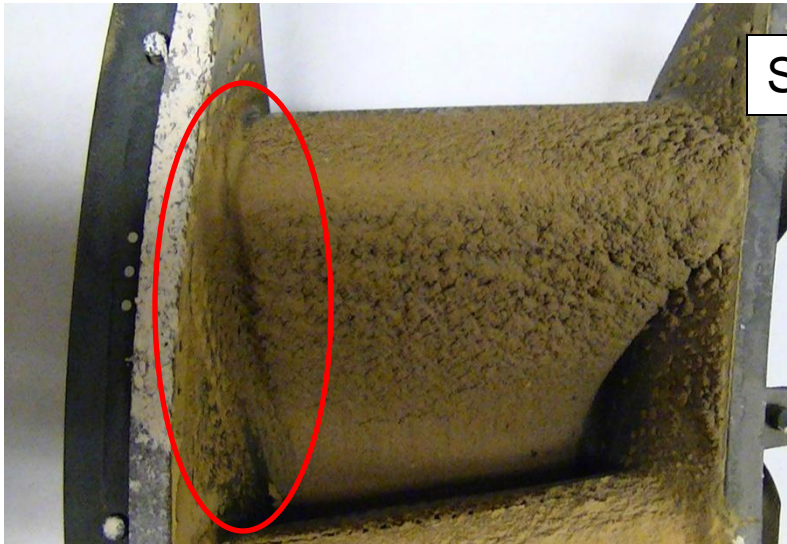
Large structures near trailing edge.
Smooth deposits last 20%.



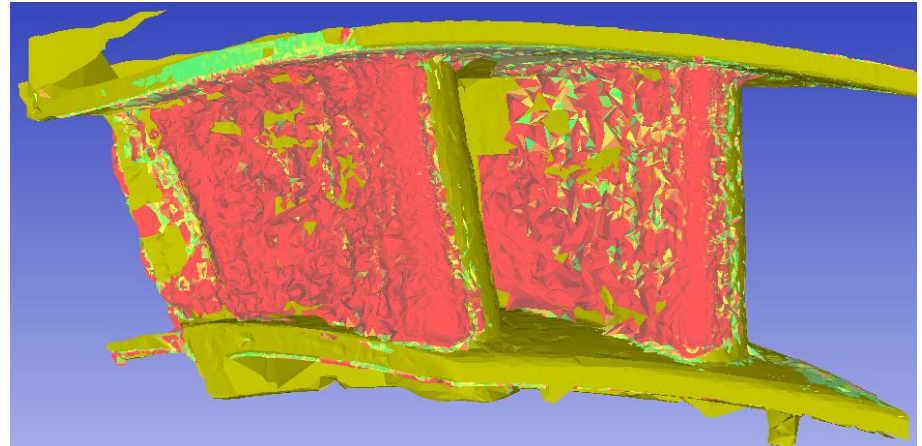
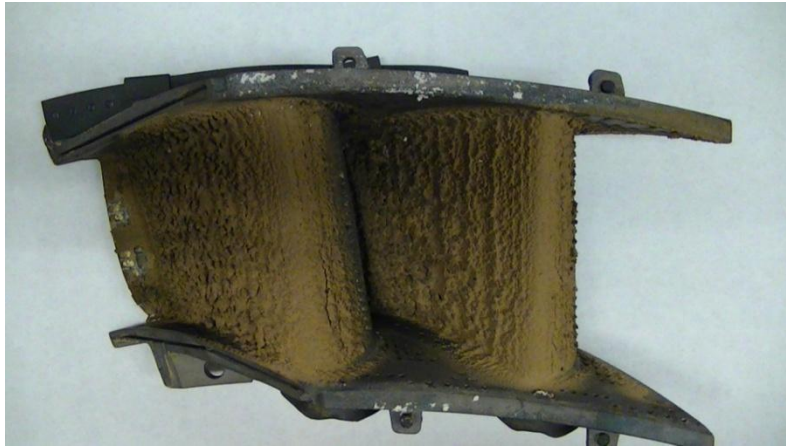
Subbituminous

Evidence of secondary flows on casing (and hub) endwall.

Jagged structures develop in upstream direction on pressure surface. Terminate toward suction surface.

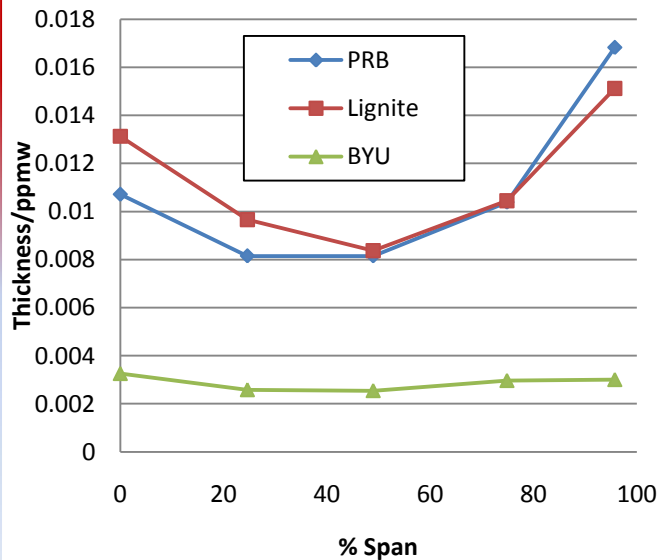


Results with Different Ash

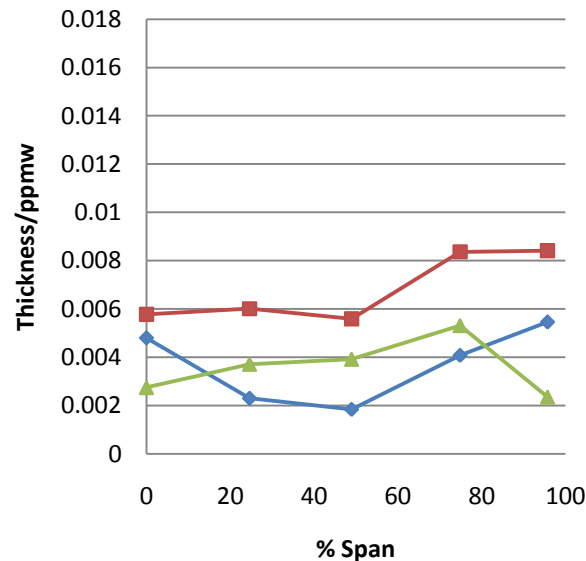


Optical Surface Metrology

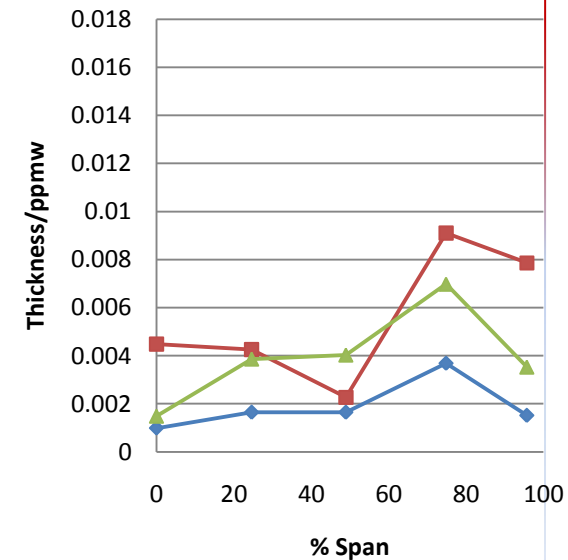
Thickness at Leading Edge



Thickness at 37% Chord



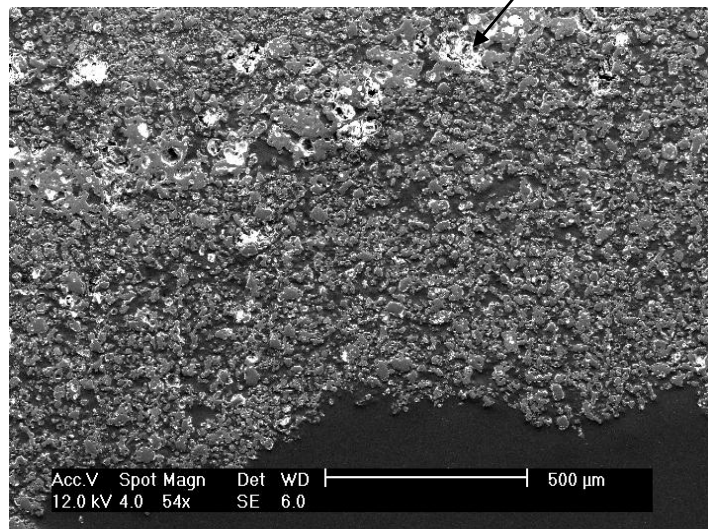
Thickness at 53% Chord



Results with Different Ash

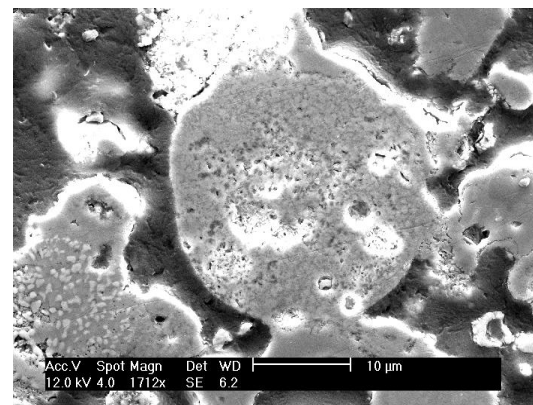
PRB Subbituminous

Vein of more dense material

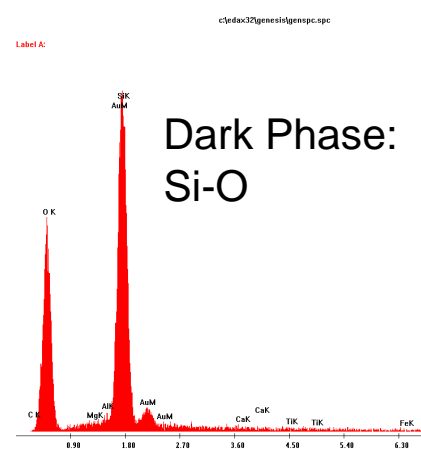
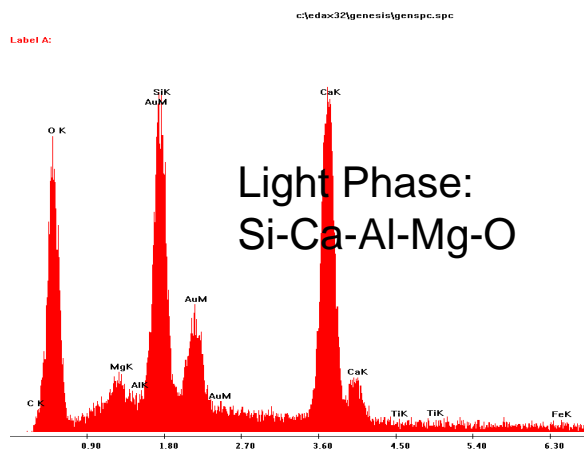
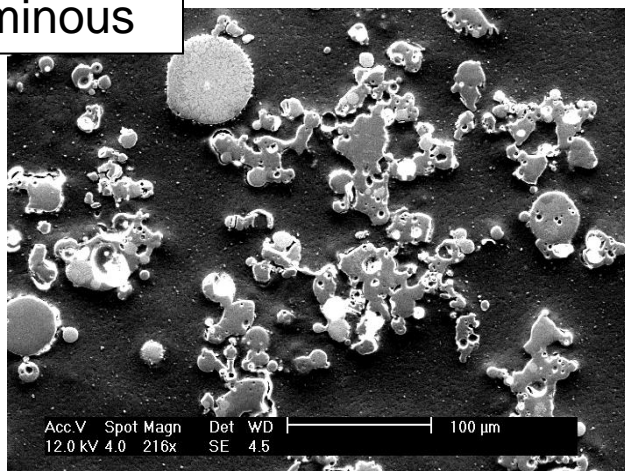


(Bottom of deposit)

Light and dark phases
observed
throughout the
entire sample



Bituminous



Very High
Porosity

*Courtesy of Dr. Nitin Padture's Materials Lab